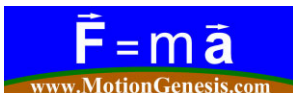
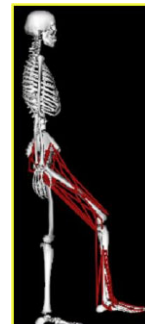
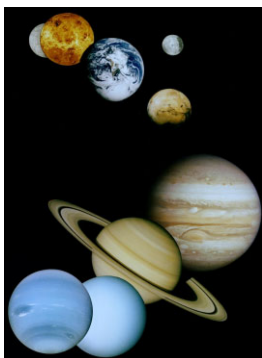
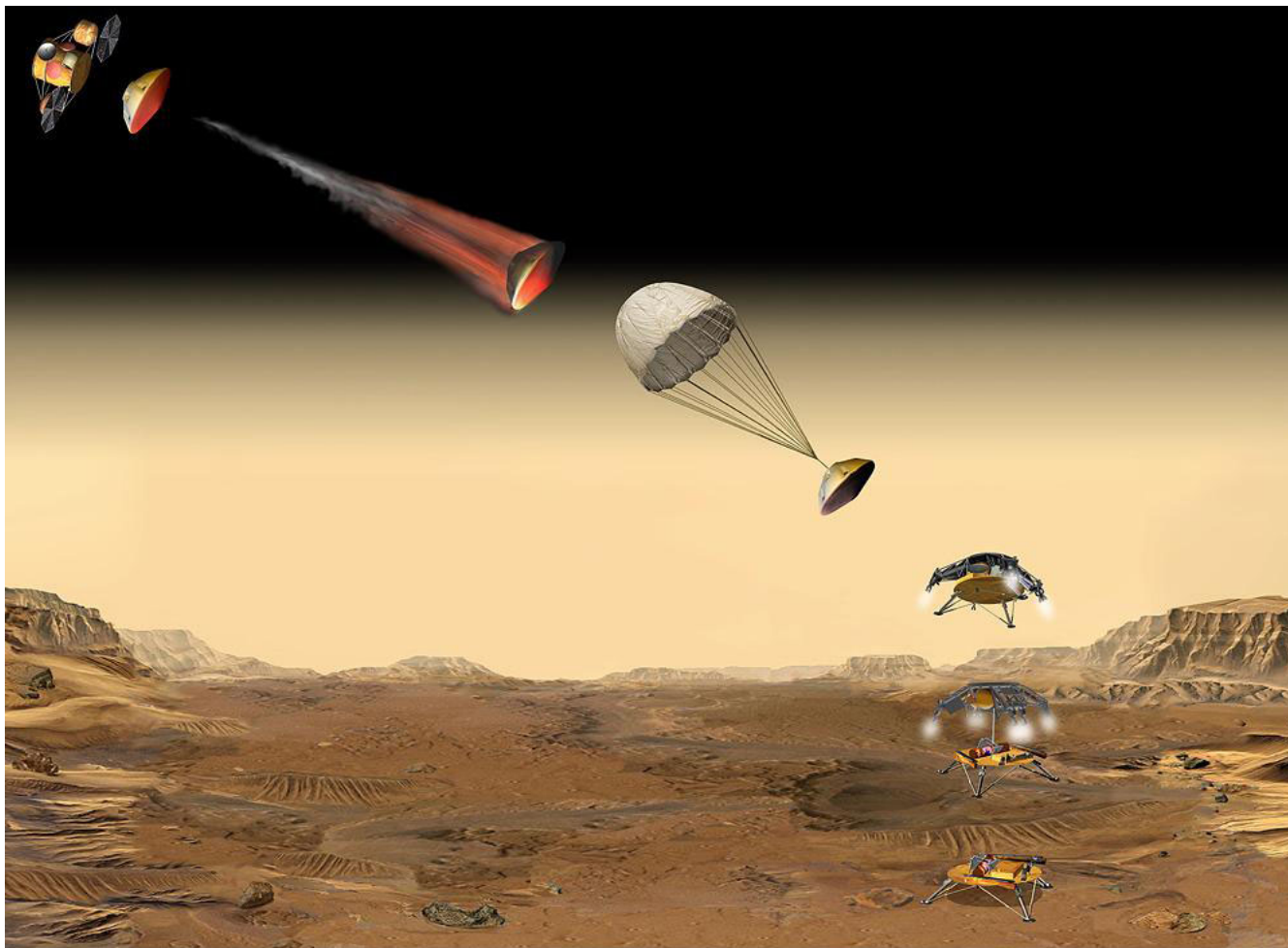


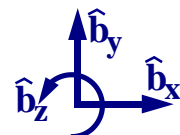
MotionGenesis™ Kane Tutorial

Software, textbooks, training, and consulting
Force, motion, and code-generation tools

$$\vec{F} = m \vec{a} \quad \text{www.MotionGenesis.com}$$



English tutorial: January 13, 2020
Math, forces, motion, & code-generation
More examples at www.MotionGenesis.com ⇒ [Get Started](#)





MotionGenesis Kane

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1 Get started & basic input/output

```
MotionGenesis Kane 5.9: Symbolic solutions for forces and motion.
Professional version. January 2, 2020

Licensed user: Motion Genesis LLC (until January 2023)

Copyright (c) 1988-2020, Motion Genesis LLC. All Rights Reserved.
Copyright includes names of commands, functions, methods, syntax, etc.

Type QUIT to end this session.
Type HELP for a list of commands. See www.MotionGenesis -> Get Started
Type PLOT (or drag-and-drop data file onto MotionGenesis icon).
-----
(1)
```

Follow the download and install directions at www.MotionGenesis.com ⇒ [Get Started](#).

Note: There are **significantly more** examples at www.MotionGenesis.com ⇒ [Get Started](#).

On line (1), type

```
sum = 2 + 2
```

Press Enter and observe the response.

Note: Output lines are preceding with an arrow . → . Some commands do not produce output.

Next, enter the following **symbolic** expression and observe the **automatic simplification**.

Note: Since the program is case-insensitive, you may use upper-case or lower-case letters (or a mix).

```
someName = 2*sin(t)^2 + cos(t)^2
-> someName = 1 + sin(t)^2
```

Saving input

To **save** input to the text file FirstDemo.txt, enter

```
Save FirstDemo.txt
```

Exit the program by typing

```
Quit
```

Running files

Modify the file FirstDemo.txt with a text-editor (e.g., NotePad, SimpleText, TextEdit, Emacs).

Put the following comment line at the top of the file and ensure the editor saves the updated file.

```
% File: FirstDemo.txt
```


To **run** the input file FirstDemo.txt, invoke the program and type¹

```
FirstDemo.txt    or    Run FirstDemo.txt
```

Saving input and output

To **save** input commands together with output responses in the file FirstDemo.html, enter

```
Save FirstDemo.html
```

Printing MotionGenesis files (e.g., to print FirstDemo.txt)

- To **print** a MotionGenesis script (e.g., FirstDemo.txt), open the file in a text editor (e.g., NotePad, SimpleText, TextEdit, Emacs) or a word-processing program with Courier Font (e.g., Microsoft Word) and print it from within that program.
- To **print** the output file FirstDemo.html, double-click on it and select **print** from within your Internet browser.

Online help

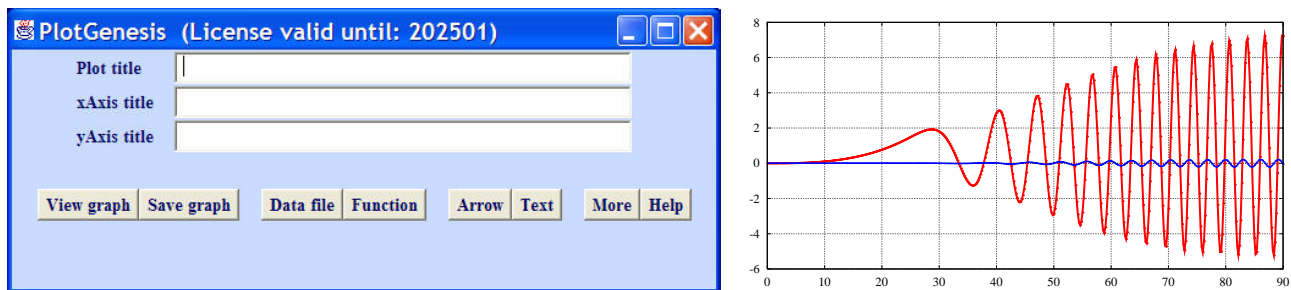
For general **help** and/or a list of commands, type **HELP**.

For help with a command, e.g., SOLVE, type **Help SOLVE**.

1.1 Running the PlotGenesis plotting program

To invoke PlotGenesis, start MotionGenesis, type **Plot**, and follow the on-screen instructions.

Alternately, drag and drop a data file on top of the MotionGenesis icon.



Optional: The PC/Windows version of MotionGenesis allows you to double click on the PlotGenesis icon. or drag and drop data files on top of the **PlotGenesis** icon to start **PlotGenesis** and load the data files.

Note: More examples at www.MotionGenesis.com ⇒ [Get Started](#).

¹Instead of interactively entering commands, it is generally **easier** to use a text editor to create a text file (e.g., FirstDemo.txt) and then execute the commands in that file. Additionally, lines read from a text file can be broken into multiple lines by using an ampersand (&) as the last non-blank character of each but the last input line – which informs MotionGenesis that the command continues on the next line. This is advantageous for entering very long lines (longer than 512 characters).

2 Mathematical declarations

2.1 Scalars

Names of scalar quantities must start with a letter and may be followed by alphanumeric characters or underscores `_`. For example, `x`, `aB3`, and `aBC_3` are acceptable names. Ordinary derivatives of a scalar variable with respect to `t` are denoted with primes `'` at the end of a name. Each prime represents one derivative (e.g., `x''` is the 2nd derivative of `x` with respect to `t`). At most 64 characters (including primes) may be used to name a scalar quantity. Before using a scalar quantity, it must be:

- Declared in **Constant**, **Specified**,² **Variable**, **SetMass**, **SetInertia**, or **SetImaginaryNumber**
- Appeared on the left-hand side of an equals sign in an assignment.

```
Constant  a                % Declares a as a constant
Constant  b = 3 meters     % Declares b as a constant with an input value of 3 meters
Constant  c+               % Declares c to be a non-negative constant
Constant  d-               % Declares d to be a non-positive constant
Specified  phi             % Declares phi as a function of time, constants, and variables
Variable   q, s            % Declares the variables q and s
Variable   x''             % Declares the variables x, x', x''
Variable   u{3}'           % Declares the variables u1, u2, u3, and u1', u2', u3'
SetImaginaryNumber( j )    % Declares j to be the imaginary number, i.e., j = sqrt(-1)
Tina = 2*pi                % Creates the scalar Tina and assigns 2*pi to Tina
```

2.2 Vectors `>`, Dyadics `>>`, and Polyadics `>>>`

The “greater than” character `>` is used to distinguish scalars, vectors, dyadics, and polyadics,

- One `>` is at the end of the name of a **vector** `a >`
- Two `>` are at the end of the name of a **dyadic** `b >>`
- Three `>` are at the end of the name of a **polyadic** `c >>>`

Names of vectors, dyadics, and polyadics must start with a letter. This first letter may be followed by alphanumeric characters or underscores (`_`). A total of 64 characters (including `>` symbols) may be used.

²A *specified* quantity varies in a **known way**, i.e., as a prescribed as a function of constants, time `t`, and other variables. For example, the torque or angular velocity of a rotational **motor** can be specified.

2.3 Matrices

Matrices start with a left bracket [and end with a right bracket]. Elements in a row are separated by a comma, while rows are separated by semicolons. For example, [1, 2, 3; 4, 5, 6] denotes $\begin{bmatrix} 1 & 2 & 3 \\ 4 & 5 & 6 \end{bmatrix}$. The name of a matrix must start with a letter. This letter may be followed by alphanumeric characters or underscores (_). Matrix names are less than 64 characters long.

The name of an element of a matrix consists of the name of the matrix, followed by a left bracket [, an expression that must evaluate to a positive integer, a comma, another expression which must evaluate to a positive integer, and a right bracket], e.g., Ed[2, 5]. Elements of a one-dimensional matrix can be referenced by in shorthand form, namely, the name of a matrix followed by a left bracket [, an expression that evaluates to a positive integer, and a right bracket]. For example, X[7] denotes the 7th element of the matrix X, regardless of whether X is a row matrix or a column matrix.

2.4 Reserved Names

- **t** is a reserved independent variable, often used to denote time.
tInitial, **tFinal**, **tStep** are built-in constants related to **t**.
- **pi** is a reserved constant with the value of 3.14159265...
- **0>** is the zero vector
- **1>>** is the unit dyadic
- **imaginary** is $\sqrt{-1}$, unless declared otherwise with SetImaginaryNumber(...)

For those needing backward compatibility with Autolev:

- **DEPENDENT** and **AUXILIARY** are reserved for matrices of expressions which one sets equal to zero to form motion constraint equations. Type **HELP Constrain** for details.
- **Variables U{n}** or **Variables U{n}'** introduce motion variables.
- **ZERO** is reserved for the matrix of expressions which one sets equal to zero to form dynamical equations of motion. Type **HELP Kane** for details.
- **ZEE_NOT** is reserved for a matrix of scalar quantities that are excluded from Z1,Z2,...

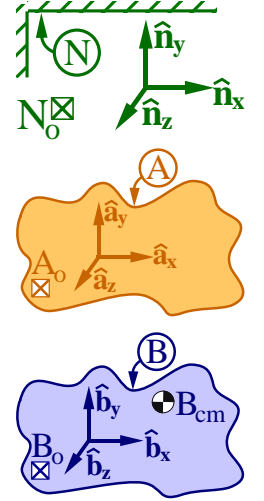
Note: More examples at www.MotionGenesis.com ⇒ [Get Started](#).

3 Physical declarations

3.1 RigidBody, RigidFrame, NewtonianFrame, Point, Particle, System

A Newtonian reference frame must be named in the `NewtonianFrame` declaration and may consist of at most 11 characters. The names of bodies, frames, particles, and points must appear in the declarations `RigidBody`, `RigidFrame`, `Particle`, and `Point` and may consist of at most 27 characters. These names must begin with a letter followed by alphanumeric characters (no underscores).

NewtonianFrame N	Declares N as a Newtonian (inertial) reference frame. Introduces orthonormal vectors \mathbf{N}_x , \mathbf{N}_y , \mathbf{N}_z fixed in N. Declares a point N_0 that is fixed on N.
RigidFrame A	Declares the (massless) reference frame A. Introduces orthonormal vectors \mathbf{A}_x , \mathbf{A}_y , \mathbf{A}_z fixed in A. Declares a point A_0 that is fixed on A.
RigidBody B	Declares the (massive) rigid body B. Introduces orthonormal vectors \mathbf{B}_x , \mathbf{B}_y , \mathbf{B}_z fixed in B. Declares a point B_0 fixed on B. Declares a point B_{cm} fixed on B and at B's center of mass.
Particle C, D	Declares the (massive) particles C and D.
Point E, F	Declares the (massless) points E and F.
Point G(A)	Declares the (massless) point G that is fixed on A.
Point H()	Declares point H as unattached to a frame or body.
System S(A, B)	Declares a system named S consisting of A and B.



3.2 Syntactical Forms

An underscore (_) separates physical names in a position vector, velocity, acceleration, rotation matrix, angular velocity, angular acceleration, force, torque, or inertia dyadic. For example:

\mathbf{p}_{O_Q}	Position vector	from point O to point Q .
\mathbf{v}_{P_N}	Velocity	of point P in N (N is a rigid frame or rigid body).
\mathbf{a}_{D_C}	Acceleration	of point D in C (C is a rigid frame or rigid body).
\mathbf{w}_{B_F}	Angular velocity	of B in F (B and F are rigid frames or rigid bodies).
\mathbf{alf}_{B_F}	Angular acceleration	of B in F .
\mathbf{Force}_P	Force	on point or particle P .
\mathbf{Force}_{P_Q}	Force	on P from point or particle Q .
\mathbf{Torque}_B	Torque	on rigid frame or rigid body B .
\mathbf{Torque}_{B_A}	Torque	on B from rigid frame or rigid body A .
$\mathbf{I}_{B_{B_0}}$	Inertia dyadic	of rigid body B about point B_0 .
\mathbf{A}_B	Rotation matrix	relating unit vectors \mathbf{A}_x , \mathbf{A}_y , \mathbf{A}_z to \mathbf{B}_x , \mathbf{B}_y , \mathbf{B}_z

$$A.B = \begin{bmatrix} \hat{a}_x \cdot \hat{b}_x & \hat{a}_x \cdot \hat{b}_y & \hat{a}_x \cdot \hat{b}_z \\ \hat{a}_y \cdot \hat{b}_x & \hat{a}_y \cdot \hat{b}_y & \hat{a}_y \cdot \hat{b}_z \\ \hat{a}_z \cdot \hat{b}_x & \hat{a}_z \cdot \hat{b}_y & \hat{a}_z \cdot \hat{b}_z \end{bmatrix}$$

3.3 Mass declarations

The masses of bodies and particles are declared by the `SetMass` command. For example:

```
(1) RigidBody A, B
(2) Particle C
(3) A.SetMass( 12.5 ); B.Setmass( mB ); C.SetMass( mC = 10 kg )
(4) massA = A.GetMass()
-> (5) massA = 12.5
(6) massAB = A.GetMass() + B.GetMass()
-> (7) massAB = 12.5 + mB
(8) TotalMass = System.GetMass()
-> (9) TotalMass = 22.5 + mB + mC
```

3.4 Inertia Declarations (for rigid bodies)

A rigid body B 's inertia dyadic about a point B_o can be set by explicit assignment. For example,

- `I_B_Bo>> = 100*Bx>*Bx> + 200*By>*By> + 250*Bz>*Bz>`

A rigid body B 's inertia about point B_o can be set with the `SetInertia` declaration. For example,

- `B.SetInertia(Bo, Ixx, Iyy, Izz, Ixy, Iyz, Izx)` is equivalent to typing
`Constant Ixx+, Iyy+, Izz+, Ixy, Iyz, Izx`
`I_B_Bo>> = Ixx*Bx>*Bx> + Ixy*Bx>*By> + Izx*Bx>*Bz>`
`+ Ixy*By>*Bx> + Iyy*By>*By> + Ixy*By>*Bz>`
`+ Izx*Bz>*Bx> + Iyz*Bz>*By> + Izz*Bz>*Bz>`
- `B.SetInertia(Bo, Ixx, Iyy, Izz)` is equivalent to typing
`Constant Ixx+, Iyy+, Izz+`
`I_B_Bo>> = Ixx*Bx>*Bx> + Iyy*By>*By> + Izz*Bz>*Bz>`

A rigid body B 's inertia about B_{cm} can be set with the `SetMassInertia` declaration. For example,

- `B.SetMassInertia(mB, Ixx, Iyy, Izz)` is equivalent to typing
`B.SetMass(mB)`
`B.SetInertia(Bcm, Ixx, Iyy, Izz)`
- Rigid body B 's inertia dyadic about any point (e.g., B_{cm}) has already been entered, **MotionGenesis** can calculate B 's inertia dyadic about a different point (e.g., B_o) provided B 's mass has been declared and appropriate position vectors have been entered. For example,

```
(1) RigidBody B
(2) B.SetMassInertia( m, I, J, K ) % Sets B's mass and B's inertia about Bcm.
(3) Constant L
(4) p_Bo_Bcm> = L*Bx>
-> (5) p_Bo_Bcm> = L*Bx>
(6) I_B_Bo>> = B.GetInertia( Bo )
-> (7) I_B_Bo>> = I*Bx>*Bx> + (J+m*L^2)*By>*By> + (K+m*L^2)*Bz>*Bz>
```

3.5 Forces and torques

`Force_P>` is the net external force on point or particle P .
`Force_P_Q>` is the net force on P from point or particle Q .
`Torque_B>` is the net external torque on a rigid frame or rigid body B .
`Torque_B_A>` is the net torque on B from rigid frame or rigid body A .

There are multiple ways to assign values to a `Force` or `Torque` vector:

```
P.AddForce( 5*Az> )      % Adds 5*Az> to the previous value of Force_P>
Force_P> -= 6*Az>        % Subtracts 6*Az> from the previous value of Force_P>
Q.AddForce(P, Vec> )     % Adds Vec> to the force on Q from P
Force_Q> := 7*Ax>        % Explicit assignment overwrites previous value for Force_Q>
A.AddTorque( Vec> )      % Adds Vec> to the previous value of Torque_A>
Torque_A> -= Vec>        % Subtracts Vec> from the previous value of Torque_A>
B.AddTorque( A, Vec> )   % Adds Vec> to the torque on B from A
Torque_B> := 7*Bz>      % Explicit assignment overwrites previous value for Torque_B>
```

Note: More examples at www.MotionGenesis.com \Rightarrow [Get Started](#).

4 Procedures for solving problems

4.1 Solving ODEs(ordinary differential equations)

To solve ODEs with **MotionGenesis** or write MATLAB[®], C, or Fortran code that solve ODEs:

- Declare all **variables** and their derivatives.
- Enter equations governing the highest derivative of each variable.
- Use **Input** statements for integration parameters and initial values of variables.
- Use **Output** statements to list quantities to be output.
- Type **ODE()** to immediately solve the ODEs and write output to the file **ODE.1**
Type **ODE() Filename** to immediately solve the ODEs and write output to **Filename.1**
Type **ODE() Filename.ext** (ext is **m**, **c**, **for**, or **f**).
ext is **m** creates ready-to-run MATLAB[®] code in **Filename.m**
ext is **c** creates ready-to-compile C code in **Filename.c** and an input file **Filename.in**.
ext is **f** or **for** creates Fortran code in **Filename.ext** and an input file **Filename.in**.

```
Variable y'' = sin(y)
Input y = 3 meters, y' = 0 m/s, tFinal = 5 second, tStep = 0.1 second
Output t seconds, y m, y' m/s, y'' m/s^2
ODE() test
```

Changing “test” to “test.m” creates the ready-to-run MATLAB[®] file **test.m**.

Changing “test” to “test.c” creates ready-to-compile C file **test.c** and an input file **test.in**. These programs numerically solve the ODE for **y** from **t** = 0 to **t** = 5 with integration steps of 0.1 s.

4.2 Solving nonlinear algebraic equations

- Declare unknowns (e.g., as **variables**).
- Create a matrix whose elements represent the nonlinear equations.
- Use **Input** statements to assign values to constants (and perhaps the convergence parameter **absError**).
- Use the **Solve** command with guessed solutions for the unknowns.

```
% Example: Solve the following nonlinear equations for x and y.
Constant a = 1.0 m, r = 1.0 m
Variable x, y
Eqn[1] = x^2 + y^2 - r % Equation of circle: x^2 + y^2 - r = 0
Eqn[2] = y - a*sin(x) % Equation of sinusoid: y - a*sin(x) = 0
Solve( Eqn = 0, x = 0.5 m, y = 0.5 m )
```

- Alternatively, replace the last line with:

```
Input x = 0.5 m, y = 0.5 m
Code Nonlinear(Eqn, x, y) katy.m % or katy.c or katy.f
```

This creates the ready-to-run MATLAB[®] file **katy.m**. To solve the nonlinear equations, start MATLAB[®] and type **katy** at the MATLAB[®] prompt. Edit **katy.m** to try a different guess for **x** or **y** or to use a different value for **a** or **b**.

4.3 Solving linear algebraic equations

- Declare unknowns (e.g., as **variables**).
- Create a matrix whose elements represent the linear equations.
- Use the **Solve** command.

```
Variable x, y
Constant a{1:2, 1:2}, b{1:2}
Eqn[1] = a11*x + a12*y - b1           % a11*x + a12*y = b1
Eqn[2] = a21*x + a22*y - b2           % a21*x + a22*y = b2
Solve( Eqn = 0, x, y )
```

- Alternatively, replace the last line with:

```
Code Algebraic(Eqn = 0, x, y) becky.c % or becky.f or becky.for
```

This creates the ready-to-compile C file **becky.c** and an input file **becky.in**. The input file is read by the executable program (e.g., **becky.exe**) which solves the linear equations for **x** and **y** for the given values of **a11**, **a12**, **a21**, **a22**.

Note: Edit **becky.in** with a text editor to try different values of **a11**, **a12**, **a21**, **a22**.

4.4 Generic recipe for forming equations of motion (examples in Chapter 9).

MotionGenesis can form statics and dynamics equations with various methods, particularly Newton/Euler, D'Alembert, and Kane's method (textbooks can be purchased via www.MotionGenesis.com ⇒ [Textbooks](#)).

- Declare a **NewtonianFrame**.
- Use the **RigidBody**, **RigidFrame**, **Point**, and **Particle** declarations.
- Declare generalized speeds, e.g., **SetGeneralizedSpeed(wx, wy, wz, x', y', z')**
- Declare generalized coordinates and their time-derivatives with a declaration of the form
Variable qx', qy', qz'
- If necessary, create kinematical differential equations relating time-derivatives of generalized coordinates to the motion variables, e.g., $qx' = wx \cos(qx) + wy \sin(qy)$
- Form rotation matrices, $\vec{\omega}$, and $\vec{\alpha}$, e.g., with the **Rotate** command.
As needed, form other angular velocities/accelerations with **SetAngularVelocityAcceleration**
- Form position vectors, \vec{v} , and \vec{a} , e.g., with the **Translate** and/or **SetPosition** commands.
As needed, form additional velocities/accelerations with **SetVelocityAcceleration**
- If necessary, impose motion constraints with a variation of the **Solve** or **SolveDt** commands
- Add forces and torques with **AddForce** or **AddTorque** commands.

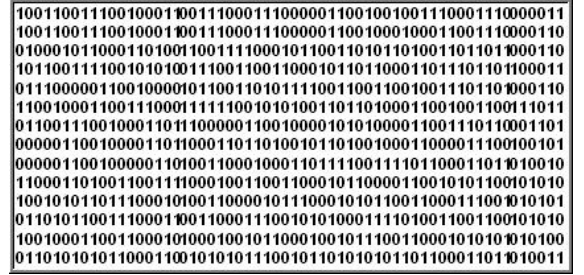
Form particle Q 's dynamics with $\vec{F} = m \vec{a}$	Zero> = Q.GetDynamics()
Form rigid body B 's rotational dynamics about B_{cm}	Zero> = B.GetDynamics(Bcm)
Form system dynamics with $\vec{F} = m \vec{a}$	Zero> = System.GetDynamics()
Form system dynamics with $\vec{M}^{\text{System}/P} = \frac{N_d \vec{H}^{\text{System}/P}}{dt} \dots$	Zero> = System.GetDynamics(P)
• Form sub-system S dynamics $\vec{F} = m \vec{a}$	Zero> = S.GetDynamics()
Form sub-system S dynamics $\vec{M}^{S/P} = \frac{N_d \vec{H}^{S/P}}{dt} \dots$	Zero> = S.GetDynamics(P)
Form system dynamics with Kane's method	Zero = System.GetDynamicsKane()

Note: For large problems, or problems which require real-time solution, type **SetAutoZ(ON)**.

Note: More examples at www.MotionGenesis.com ⇒ [Get Started](#).

Chapter 5

Computer techniques



Do the basic two minute exercises at www.MotionGenesis.com ⇒ [Get Started](#)

5.1 Declaring scalars (constant, variable, specified) in MotionGenesis

Declaration	Description
Constant $g = 9.8 \text{ m/s}^2$	Declares g as a constant and assigns its input value to $9.8 \frac{\text{m}}{\text{s}^2}$
Constant a , b , Fred	Declares a , b , and Fred as constants.
Variable x	Declares x as a variable (unknown)
Variable y'	Declares y and y' (i.e., \dot{y}) as variables (unknowns)
Variable z''	Declares z , z' , and z'' as variables (unknowns)
Variable $z'' = 2\pi t + z$	Declares z , z' , and z'' as variables and assigns $\ddot{z} = 2\pi z + z$
Specified motorSpeed	Declares motorSpeed as specified (known or prescribed)
Specified s'	Declares s and s' as specified (known or prescribed)
Specified h'''	Declares h , h' , h'' , and h''' as specified (known or prescribed)
Specified $h' = \sin(2\pi t * h)$	Declares h and h' (i.e., \dot{h}) as specified and assigns $h' = \sin(2\pi t h)$
SetImaginaryNumber(i)	Declares i as the imaginary number, i.e., $i = \sqrt{-1}$
SetGeneralizedSpeed(q' , v , w)	Declares q' , v , and w as generalized speeds

By default, MotionGenesis defines t as the independent variable, Pi as π , and imaginary as $\sqrt{-1}$.

5.2 Converting units with MotionGenesis



Note: MotionGenesis output results are marked with ->

```
(1) %-----
(2) %Example: ConvertUnits
(3) %-----
(4) inchToCentimeter = ConvertUnits( inch, cm )
-> (5) inchToCentimeter = 2.54

(6) ounceMassToMilligram = ConvertUnits( ozm, mg )
-> (7) ounceMassToMilligram = 28349.52

(8) poundForceToNewton = ConvertUnits( lbf, Newton )
-> (9) poundForceToNewton = 4.448222

(10) sevenMPHToMeterPerSecond = 7 * ConvertUnits( MPH, m/sec )
-> (11) sevenMPHToMeterPerSecond = 3.12928

(12) sixtyMPHToMeterPerSecond = ConvertUnits( (30+30) MPH, m/sec )
-> (13) sixtyMPHToMeterPerSecond = 26.8224

(14) tMinuteToSecond = ConvertUnits( t minutes, seconds )
-> (15) tMinuteToSecond = 60*t
```

5.3 Symbolic differentiation with MotionGenesis



MotionGenesis symbolically calculates *partial derivatives* and *ordinary time-derivatives*.

Note: MotionGenesis output results are marked with ->

```
(1) Variable x, y
(2) z = y*cos(x) + 2*x^2*sin(y)
-> (3) z = y*cos(x) + 2*x^2*sin(y)

(4) partialDerivativeOfZwithRespectToY = D( z, y )
-> (5) partialDerivativeOfZwithRespectToY = cos(x) + 2*x^2*cos(y)

(6) partialDerivativeOfZwithRespectToX = D( z, x )
-> (7) partialDerivativeOfZwithRespectToX = 4*x*sin(y) - y*sin(x)

(8) Variable s' % Declares s as a variable and s' as it's ordinary time-derivative
(9) funct = log(s) + s*exp(s)
-> (10) funct = log(s) + s*exp(s)

(11) ordinaryTimeDerivativeOfFunct = Dt( funct )
-> (12) ordinaryTimeDerivativeOfFunct = (1/s+exp(s)+s*exp(s))*s'
```

5.4 Optional: Numerical calculation of integrals with MotionGenesis

The MotionGenesis **Integrate** command numerically calculates single, double, triple integrals.

The website www.Mathematica.com is a valuable resource for *symbolically* calculating integrals.

```
(1) Variable x, y
(2) integralA = Integrate( 2*x, x=0:4 )
-> (3) integralA = 16

(4) integralB = Integrate( 2*x*abs(cos(x))^3.4*sqrt(exp(x)), x=0:3 )
-> (5) integralB = 10.79699

(6) integralC = Integrate( Integrate( x*y, x=0:y ), y=0:2 )
-> (7) integralC = 2

(8) integralD = Integrate( y^2 * Integrate( sin(x*y), x=0:y ), y=0:2 )
-> (9) integralD = 2.378401
```

5.5 Solutions of *linear algebraic equations*



The next section shows how to solve a single *linear algebraic equation*, e.g., solving for x in

$$3x + 9 \sin(t) - 12 = 0 \quad \text{or} \quad \begin{bmatrix} 3 \end{bmatrix} \begin{bmatrix} x \end{bmatrix} = \begin{bmatrix} -9 \sin(t) + 12 \end{bmatrix}$$

It also shows how to solve two *linear algebraic equations* for y and z , e.g.,

$$\begin{aligned} 3y + 2z + 9 \sin(t) - 12 &= 0 \\ 2y + 4z + 5 \cos(t) - 11 &= 0 \end{aligned} \quad \text{or} \quad \begin{bmatrix} 3 & 2 \\ 2 & 4 \end{bmatrix} \begin{bmatrix} y \\ z \end{bmatrix} = \begin{bmatrix} -9 \sin(t) + 12 \\ -5 \cos(t) + 11 \end{bmatrix}$$

It also shows how to solve four *linear algebraic equations* for x_1, x_2, x_3, x_4 , e.g.,

$$\begin{aligned} 3x_1 + 2x_2 + 2x_3 + 3x_4 &= 9 \sin(t) \\ 2x_1 + 4x_2 + 2x_3 + 3x_4 &= 5 \cos(t) \\ 4x_1 + 5x_2 + 6x_3 + 7x_4 &= 11 \\ 9x_1 + 8x_2 + 7x_3 + 6x_4 &= 15 \end{aligned} \quad \text{or} \quad \begin{bmatrix} 3 & 2 & 2 & 3 \\ 2 & 4 & 2 & 3 \\ 4 & 5 & 6 & 7 \\ 9 & 8 & 7 & 6 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} = \begin{bmatrix} 9 \sin(t) \\ 5 \cos(t) \\ 11 \\ 15 \end{bmatrix}$$

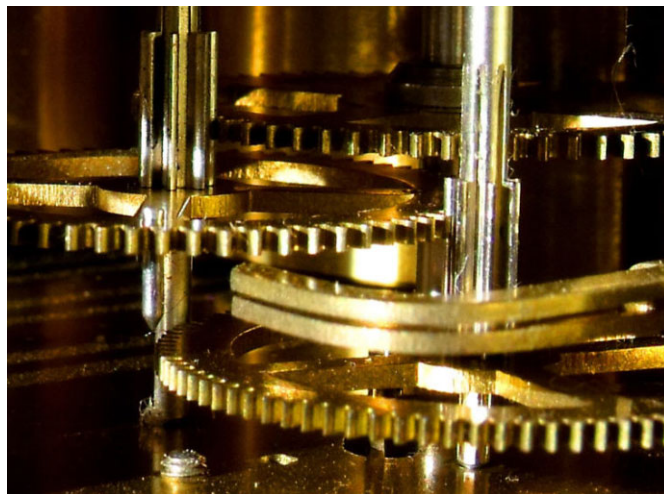
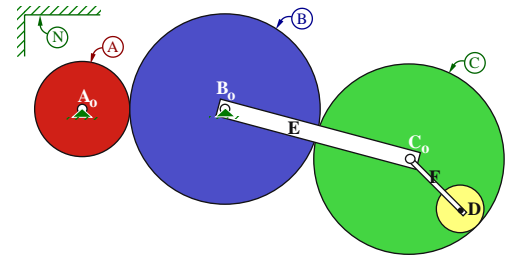
Solutions of previous *linear* algebraic equations with MotionGenesis (symbolic)

```
Variable x
Equation = 3*x + 9*sin(t) - 12
Solve( Equation = 0, x )
%-----
Variable y, z
Zero[1] = 3*y + 2*z + 9*sin(t) - 12
Zero[2] = 2*y + 4*z + 5*cos(t) - 11
Solve( Zero = 0, y, z )
%-----
Variable x{1:4}
Eqn[1] = 3*x1 + 2*x2 + 2*x3 + 3*x4 - 9*sin(t)
Eqn[2] = 2*x1 + 4*x2 + 2*x3 + 3*x4 - 5*cos(t)
Eqn[3] = 4*x1 + 5*x2 + 6*x3 + 7*x4 - 11
Eqn[4] = 9*x1 + 8*x2 + 7*x3 + 6*x4 - 15
Solve( Eqn = 0, x1, x2, x3, x4 )
%-----
Save SolveLinearEquations.all
Quit
```

Example: Symbolic solution of linear equations for gear constraints with MotionGenesis

The following MotionGenesis file solves a set of seven coupled linear algebraic equations. This analysis regards ${}^N\omega^E$ and ${}^N\omega^F$ as free variables (associated with the two degrees of freedom) and ${}^N\omega^A$ as *specified*.

```
% File: GearTrainABCD.txt
%-----
Constant  rA, rB, rC, rD      % Gear radii
Specified wAN                    % Known function of time
Variable  wBN, wCN, wDN      % Unknown gear angular rates
Variable  wEN, wFN            % Unknown rigid rod angular rates
Variable  wBE, wCE            % Useful for analysis
Variable  wCF, wDF            % Useful for analysis
%-----
%      Motion constraints
Zero[1] = wAN*rA + wBN*rB      % Rolling contact between A and B
Zero[2] = wBE*rB + wCE*rC      % Rolling contact between B and C
Zero[3] = wCF*rC - wDF*rD      % Rolling contact between C and D
Zero[4] = wBN - (wEN + wBE)    % Angular velocity addition theorem
Zero[5] = wCN - (wEN + wCE)    % Angular velocity addition theorem
Zero[6] = wCN - (wFN + wCF)    % Angular velocity addition theorem
Zero[7] = wDN - (wFN + wDF)    % Angular velocity addition theorem
Solve( Zero, wBN, wCN, wDN, wBE, wCE, wCF, wDF )
%-----
Save GearTrainABCD.all
Quit
```



5.6 Solution of *quadratic* and *polynomial* equations (roots)

Polynomial equations are a special class of nonlinear algebraic equations. Although there are closed-form solutions for linear, quadratic, cubic, and quartic polynomial equations, there are no general closed-form solutions for 5th and higher-order polynomials.

Symbolic roots of quadratic equation $ax^2 + bx + c = 0$ with MotionGenesis

```
(1) %-----
(2) % Example 1: GetLinearRoots      (roots of linear equation).
(3) %-----
(4) Variable x
(5) Constant a, b, c
(6) root1A = GetLinearRoot( a*x = b,  x )
-> (7) root1A = b/a

(8) root1B = GetLinearRoot( [a; -b] )
-> (9) root1B = b/a

(10) %-----
(11) % Example 2: GetQuadraticRoots  (roots of quadratic equation).
(12) %-----
(13) root2A = GetQuadraticRoots( a*x^2 + b*x + c = 0,  x )
-> (14) root2A[1] = -0.5*(b+sqrt(b^2-4*a*c))/a
-> (15) root2A[2] = -0.5*(b-sqrt(b^2-4*a*c))/a

(16) positiveRootA = GetQuadraticPositiveRoot( a*x^2 + b*x + c = 0,  x )
-> (17) positiveRootA = -0.5*(b-sqrt(b^2-4*a*c))/a

(18) negativeRootA = GetQuadraticNegativeRoot( a*x^2 + b*x + c = 0,  x )
-> (19) negativeRootA = -0.5*(b+sqrt(b^2-4*a*c))/a

(20) root2B = GetQuadraticRoots( [a; b; c] )
-> (21) root2B[1] = -0.5*(b+sqrt(b^2-4*a*c))/a
-> (22) root2B[2] = -0.5*(b-sqrt(b^2-4*a*c))/a

(23) %-----
(24) % Example 3: GetCubicRoots      (roots of 3rd-order polynomial).
(25) %-----
(26) SetImaginaryNumber( i )
(27) Variable p
(28) root3A = GetCubicRoots( 3*p^3 + 5*p^2 + 9*p + 17 = 0,  p )
-> (29) root3A = [-1.775053; 0.05419336 - 1.785905*i; 0.05419336 + 1.785905*i]

(30) root3B = GetCubicRoots( [3, 5, 9, 17] )
-> (31) root3B = [-1.775053, 0.05419336 - 1.785905*i, 0.05419336 + 1.785905*i]

(32) %-----
(33) % Example 4: GetQuarticRoots    (roots of 4th-order polynomial).
(34) %-----
(35) root4A = GetQuarticRoots( 2*p^4 + 3*p^3 + 5*p^2 + 9*p + 17 = 0,  p )
-> (36) root4A = [-1.361636 - 1.014663*i; -1.361636 + 1.014663*i; 0.6116363
- 1.604248*i; 0.6116363 + 1.604248*i]

(37) root4B = GetQuarticRoots( [2, 3, 5, 9, 17] )
-> (38) root4B = [-1.361636 - 1.014663*i, -1.361636 + 1.014663*i, 0.6116363 - 1.604248*i, 0.6116363 + 1.604248*i]

(39) %-----
(40) % Example 5: GetQuinticRoots    (roots of 5th-order polynomial).
(41) %-----
(42) root5A = GetQuinticRoots( p^5 + 2*p^4 + 3*p^3 + 5*p^2 + 9*p + 17 = 0,  p )
-> (43) root5A = [-1.857621; -0.9475112 - 1.507048*i; -0.9475112 + 1.507048*i;
0.8763218 - 1.455989*i; 0.8763218 + 1.455989*i]

(44) root5B = GetQuinticRoots( [1, 2, 3, 5, 9, 17] )
-> (45) root5B = [-1.857621, -0.9475112 - 1.507048*i, -0.9475112 + 1.507048*i, 0.8763218 - 1.455989*i, 0.8763218 + 1.455989*i]
```

```

(46) %-----
(47) % Example 6: GetPolynomialRoots (roots of nth-order polynomial).
(48) %-----
(49) root7A = GetPolynomialRoots( p^7 + 2*p^3 + 13 = 0, p, 7 )
-> (50) root7A = [-1.348251; -0.9844658 - 1.16944*i; -0.9844658 + 1.16944*i;
0.3798201 - 1.331288*i; 0.3798201 + 1.331288*i; 1.278771 - 0.7194889*i; 1.278771 + 0.7194889*i]

(51) root7B = GetPolynomialRoots( [1, 0, 0, 0, 2, 0, 0, 13] )
-> (52) root7B = [-1.348251, -0.9844658 - 1.16944*i, -0.9844658 + 1.16944*i,
0.3798201 - 1.331288*i, 0.3798201 + 1.331288*i, 1.278771 - 0.7194889*i, 1.278771 + 0.7194889*i]

(53) %-----
(54) % Example 7: GetPolynomial
(55) %-----
(56) quadraticFunctionA = GetPolynomial( [t, 5, 3], x )
-> (57) quadraticFunctionA = 3 + 5*x + t*x^2

(58) approximateAsCubic = GetCubicPolynomial( sin(x), x )
-> (59) approximateAsCubic = [-0.1666667; 0; 1; 0]

(60) approximateAs5thOrder = GetPolynomial( cos(x), x, 5 )
-> (61) approximateAs5thOrder = [0; 0.04166667; 0; -0.5; 0; 1]

```

Roots of 5th-order polynomial $p^5 + 2p^4 + 3p^3 + 5p^2 + 9p + 17 = 0$ with MotionGenesis

```

(1) %-----
(2) % Example 1: GetLinearRoots (roots of linear equation).
(3) %-----
(4) Variable x
(5) Constant a, b, c
(6) root1A = GetLinearRoot( a*x = b, x )
-> (7) root1A = b/a

(8) root1B = GetLinearRoot( [a; -b] )
-> (9) root1B = b/a

(10) %-----
(11) % Example 2: GetQuadraticRoots (roots of quadratic equation).
(12) %-----
(13) root2A = GetQuadraticRoots( a*x^2 + b*x + c = 0, x )
-> (14) root2A[1] = -0.5*(b+sqrt(b^2-4*a*c))/a
-> (15) root2A[2] = -0.5*(b-sqrt(b^2-4*a*c))/a

(16) positiveRootA = GetQuadraticPositiveRoot( a*x^2 + b*x + c = 0, x )
-> (17) positiveRootA = -0.5*(b-sqrt(b^2-4*a*c))/a

(18) negativeRootA = GetQuadraticNegativeRoot( a*x^2 + b*x + c = 0, x )
-> (19) negativeRootA = -0.5*(b+sqrt(b^2-4*a*c))/a

(20) root2B = GetQuadraticRoots( [a; b; c] )
-> (21) root2B[1] = -0.5*(b+sqrt(b^2-4*a*c))/a
-> (22) root2B[2] = -0.5*(b-sqrt(b^2-4*a*c))/a

(23) %-----
(24) % Example 3: GetCubicRoots (roots of 3rd-order polynomial).
(25) %-----
(26) SetImaginaryNumber( i )
(27) Variable p
(28) root3A = GetCubicRoots( 3*p^3 + 5*p^2 + 9*p + 17 = 0, p )
-> (29) root3A = [-1.775053; 0.05419336 - 1.785905*i; 0.05419336 + 1.785905*i]

(30) root3B = GetCubicRoots( [3, 5, 9, 17] )
-> (31) root3B = [-1.775053, 0.05419336 - 1.785905*i, 0.05419336 + 1.785905*i]

(32) %-----
(33) % Example 4: GetQuarticRoots (roots of 4th-order polynomial).
(34) %-----

```

```

(35) root4A = GetQuarticRoots( 2*p^4 + 3*p^3 + 5*p^2 + 9*p + 17 = 0, p )
-> (36) root4A = [-1.361636 - 1.014663*i; -1.361636 + 1.014663*i; 0.6116363
- 1.604248*i; 0.6116363 + 1.604248*i]

(37) root4B = GetQuarticRoots( [2, 3, 5, 9, 17] )
-> (38) root4B = [-1.361636 - 1.014663*i, -1.361636 + 1.014663*i, 0.6116363 - 1.604248*i, 0.6116363 + 1.604248*i]

(39) %-----
(40) % Example 5: GetQuinticRoots (roots of 5th-order polynomial).
(41) %-----
(42) root5A = GetQuinticRoots( p^5 + 2*p^4 + 3*p^3 + 5*p^2 + 9*p + 17 = 0, p )
-> (43) root5A = [-1.857621; -0.9475112 - 1.507048*i; -0.9475112 + 1.507048*i;
0.8763218 - 1.455989*i; 0.8763218 + 1.455989*i]

(44) root5B = GetQuinticRoots( [1, 2, 3, 5, 9, 17] )
-> (45) root5B = [-1.857621, -0.9475112 - 1.507048*i, -0.9475112 + 1.507048*i, 0.8763218 - 1.455989*i, 0.8763218 + 1.455989*i]

(46) %-----
(47) % Example 6: GetPolynomialRoots (roots of nth-order polynomial).
(48) %-----
(49) root7A = GetPolynomialRoots( p^7 + 2*p^3 + 13 = 0, p, 7 )
-> (50) root7A = [-1.348251; -0.9844658 - 1.16944*i; -0.9844658 + 1.16944*i;
0.3798201 - 1.331288*i; 0.3798201 + 1.331288*i; 1.278771 - 0.7194889*i; 1.278771 + 0.7194889*i]

(51) root7B = GetPolynomialRoots( [1, 0, 0, 0, 2, 0, 0, 13] )
-> (52) root7B = [-1.348251, -0.9844658 - 1.16944*i, -0.9844658 + 1.16944*i,
0.3798201 - 1.331288*i, 0.3798201 + 1.331288*i, 1.278771 - 0.7194889*i, 1.278771 + 0.7194889*i]

(53) %-----
(54) % Example 7: GetPolynomial
(55) %-----
(56) quadraticFunctionA = GetPolynomial( [t, 5, 3], x )
-> (57) quadraticFunctionA = 3 + 5*x + t*x^2

(58) approximateAsCubic = GetCubicPolynomial( sin(x), x )
-> (59) approximateAsCubic = [-0.1666667; 0; 1; 0]

(60) approximateAs5thOrder = GetPolynomial( cos(x), x, 5 )
-> (61) approximateAs5thOrder = [0; 0.04166667; 0; -0.5; 0; 1]

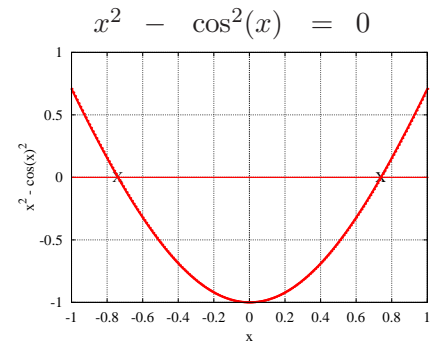
```


5.7 Solutions of *nonlinear* algebraic equations

The graph of the function $x^2 - \cos^2(x)$ is **nonlinear** (i.e., it is **not a line**) and has two solutions, namely $x \approx 0.74$ and $x \approx -0.74$.

MotionGenesis solves *nonlinear algebraic equations* using a numerical algorithm that requires a **guess** and iterates towards a solution (frequently it converges to the solution closest to the guess).

```
(1) Variable x
(2) Solve( x^2 - cos(x)^2 = 0, x = 2 ) % x = 2 is a guess
-> (3) x = 0.7390851
```



5.7.1 Solutions of *coupled nonlinear* algebraic equations with MotionGenesis

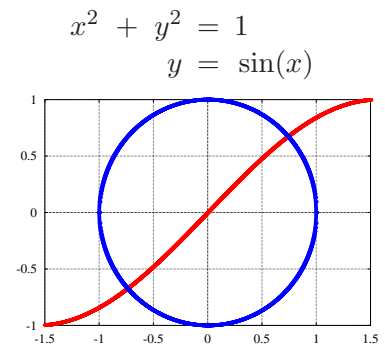
The algebraic equations shown below are **nonlinear** in x and y . These two curves (**not lines**) intersect at two locations (there are two solutions to these equations), namely $x \approx 0.74$, $y \approx 0.67$ and $x \approx -0.74$, $y \approx -0.67$.

In general, it is difficult to determine the *number of solutions* to nonlinear algebraic equations. MotionGenesis solves *nonlinear algebraic equations* using a numerical algorithm that requires a **guess** and iterates towards a solution.

```
(1) Variable x, y
(2) Zero[1] = x^2 + y^2 - 1 % x^2 + y^2 = 1 (unit circle)
-> (3) Zero[1] = -1 + x^2 + y^2

(4) Zero[2] = y - sin(x) % y = sin(x) (sine wave)
-> (5) Zero[2] = y - sin(x)

(6) Solve( Zero = 0, x = 1.5, y = 0 ) % x = 1.5, y = 0 is a guess.
-> (7) x = 0.7390851
-> (8) y = 0.673612
```



Nonlinear algebraic equations frequently arise in determining equilibrium configurations of static systems. Although nonlinear equations with **one** or **two** unknowns can be solved by **graphing**, generally, *Newton-Raphson* techniques are used to solve sets of nonlinear equations.

Creating nonlinear algebraic MATLAB®, C, or Fortran code with MotionGenesis

As shown in the following example, MotionGenesis can also produce efficient **distributable** MATLAB®, C, or Fortran codes that solve coupled nonlinear algebraic equations.

```
Variable x, y
Zero[1] = x^2 + y^2 - 1 % x^2 + y^2 = 1 (unit circle)
Zero[2] = y - sin(x) % y = sin(x) (sine wave)
Input x = 1.5, y = 0 % x = 1.5, y = 0 is guess for solution
CODE Nonlinear( Zero = 0, x, y ) NonlinearSolve.m % Writes MATLAB .m file.
Quit
```

5.7.2 Starting guesses and solutions of coupled *nonlinear* algebraic equations

In the range $0 \leq x \leq 24$, the following **nonlinear** algebraic equation has **5** solutions.

$$-10 + 5 \cos(x) + 40 \sin(0.023x) = 0 \quad \Rightarrow \quad x \approx 5.88, 7.07, 11.0, 14.9, 16.15$$

(Solutions)

The following MotionGenesis file shows solutions to this nonlinear equation are **highly sensitive** to guesses near $x = 6.45$, $x = 9.25$, $x = 12.75$, $x = 15.55$. Guesses of $x < 2.5$ or $x > 20$ may not converge to a solution. Other guesses (especially near local maximum or minimum) may not converge to their closest solution.


```

(1) % File:  HighlyNonlinearEquation.al
(2) %-----
(3) Variable  x
(4) Eqn = -10 + 5*cos(x) + 40*sin(0.023*x)
-> (5) Eqn = -10 + 5*cos(x) + 40*sin(0.023*x)

(6) ans0 = Solve( Eqn = 0,  x = 3.2 )
-> (7) ans0 = [5.882719]

(8) ans1 = Solve( Eqn = 0,  x = 6.4 )
-> (9) ans1 = [5.882719]

(10) ans2 = Solve( Eqn = 0,  x = 6.5 )
-> (11) ans2 = [7.072387]

(12) ans3 = Solve( Eqn = 0,  x = 9.2 )
-> (13) ans3 = [7.072387]

(14) ans4 = Solve( Eqn = 0,  x = 9.3 )
-> (15) ans4 = [10.99414]

(16) ans5 = Solve( Eqn = 0,  x = 12.7 )
-> (17) ans5 = [10.99414]

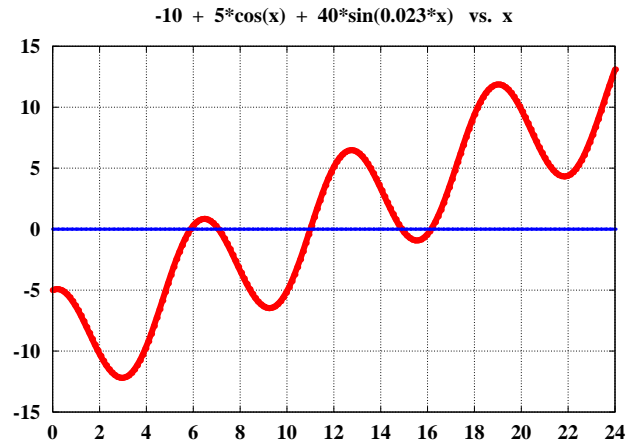
(18) ans6 = Solve( Eqn = 0,  x = 12.8 )
-> (19) ans6 = [14.89506]

(20) ans7 = Solve( Eqn = 0,  x = 15.5 )
-> (21) ans7 = [14.89506]

(22) ans8 = Solve( Eqn = 0,  x = 15.6 )
-> (23) ans8 = [16.15025]

(24) ans9 = Solve( Eqn = 0,  x = 18.95 )
-> (25) ans9 = [16.15025]

```



This nonlinear equation is associated with a radiation machine targeting a cancer cell.

5.7.3 Continuous solutions of *nonlinear* algebraic equations

One way to find a continuous solution for x in the range $0 \leq t \leq 8$ for

$$x^2 - \cos^2(x) = 0.3 \sin(t)$$

is to differentiate this *nonlinear* equation with respect to t and then solve the derivative equation that is *linear* in \dot{x} as

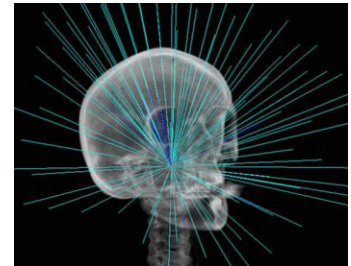
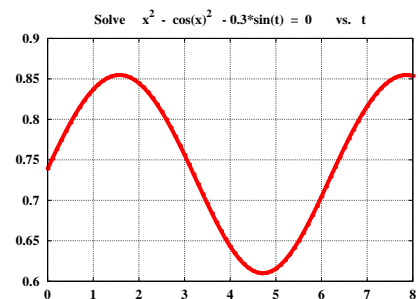
$$2x\dot{x} + 2\cos(x)\sin(x)\dot{x} = 0.3\sin(t) \quad \Rightarrow \quad \dot{x} = \frac{0.3\cos(t)}{2x + 2\cos(x)\sin(x)}$$

Solving the nonlinear equation *once* at $t = 0$ gives $x(t=0) \approx 0.74$. With this initial value for x and continuous formula for \dot{x} , ODE techniques can numerically integrate $\dot{x}(t)$ to solve for $x(t)$.

```

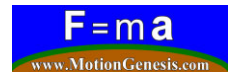
% File:  NonlinearSolveContinuous.al
%-----
Variable  x'
eqn = x^2 - cos(x)^2 - 0.3*sin(t)
Solve( Dt(eqn) = 0,  x' )
SolveSetInput( Evaluate(eqn, t = 0),  x = 1 )
Input  tFinal = 8 sec,  tStep = 0.1 sec
Output t,  x
ODE() NonlinearSolveContinuous
Quit

```



Courtesy Accuray Inc.

5.8 Solution of ordinary differential equations (ODEs)



Computers have revolutionized **numerical solution** of ODEs. **Compiled** codes such as C and Fortran optimize for a specific operating system, microprocessor, and cache and can be **100x faster** than **interpreted** codes such as MATLAB[®]. This difference is significant for real-time operation or when compiled code require minutes to execute (which means interpreted code may require hours).

5.8.1 Solution of 1st-order ODE (numerical integration)

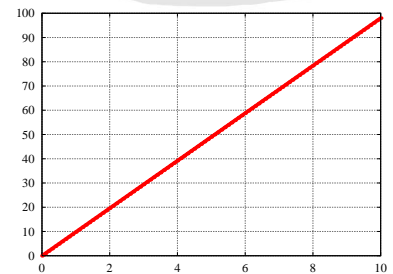
The figure to the right shows a parachutist in vertical free-fall. When air-resistance and other forces than gravity are neglected, the parachutist's downward speed v is governed by the 1st-order ODE

$$\frac{dv}{dt} = 9.8$$

Although this ODE is easily solvable by *separation of variables* and integration as $v(t) = v(0) + 9.8t$, it can also be solved by computer numerical integration as shown in the following MotionGenesis file.

```
% File: ParachutistFreeFallSpeed.txt
%-----
Variable v' = 9.8
Input    v = 0          % Initial value
ODE() ParachutistFreeFallSpeed
Quit
```

Note: To generate MATLAB[®], C, or Fortran code to solve the ODE, append the suffix .m, .c, or .for, to the filename. For example, to generate MATLAB[®] code, replace the last line with `ODE() ParachutistFreeFallSpeed.m`



5.8.2 Solution of 2nd-order ODEs (numerical integration)

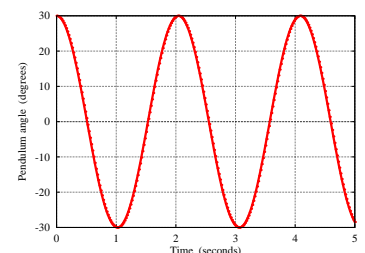
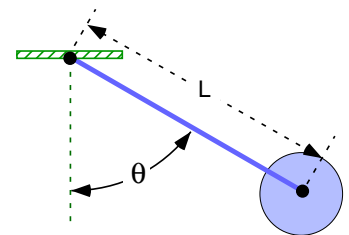
The figure to the right shows a 1 m pendulum swinging on Earth's surface. The pendulum's motion is governed by the **nonlinear** 2nd-order ODE

$$\ddot{\theta} = -9.8 \sin(\theta)$$

The MotionGenesis solution to this **nonlinear** ODE is shown below.

```
% File: ClassicParticlePendulumShort.al
%-----
Variable theta'' = -9.8*sin(theta)
Input  theta = 30 deg, theta' = 0, tFinal = 5, tStep = 0.02
Output t sec, theta deg
ODE() ClassicParticlePendulum
Quit
```

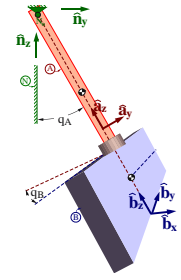
Note: To generate MATLAB[®], C, or Fortran code to solve the ODE, append the suffix .m, .c, or .for, to the filename. For example, to generate MATLAB[®] code, replace the last line with `ODE() ClassicParticlePendulum.m`



5.8.3 Solution of *coupled nonlinear 2nd*-order ODEs

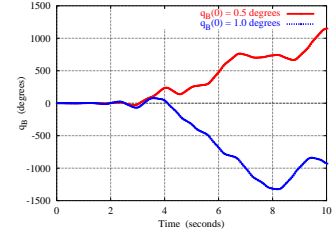
The motion of the system to the right is governed by ODEs that can exhibit “chaotic” behavior (small changes in initial values, physical parameters, or numerical integration accuracy lead to dramatically different behavior).

$$\ddot{q}_A = \frac{2 [508.89 \sin(q_A) - \sin(q_B) \cos(q_B) \dot{q}_A \dot{q}_B]}{-21.556 + \sin^2(q_B)} \quad \ddot{q}_B = -\sin(q_B) \cos(q_B) \dot{q}_A^2$$



The following **MotionGenesis** commands solve these ODEs.

```
Variable qA'', qB''          % Angles and first/second time-derivatives.
%-----
qA'' = 2*( 508.89*sin(qA) - sin(qB)*cos(qB)*qA'*qB' ) / (-21.556 + sin(qB)^2)
qB'' = -sin(qB)*cos(qB)*qA'^2
%-----
Input  tFinal = 10 sec, tStep = 0.02 sec, absError = 1.0E-07
Input  qA = 90 deg, qB = 1.0 deg, qA' = 0.0 rad/sec, qB' = 0.0 rad/sec
OutputPlot t sec, qA degrees, qB degrees
%-----
ODE() solveBabybootODE
Quit
```



Note: To generate MATLAB[®], C, or Fortran code to solve the ODE, append the suffix .m, .c, or .for, to the filename. For example, to generate MATLAB[®] code, replace the last line with `ODE() solveBabybootODE.m`

5.8.4 Solution of coupled ODEs with additional output (spinning rigid body)

The ODEs governing 3D rotational motions of a torque-free rigid body *B* are:

Quantity	Symbol	Value
<i>B</i> 's central moment of inertia for $\hat{\mathbf{b}}_x$	I_{xx}	1 kg m ²
<i>B</i> 's central moment of inertia for $\hat{\mathbf{b}}_y$	I_{yy}	2 kg m ²
<i>B</i> 's central moment of inertia for $\hat{\mathbf{b}}_z$	I_{zz}	3 kg m ²
$\hat{\mathbf{b}}_x$ measure of $\vec{\omega}^B$	ω_x	Variable
$\hat{\mathbf{b}}_y$ measure of $\vec{\omega}^B$	ω_y	Variable
$\hat{\mathbf{b}}_z$ measure of $\vec{\omega}^B$	ω_z	Variable

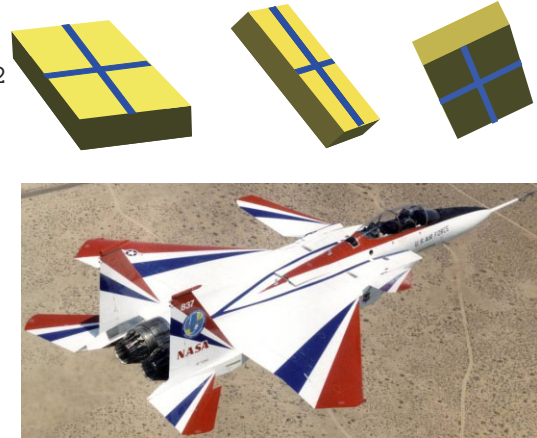
$$\dot{\omega}_x = \frac{(I_{yy} - I_{zz})}{I_{xx}} \omega_z \omega_y$$

$$\dot{\omega}_y = \frac{(I_{zz} - I_{xx})}{I_{yy}} \omega_x \omega_z$$

$$\dot{\omega}_z = \frac{(I_{xx} - I_{yy})}{I_{zz}} \omega_y \omega_x$$

A **MotionGenesis** solution¹ to these ODEs for $0 \leq t \leq 4$ with initial values of $\omega_x = 7$, $\omega_y = 0.2$, $\omega_z = 0.2$ is provided below. The output from this program includes time, kinetic energy, and various measures of angular momentum, i.e., t , ω_x , ω_y , ω_z , H_x , H_y , H_z , $H_{mag} \triangleq |\vec{H}|$, and K .

```
% File: SpinningBookODE.al (solve coupled odes)
%-----
Constant  Ixx = 1 kg*m^2, Iyy = 2 kg*m^2, Izz = 3 kg*m^2
Variable  wx', wy', wz'
wx' = ( (Iyy - Izz)*wz*wy ) / Ixx
wy' = ( (Izz - Ixx)*wx*wz ) / Iyy
wz' = ( (Ixx - Iyy)*wy*wx ) / Izz
%-- Angular momentum and rotational kinetic energy --
Hx = Ixx*wx; Hy = Iyy*wy; Hz = Izz*wz
Hmag = sqrt( Hx^2 + Hy^2 + Hz^2 )
K = 1/2*(Ixx*wx^2 + Iyy*wy^2 + Izz*wz^2)
%-----
Input  wx = 7.0, wy = 0.2, wz = 0.2, tFinal = 4 sec
Output t, wx, wy, wz, Hx, Hy, Hz, Hmag, K Joules
ODE() SpinningBook
Save  SpinningBookODE.all
Quit
```



¹To produce a MATLAB[®] file to solve these ODEs, change the ODE command to `ODE() SpinningBook.m`

5.9 Matrix calculations with MotionGenesis



Note: More at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Matrices and matrix commands.

Matrices in MotionGenesis

```
RowMatrix      = [ 1, 2, 3 ]
ColumnMatrix = [ 1; 2; 3 ]
MatrixWithTwoRowsAndThreeColumns = [ 1, 2, 3; 4, 5, pi ]
MatrixWithThreeRowsAndTwoColumns = [ 1, 2; 3, 4; 5, pi ]
```

Matrix addition

```
A = [ 1, 2, 3; 4, 5, 6 ]
B = [ 7, 8, 9; pi, i, t ]
AddMatrices = A + B
```

Multiplication of a matrix with a scalar

```
ScalarMultiplicationExample = 7 * [ 1, 2, 3; 4, 5, 6 ]
```

Multiplication of two matrices

```
A = [ 11, 12, 13; 21, 22, 23 ]
B = [ 11, 12; 21, 22; 31, 32 ]
C = A * B
```

The zero matrix and identity matrix in MotionGenesis

```
A = GetZeroMatrix( 3 )           % 3x3 matrix of zeros
B = GetZeroMatrix( 2, 3 )        % 2x3 matrix of zeros
C = GetIdentityMatrix( 3 )       % 3x3 identity matrix
D = GetIdentityMatrix( 2, 3 )    % 2x3 matrix with 1 along the diagonal and 0 elsewhere
```

Partial and ordinary derivative of a matrix with MotionGenesis

```
Variables x, y, z
A = [ x^2; x*sin(y); exp(x)*cosh(y) ]
PartialDerivativeOfAWithRespectToX = D( A, x )
PartialDerivativeOfAWithRespectToXandY = D( A, [x,y] )
```

Transpose of a matrix with MotionGenesis

```
A = [ 1, 2, 3; 4, 5, 6 ]
B = GetTranspose( A )
```

Submatrices and rows or columns with MotionGenesis

```
A = [1, 2, 3, 4; 5, 6, 7, 8; 9, 10, 11, 12]
B = GetRows( A, 2 )           % 1x4 matrix with row 2 of A
C = GetRows( A, 3,1 )         % 2x4 matrix with row 3 and row 1 of A
D = GetRows( A, 1:3, 3:2 )    % 5x4 matrix with rows 1 to 3 and rows 3 to 2 of A
F = GetColumn( A, 2 )         % 3x1 matrix with column 2 of A
G = GetColumns(A, 2:4)        % 3x3 matrix with columns 2 to 4 of A
H = GetColumns( GetRows(A,2:3), 2 ) % 1x2 matrix with elements 2,2 and 3,2 of A
```

Determinant and inverse of a matrix with MotionGenesis

```
A = [ 1, 2, 3; 4, 5, 6; 7, 8, 9 ]
DeterminantOfA = GetDeterminant( A )
InverseOfA = GetInverse( A )
```

Solving linear algebraic equations with MotionGenesis (symbolic or numerical)

```
Variable x1, x2, x3
Constant b1, b2, b3
Zero[1] = 2*x1 + 3*x2 + 4*x3 - b1
Zero[2] = 3*x1 + 4*x2 + 5*x3 - b2
Zero[3] = 6*x1 + 7*x2 + 9*x3 - b3
Solve( Zero = 0, x1, x2, x3 )
```

Forming matrices from linear algebraic equations with MotionGenesis

```
Constant b1, b2, b3
Variable x1, x2, x3
Zero[1] = 2*x1 + 3*x2 + 4*x3 - b1
Zero[2] = 3*x1 + 4*x2 + 5*x3 - b2
Zero[3] = 6*x1 + 7*x2 + 9*x3 - b3
CoefficientMatrix = D( Zero, [x1, x2, x3] ) % Forms 3x3 matrix
RemainderMatrix = Exclude( Zero, [x1, x2, x3] ) % Forms [-b1; -b2; -b3]
```

Eigenvalues and eigenvectors with MotionGenesis

```
A = [ 1, 2, 3; 4, 5, 6; 7, 8, 9 ]
eigenValuesOfA = GetEigen( A, eigenVectorsOfA )
eigenVector1 = GetColumn( eigenVectorsOfA, 1 )
eigenVector3 = GetColumn( eigenVectorsOfA, 3 )
```



5.10 Miscellaneous mathematical examples



Note: MotionGenesis file at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Sample Mathematics.

```
(1) % MotionGenesis file: MiscMathExamples.txt
(2) % Copyright (c) 2009-20 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) Variable    x', y'                                % Declare variables x, x', y, y'.
(5) Constant    a, b, c, d                            % Declare constants a, b, c, d.
(6) SetImaginaryNumber( i )                          % Declare i = sqrt(-1).
(7) %-----
(8) E = (x+2*y)^2 + 3*(7+x)*(x+y)                    % Create an expression.
-> (9) E = (x+2*y)^2 + 3*(7+x)*(x+y)

(10) Expand( E, 1:2 )                                % Clear parentheses.
-> (11) E = 21*x + 21*y + 4*x^2 + 4*y^2 + 7*x*y

(12) Factor( E, x )                                  % Factor on x.
-> (13) E = 21*y + 4*y^2 + 4*x*(5.25+x+1.75*y)

(14) %-----
(15) %          Mathematical commands.
(16) Dy = D( E, y )                                  % Partial derivative wrt. y
-> (17) Dy = 21 + 7*x + 8*y

(18) Dt = Dt( E )                                    % Total derivative wrt. t
-> (19) Dt = 21*y' + 8*y*y' + 7*(3+y)*x' + 7*x*(y'+1.142857*x')

(20) Ty = GetTaylorSeries( x*cos(y), 0:7, x=0,y=0)
-> (21) Ty = 0.001388889*x*(720+30*y^4-360*y^2-y^6)

(22) F = Evaluate( Ty, x=1, y=0.5 )                  % Symbolic/numerical evaluation
-> (23) F = 0.8775825

(24) Poly = GetPolynomial( [a,b,c], x )              % Creates a*x^2 +b*x +c
-> (25) Poly = c + b*x + a*x^2

(26) Root1 = GetRoots( [1; 2; 3; 4] )                % Roots of x^3 + 2*x^2 + 3*x + 4 = 0
-> (27) Root1 = [-1.650629; -0.1746854 - 1.546869*i; -0.1746854 + 1.546869*i]

(28) Root2 = GetQuadraticRoots( Poly = 0, x )
-> (29) Root2[1] = -0.5*(b+sqrt(b^2-4*a*c))/a
-> (30) Root2[2] = -0.5*(b-sqrt(b^2-4*a*c))/a

(31) %-----
(32) %          Creating row or column matrices.
(33) RowMatrix = [1, 2, 3, 4]                        % Create a 1x4 matrix
-> (34) RowMatrix = [1, 2, 3, 4]

(35) ColMatrix = [1; 2; 3; 4]                        % Create a 4x1 matrix
-> (36) ColMatrix = [1; 2; 3; 4]

(37) Zero[1] = a*x' + b*y' - 1                       % Assign elements of column matrix
-> (38) Zero[1] = -1 + a*x' + b*y'

(39) Zero[2] = c*x' + d*y' - Pi
-> (40) Zero[2] = -3.141593 + c*x' + d*y'

(41) %-----
(42) %          Solve a set of linear equations.
(43) Solve( Zero = 0, x', y' )
-> (44) x' = -(pi*b-d)/(a*d-b*c)
-> (45) y' = (pi*a-c)/(a*d-b*c)

(46) %-----
(47) %          Creating rectangular matrices.
```



```

(48) M0 = [a, b; c, 0] % Create a 2x2 matrix
-> (49) M0 = [a, b; c, 0]

(50) M0[2,2] := d % Assign element of rectangular matrix
-> (51) M0[2,2] = d

(52) M1 = [M0, [1,2; 3,4] ] % Elements of matrices can be matrices
-> (53) M1 = [a, b, 1, 2; c, d, 3, 4]

(54) %-----
(55) % Matrix commands.
(56) M2 = M0 + M0 % Matrix addition
-> (57) M2 = [2*a, 2*b; 2*c, 2*d]

(58) M3 = M0 * M0 % Matrix multiplication
-> (59) M3 = [a^2 + b*c, b*(a+d); c*(a+d), b*c + d^2]

(60) M4 = GetTranspose( M0 )
-> (61) M4 = [a, c; b, d]

(62) M5 = GetInverse( M0 )
-> (63) M5[1,1] = d/(a*d-b*c)
-> (64) M5[1,2] = -b/(a*d-b*c)
-> (65) M5[2,1] = -c/(a*d-b*c)
-> (66) M5[2,2] = a/(a*d-b*c)

(67) M6 = GetIdentityMatrix( 3 )
-> (68) M6 = [1, 0, 0; 0, 1, 0; 0, 0, 1]

(69) M7 = GetZeroMatrix( 3, 3 )
-> (70) M7 = [0, 0, 0; 0, 0, 0; 0, 0, 0]

(71) M8 = GetDiagonalMatrix(3,4, 5 )
-> (72) M8 = [5, 0, 0, 0; 0, 5, 0, 0; 0, 0, 5, 0]

(73) C0 = GetColumns( M0, 1 ) % Returns column 1 of M0
-> (74) C0 = [a; c]

(75) R0 = GetRows( M0, 2, 1:2) % Returns rows 2 and rows 1 through 2 of M0
-> (76) R0 = [c, d; a, b; c, d]

(77) N1 = GetRows( M0 ) % Returns the number of rows in M0
-> (78) N1 = 2

(79) det = GetDeterminant( M0 )
-> (80) det = a*d - b*c

(81) M9 = Evaluate( M0, a=1, b=2, c=3, d=4 )
-> (82) M9 = [1, 2; 3, 4]

(83) Lambda = GetEigen( M9 ) % Eigenvalues
-> (84) Lambda = [-0.3722813; 5.372281]

(85) EigValue = GetEigen( M9, EigVec) % Eigenvalues/eigenvectors
-> (86) EigVec = [-0.8245648, -0.4159736; 0.5657675, -0.9093767]
-> (87) EigValue = [-0.3722813; 5.372281]

```




Chapter 6

Computing with vectors

6.1 MotionGenesis vector commands



Command	Description
Cross(a>, b>)	Returns $\vec{a} \times \vec{b}$
Dot(a>, b>)	Returns $\vec{a} \cdot \vec{b}$
GetMagnitude(v>)	Returns $ \vec{v} $
GetMagnitudeSquared(v>)	Returns $ \vec{v} ^2$
GetUnitVector(v>)	Returns $\vec{v} / \vec{v} $
GetAngleBetweenVectors(a>, b>)	Returns the angle between vectors \vec{a} and \vec{b}
GetAngleBetweenVectors(Ax>, By>)	Returns the angle between unit vectors \hat{A}_x and \hat{B}_y
Vector(A, x, y, z)	Returns the vector $x\hat{A}_x + y\hat{A}_y + z\hat{A}_z$
Vector(A, [x, y, z])	Returns the vector $x\hat{A}_x + y\hat{A}_y + z\hat{A}_z$

MotionGenesis vector dot-products and cross products

The figure below shows a rigid frame N with right-handed orthogonal unit vectors \hat{n}_x , \hat{n}_y , \hat{n}_z . The vectors \vec{u} , \vec{v} , \vec{w} are defined as shown right.

The following shows calculations of $\vec{u} \cdot \vec{v}$, $\vec{u} \cdot \vec{w}$, $\vec{u} \times \vec{v}$, $\vec{v} \times \vec{w}$.

$$\begin{aligned}\vec{u} &= 2\hat{n}_x + 3\hat{n}_y + 4\hat{n}_z \\ \vec{v} &= x\hat{n}_x + y\hat{n}_y + z\hat{n}_z \\ \vec{w} &= 5\hat{n}_x - 6\hat{n}_y + 7\hat{n}_z\end{aligned}$$

```
(1) % File: CalculateDotCrossProductsWithBasis.txt
(2) %-----
(3) RigidBodyFrame N
(4) Variable x, y, z
(5) u> = x*Nx> + y*Ny> + z*Nz>
-> (6) u> = x*Nx> + y*Ny> + z*Nz>

(7) v> = 2*Nx> + 3*Ny> + 4*Nz>
-> (8) v> = 2*Nx> + 3*Ny> + 4*Nz>

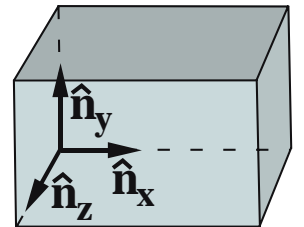
(9) w> = 5*Nx> + 6*Ny> + 7*Nz>
-> (10) w> = 5*Nx> + 6*Ny> + 7*Nz>

(11) uDotv = Dot( u>, v> )
-> (12) uDotv = 2*x + 3*y + 4*z

(13) uDotw = Dot( u>, w> )
-> (14) uDotw = 5*x + 6*y + 7*z

(15) uCrossv> = Cross( u>, v> )
-> (16) uCrossv> = (4*y-3*z)*Nx> + (2*z-4*x)*Ny> + (3*x-2*y)*Nz>

(17) vCrossw> = Cross( v>, w> )
-> (18) vCrossw> = -3*Nx> + 6*Ny> - 3*Nz>
```



Calculating angles with MotionGenesis

The figure to the right shows a rectangular parallelepiped (block) of sides 2, 3, 4 with points A, B, C located at corners. Right-handed orthogonal unit vectors $\hat{n}_x, \hat{n}_y, \hat{n}_z$ are directed with \hat{n}_x from B to C and \hat{n}_y from B to A .

The following commands calculate the angle between lines \overline{AB} and \overline{AC} .

```
(1) % File: DotProductsToCalculateAngles.txt
(2) %-----
(3) RigidBody N
(4) Point      A, B, C, D
(5) B.SetPosition( A, -2*Ny> )
-> (6) P_A_B> = -2*Ny>

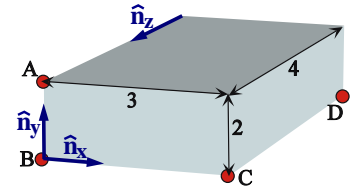
      (7) C.SetPosition( B, 3*Nx> )
-> (8) P_B_C> = 3*Nx>

      (9) distanceFromAToC = C.GetDistance(A)
-> (10) distanceFromAToC = 3.605551

      (11) u> = C.GetPosition(A) / distanceFromAToC
-> (12) u> = 0.8320503*Nx> - 0.5547002*Ny>

      (13) angleBACRad = GetAngleBetweenVectors( B.GetPosition(A), C.GetPosition(A) )
-> (14) angleBACRad = 0.9827937

      (15) angleBACDeg = angleBACRad * GetConversionFactor( radians, degrees )
-> (16) angleBACDeg = 56.30993
```

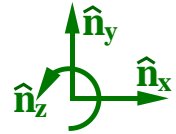


6.2 MotionGenesis vector operations

Given right-handed orthogonal unit vectors $\hat{n}_x, \hat{n}_y, \hat{n}_z$ and vectors \vec{v} and \vec{w} , perform the following calculations.

$$\vec{v} = 2\hat{n}_x + 3\hat{n}_y + 4\hat{n}_z$$

$$\vec{w} = 5\hat{n}_x - 6\hat{n}_y + 7\hat{n}_z$$



- a $10 * \vec{v} / 2 = 10\hat{n}_x + 15\hat{n}_y + 20\hat{n}_z$
- b $\vec{v} + \vec{w} = 7\hat{n}_x - 3\hat{n}_y + 11\hat{n}_z$
- c $\vec{v} \times \vec{w} = 45\hat{n}_x + 6\hat{n}_y - 27\hat{n}_z$
- d $\vec{v} \cdot \vec{w} = 20$
- e $\vec{v} \cdot \vec{v} = 29$
- f $|\vec{v}| = \sqrt{29} = 5.38516$
- g $\vec{v}^2 = 29$
- h $\frac{\vec{v}}{|\vec{v}|} = \frac{2\hat{n}_x + 3\hat{n}_y + 4\hat{n}_z}{\sqrt{29}}$
- i $\angle(\vec{v}, \vec{w}) = 1.20884$ radians or 69.2613°
- j $\hat{n}_x * \vec{v} = 2\hat{n}_x\hat{n}_x + 3\hat{n}_x\hat{n}_y + 4\hat{n}_x\hat{n}_z$
- k $\vec{v} * \hat{n}_x = 2\hat{n}_x\hat{n}_x + 3\hat{n}_y\hat{n}_x + 4\hat{n}_z\hat{n}_x$

```
% File: VectorCalculations.txt
%-----
RigidBody N
v> = 2*Nx> + 3*Ny> + 4*Nz>
w> = 5*Nx> - 6*Ny> + 7*Nz>
%-----
a> = 10 * v> / 2
b> = v> + w>
c> = Cross( v>, w> )
d = Dot( v>, w> )
e = Dot( v>, v> )
f = GetMagnitude( v> )
g = GetMagnitudeSquared( v> )
h> = GetUnitvector( v> )
i = GetAngleBetweenVectors( v>, w> )
j>> = Nx> * v>
k>> = v> * Nx>
%-----
Save VectorCalculations.all
Quit
```

Note: More at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow Vectors and vector commands.

6.3 MotionGenesis position vector commands

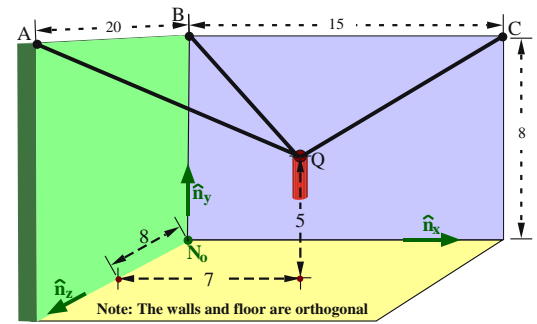


Command	Description and associated formula
<code>Q.GetPosition(No)</code>	Gets Q 's position from N_o , i.e., \vec{r}^{Q/N_o} .
<code>Q.SetPosition(P, posVector)</code>	Sets Q 's position from P , i.e., $\vec{r}^{Q/P} = \text{posVector}$
<code>Q.GetDistance(P)</code>	Gets Q 's distance from P , i.e., $ \vec{r}^{Q/P} $.

6.3.1 MotionGenesis: Microphone cable lengths, angles, and area (orthogonal walls)

The following MotionGenesis commands determine: L_A (the length of the cable joining A and Q); the angle ϕ between line AQ and line AB ; the surface area $|\Delta|$ of the triangle formed by points A, B, Q ; and a unit vector \hat{u} perpendicular to the surface area.

```
% File: MicrophoneCableLengthsOrthogonalWalls.al
%-----
RigidFrame N          % Back Wall
Point    A, B, C      % Points attached to walls
Particle  Q            % Microphone attached to cables
%-----
%      Given position vectors
B.SetPosition( No, 8*Ny> )
C.SetPosition( B, 15*Nx> )
A.SetPosition( B, 20*Nz> )
Q.SetPosition( No, 7*Nx> + 5*Ny> + 8*Nz> )
%-----
LA = Q.GetDistance( A )
%-----
phiRadians = GetAngleBetweenVectors( Q.GetPosition(A), B.GetPosition(A) )
phiDegrees = phiRadians * ConvertUnits( radian, degree )
%-----
AreaABQ> = 1/2 * Cross( A.GetPosition(B), Q.GetPosition(B) )
AreaABQ = GetMagnitude( AreaABQ> )
UnitVectorPerpendicularToTriangleABQ> = GetUnitVector( AreaABQ> )
Save MicrophoneCableLengthsOrthogonalWalls.all
Quit
```



6.3.2 MotionGenesis: Position vectors and geometry (single basis)

Each figure shows a laser A pointing at an object with a beam that passes through a point A_o in the direction of $3\hat{n}_x - \hat{n}_y + \hat{n}_z$ where $\hat{n}_x, \hat{n}_y, \hat{n}_z$ are right-handed orthogonal unit vectors fixed in a flat horizontal plane N with \hat{n}_x horizontally-right and \hat{n}_y vertically-upward. Point A_o 's position vector from N_o (a point fixed in N) is $\vec{r}^{A_o/N_o} = \hat{n}_x + 2\hat{n}_y$ (all lengths are in meters).

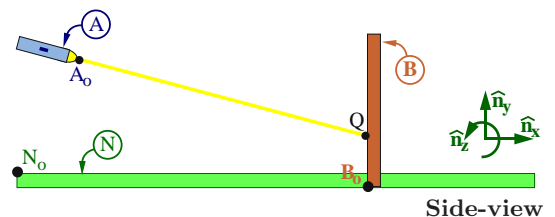
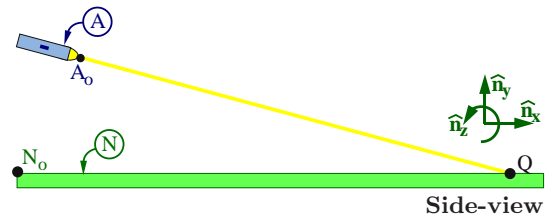
Consider a laser beam that hits point Q of N .

Find the distance from A_o to Q .

Result: $|\vec{r}^{Q/A_o}| = 6.63 \text{ m}$

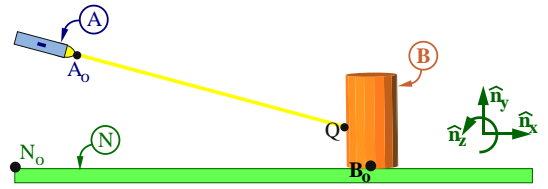
A laser beam hits point Q of a vertical wall B that is in the \hat{n}_y - \hat{n}_z plane (the wall is perpendicular to \hat{n}_x). The wall is 5 m from N_o and its lower edge is parallel to \hat{n}_z . Find the distance from A_o to Q .

Result: $|\vec{r}^{Q/A_o}| = 4.42 \text{ m}$



A laser beam hits point Q of a vertical cylinder B of radius 1 m. The point of B 's symmetry axis in contact with N is denoted B_o and its position from N_o is $\vec{r}^{B_o/N_o} = 5\hat{n}_x + 2\hat{n}_z$. Find the distance from A_o to Q .

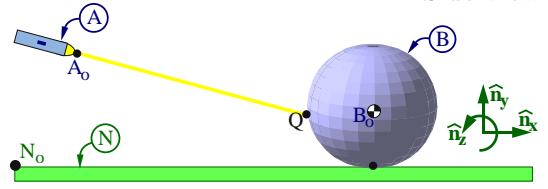
Result: $|\vec{r}^{Q/A_o}| = \boxed{3.83} \text{ m}$



Side-view

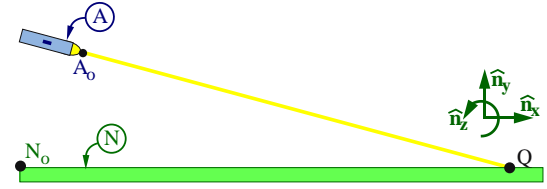
A laser beam hits point Q of a sphere B of radius 1 m. The position to the point of B in contact with N is $5\hat{n}_x + 2\hat{n}_z$. Find the distance from A_o to Q .

Result: $|\vec{r}^{Q/A_o}| = \boxed{3.85} \text{ m}$

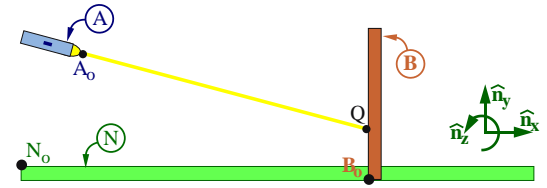


Side-view

```
% MotionGenesis file: MGLaserBeamOnHorizontalPlane.txt
% Copyright (c) 2016 Motion Genesis LLC. All rights reserved.
%-----
RigidFrame N % Horizontal plane.
RigidFrame A % Laser (also creates point Ao).
Point Q % Point of beam on horizontal plane.
%-----
Ao.SetPosition( No, Nx> + 2*Ny> ) % Given info.
LaserDirection> = 3*Nx> - Ny> + Nz> % Given info.
%-----
Variable d % Introduce unknown to write position vector.
Q.SetPosition( Ao, d * GetUnitVector( LaserDirection> ) )
%-----
ZeroVertical = Dot( Q.GetPosition(No), Ny> )
Solve( ZeroVertical = 0, d )
%-----
Save MGLaserBeamOnHorizontalPlane.html
Quit
```



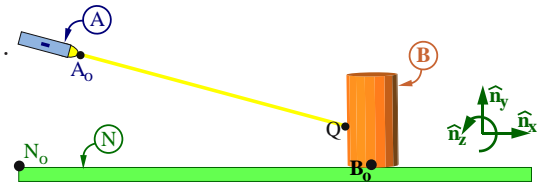
```
% MotionGenesis file: MGLaserBeamOnVerticalWall.txt
% Copyright (c) 2016 Motion Genesis LLC. All rights reserved.
%-----
RigidFrame N % Horizontal plane.
RigidFrame A % Laser (also creates point Ao).
RigidFrame B % Vertical wall (also creates point Bo).
Point Q % Point of beam that hits vertical wall.
%-----
% Known position vector and direction of laser.
Ao.SetPosition( No, Nx> + 2*Ny> )
LaserDirection> = 3*Nx> - Ny> + Nz>
%-----
% Position vector to point Bo at base of vertical wall.
Bo.SetPosition( No, 5*Nx> )
%-----
% Introduce unknowns to be able to write position vectors.
Variable Lambda, y, z
Q.SetPosition( Ao, Lambda * LaserDirection> )
Q.SetPosition( Bo, y*Ny> + z*Nz> )
%-----
% Solve for unknowns and determine Q's distance from Ao.
Loop> = P_No_Ao> + P_Ao_Q> + P_Q_Bo> + P_Bo_No>
Zero[1] = Dot( Loop>, Nx> )
Zero[2] = Dot( Loop>, Ny> )
Zero[3] = Dot( Loop>, Nz> )
Solve( Zero, Lambda, y, z )
DistanceBetweenQAndAo = Explicit( Q.GetDistance(Ao) )
%-----
Save MGLaserBeamOnVerticalWall.html
Quit
```



```

% MotionGenesis file: MGLaserBeamOnVerticalCylinder.txt
% Copyright (c) 2016-18 Motion Genesis LLC. All rights reserved.
%-----
RigidFrame N    % Horizontal plane.
RigidFrame A    % Laser (also creates point Ao).
RigidFrame B    % Vertical cylinder (also creates point Bo).
Point    Q      % Point of beam that hits vertical cylinder.
%-----
%      Known position vector and direction of laser.
Ao.SetPosition( No, Nx> + 2*Ny> )
LaserDirection> = 3*Nx> - Ny> + Nz>
%-----
%      Position vector to point Bo from No.
Bo.SetPosition( No, 5*Nx> + 2*Nz> )
%-----
%      Introduce unknowns to be able to write position vectors.
Variable    Lambda, x, y, z
Q.SetPosition( Ao, Lambda * LaserDirection> )
Q.SetPosition( Bo, x*Nx> + y*Ny> + z*Nz> )
%-----
%      Solve for unknowns and determine Q's distance from Ao.
Loop> = P_No_Ao> + P_Ao_Q> + P_Q_Bo> + P_Bo_No>
Zero[1] = Dot( Loop>, Nx> )
Zero[2] = Dot( Loop>, Ny> )
Zero[3] = Dot( Loop>, Nz> )
Solve( Zero, Lambda, y, z )
EqnOfVerticalCylinderOfRadius1 = x^2 + z^2 - 1
SolveQuadraticNegativeRoot( EqnOfVerticalCylinderOfRadius1 = 0, x )
DistanceBetweenQAndAo = Explicit( Q.GetDistance(Ao) )
%-----
Save MGLaserBeamOnVerticalCylinder.html
Quit

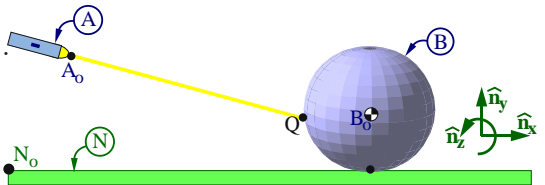
```



```

% MotionGenesis file: MGLaserBeamOnSphere.txt
% Copyright (c) 2015-18 Motion Genesis LLC. All rights reserved.
%-----
RigidFrame N    % Horizontal plane.
RigidFrame A    % Laser (also creates point Ao).
RigidFrame B    % Sphere (also creates point Bo).
Point    Q      % Point of beam that hits sphere.
%-----
%      Known position vector and direction of laser.
Ao.SetPosition( No, Nx> + 2*Ny> )
LaserDirection> = 3*Nx> - Ny> + Nz>
%-----
%      Position vector to point Bo (center of sphere) from No.
Bo.SetPosition( No, 5*Nx> + Ny> + 2*Nz> )
%-----
%      Introduce unknowns to write position vectors.
Variable    Lambda, x, y, z
Q.SetPosition( Ao, Lambda * LaserDirection> )
Q.SetPosition( Bo, x*Nx> + y*Ny> + z*Nz> )
%-----
%      Solve for unknowns and determine Q's distance from Ao.
Loop> = P_No_Ao> + P_Ao_Q> + P_Q_Bo> + P_Bo_No>
Zero[1] = Dot( Loop>, Nx> )
Zero[2] = Dot( Loop>, Ny> )
Zero[3] = Dot( Loop>, Nz> )
Solve( Zero, Lambda, y, z )
EqnOfSphereOfRadius1 = x^2 + y^2 + z^2 - 1
SolveQuadraticNegativeRoot( EqnOfSphereOfRadius1 = 0, x )
DistanceBetweenQAndAo = Explicit( Q.GetDistance(Ao) )
%-----
Save MGLaserBeamOnSphere.html
Quit

```



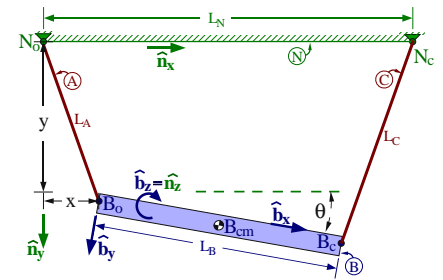
6.4 MotionGenesis rotation commands



Command	Description
B.GetRotationMatrix(A)	Gets the ${}^B R^A$ rotation matrix
B.SetRotationMatrixX(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $\hat{\mathbf{b}}_x = \hat{\mathbf{a}}_x$ by an angle θ
B.SetRotationMatrixY(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $\hat{\mathbf{b}}_y = \hat{\mathbf{a}}_y$ by an angle θ
B.SetRotationMatrixZ(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $\hat{\mathbf{b}}_z = \hat{\mathbf{a}}_z$ by an angle θ
B.SetRotationMatrixNegativeX(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $-\hat{\mathbf{b}}_x = -\hat{\mathbf{a}}_x$ by an angle θ
B.SetRotationMatrixNegativeY(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $-\hat{\mathbf{b}}_y = -\hat{\mathbf{a}}_y$ by an angle θ
B.SetRotationMatrixNegativeZ(A, theta)	Assigns ${}^B R^A$ for B rotated in A about $-\hat{\mathbf{b}}_z = -\hat{\mathbf{a}}_z$ by an angle θ
B.SetRotationMatrix(A, Sequence, ...)	Assigns ${}^B R^A$ using a Bodyijk or Spaceijk rotation ($i, j, k = x, y, z$)
B.SetRotationMatrix(A, Quaternion, ...)	Assigns ${}^B R^A$ using a quaternion
B.SetRotationMatrix(A, EulerParameters, ...)	Assigns ${}^B R^A$ using Euler parameters (quaternion)
B.SetRotationMatrix(A, RodriguesParameters, ...)	Assigns ${}^B R^A$ using Rodrigues parameters
B.SetRotationMatrix(A, [Rxx, Rxy, Rxz; ...])	Assigns ${}^B R^A$ directly from a rotation matrix
B.RotateX(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.RotateY(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.RotateZ(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.RotateNegativeX(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.RotateNegativeY(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.RotateNegativeZ(A, theta)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.Rotate(A, Sequence, ...)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.Rotate(A, Quaternion, ...)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.Rotate(A, EulerParameters, ...)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.Rotate(A, RodriguesParameters, ...)	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.
B.Rotate(A, [Rxx, Rxy, Rxz; ...])	Assigns ${}^B R^A$ and possibly ${}^A \vec{\omega}^B$, ${}^A \vec{\alpha}^B$, etc.

6.4.1 Cable lengths to position a beam (with rotation matrix).

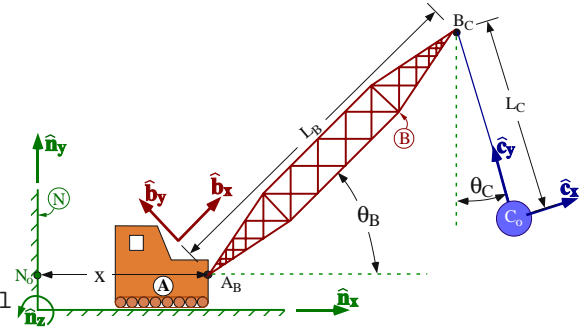
```
% File: MGBeamOnTwoCablesKinematics.txt
% Problem: Beam position/orientation ... from cable lengths.
% Note: MotionGenesis script with student blanks at
%       www.MotionGenesis.com -> TextbookResources
%-----
NewtonianFrame N      % Nx> horizontally right, Ny> down.
RigidBody B           % Beam with Bx> pointed from Bo to Bc.
Point Nc(N)          % Point of N attached to cable C.
Point Bc(B)          % Point of B attached to cable C.
%-----
Constant LN = 6 m     % Distance between No and Nc.
Constant LB = 4 m     % Distance between Bo and Bc.
Constant LA = 2.7 m   % Length of cable A.
Constant LC = 3.7 m   % Length of cable C.
Variable x'', y''     % Nx> and Ny> measures of Bo's position from No.
Variable q''          % Bz> measure of angle from Nx> to Bx>.
%-----
% Rotation: Beam B rotates relative to N about Bz> = Nz> by angle q.
B.RotateZ( N, q )
%-----
% Set point Nc's position from No to LN*Nx>. Similarly for Bo, Bc.
Nc.SetPosition( No, LN*Nx> )
Bo.SetPosition( No, x*Nx> + y*Ny> )
Bc.SetPosition( Bo, LB*Bx> )
%-----
% Constraints arising from rope lengths ("just geometry").
Eqn[1] = Bo.GetDistanceSquared( No ) - LA^2
Eqn[2] = Bc.GetDistanceSquared( Nc ) - LC^2
%-----
% Solve nonlinear equations (requires a numerical guess).
Input x = 1 meter, x' = 0.4 m/s
Solve( Eqn = 0, y = 3 meters, q = 20 deg )
%-----
% Differentiate constraint equations and solve for y' and q'.
Solve( Dt(Eqn) = 0, y' = 0 m/s, q' = 0 rad/sec )
%-----
% Calculate Bcm's (B's center of mass) velocity and its square.
Bcm.SetPosition( Bo, 0.5*LB*Bx> )
v> = Dt( Bcm.GetPosition(No), N )
vSquared = Dot( v>, v> )
vSquaredNumerical = EvaluateToNumber( vSquared )
%-----
% Calculate Bcm's acceleration in N.
a> = Dt( v>, N )
%-----
Save MGBeamOnTwoCablesKinematics.html
Quit
```



6.4.2 Calculating rotation matrices for a crane and wrecking ball with MotionGenesis

The following **MotionGenesis** commands calculate rotation matrices relating unit vectors $\hat{n}_x, \hat{n}_y, \hat{n}_z$ and $\hat{b}_x, \hat{b}_y, \hat{b}_z$ and $\hat{c}_x, \hat{c}_y, \hat{c}_z$.

```
% File: CraneAndWreckingBallRotationMatrices.al
%-----
RigidFrame N, B, C
Variable qB, qC
B.RotateZ( N, qB )
C.RotateZ( N, qC )
BC = B.GetRotationMatrix( C )
Save CraneAndWreckingBallRotationMatrices.all
Quit
```



```
(1) % File: CraneAndWreckingBallRotationMatrices.al
(2) %-----
(3) RigidFrame N, B, C
(4) Variable qB, qC
(5) B.RotateZ( N, qB )
-> (6) B_N = [cos(qB), sin(qB), 0; -sin(qB), cos(qB), 0; 0, 0, 1]

(7) C.RotateZ( N, qC )
-> (8) C_N = [cos(qC), sin(qC), 0; -sin(qC), cos(qC), 0; 0, 0, 1]

(9) BC = B.GetRotationMatrix( C )
-> (10) BC = [cos(qB-qC), sin(qB-qC), 0; -sin(qB-qC), cos(qB-qC), 0; 0, 0, 1]
```

6.4.3 Calculating rotation matrices for a chaotic double-pendulum with MotionGenesis

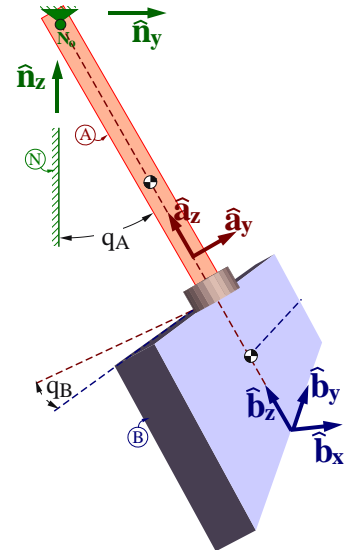
The following **MotionGenesis** commands calculate rotation matrices relating unit vectors $\hat{n}_x, \hat{n}_y, \hat{n}_z$ and $\hat{a}_x, \hat{a}_y, \hat{a}_z$ and $\hat{b}_x, \hat{b}_y, \hat{b}_z$.

Related: Sections 6.5 and 9.12 and www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Chaotic pendulum.

```
(1) % File: BabybootRotationMatrices.txt
(2) %-----
(3) RigidFrame N          % Reference frame
(4) RigidBody A           % Upper rod
(5) RigidBody B           % Lower plate
(6) Variable qA           % Pendulum angle
(7) Variable qB           % Plate angle
(8) %-----
(9) A.RotateX( N, qA )    % A rotates "about +x" in N by qA
-> (10) A_N = [1, 0, 0; 0, cos(qA), sin(qA); 0, -sin(qA), cos(qA)]

(11) B.RotateZ( A, qB )   % B rotates "about +z" in A by qB
-> (12) B_A = [cos(qB), sin(qB), 0; -sin(qB), cos(qB), 0; 0, 0, 1]

(13) B_N = B.GetRotationMatrix( N )
-> (14) B_N[1,1] = cos(qB)
-> (15) B_N[1,2] = sin(qB)*cos(qA)
-> (16) B_N[1,3] = sin(qA)*sin(qB)
-> (17) B_N[2,1] = -sin(qB)
-> (18) B_N[2,2] = cos(qA)*cos(qB)
-> (19) B_N[2,3] = sin(qA)*cos(qB)
-> (20) B_N[3,1] = 0
-> (21) B_N[3,2] = -sin(qA)
-> (22) B_N[3,3] = cos(qA)
```



6.4.4 Rotation matrices and vertical displacement of a bifilar pendulum.

Bifilar and trifilar pendulum are used to determine inertia properties of rigid bodies (e.g., aircraft, spacecraft, and biological structures such as humans limbs). The following shows a rigid human bone B suspended by two inextensible cables A_1 and A_2 , each of which is attached to a flat ceiling N .

- Cable A_1 attaches to the ceiling at point N_1 of N and to the bone at point B_1 of B .
- Cable A_2 attaches to the ceiling at point N_2 of N and to the bone at point B_2 of B .
- Point N_o of N is centered between N_1 and N_2 . Point B_o of B is centered between B_1 and B_2 .
- Point B_{cm} (B 's center of mass) and point B_o **always** lie directly below N_o .
- Initially, B_i lies directly below N_i ($i=1, 2$), respectively.
- B is rotated by an angle θ about the vertical line through B_o and N_o .

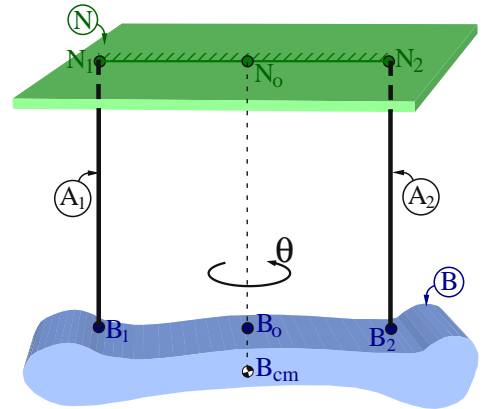
Relate y to L , h , and θ (defined in the following table).

Result: $y^2 + \frac{1}{2} L^2 [1 - \cos(\theta)] - h^2 = 0$

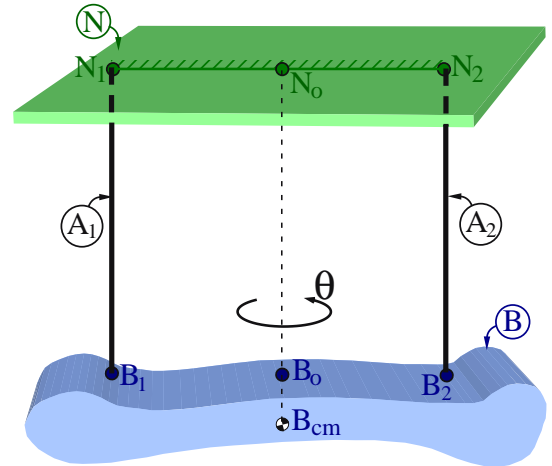
Calculate numerical values for y and \dot{y} (3 significant digits).

Description	Symbol	Value
Distance between N_1 and N_2	L	1 m
Distance between N_i and B_i ($i=1, 2$)	h	1 m
B 's rotation angle in N	θ	135°
B 's rotation rate in N	$\dot{\theta}$	$0.5 \frac{\text{rad}}{\text{sec}}$
Distance between N_o and B_o	y	0.383 m
Time-derivative of y	\dot{y}	-0.231 $\frac{\text{m}}{\text{s}}$

Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow 2D/3D geometry.



```
% MotionGenesis file: MGBifilarPendulumYInTermsOfTheta.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
RigidFrame  N
RigidBody    B
Point       N1(N), N2(N), B1(B), B2(B)
%-----
Constant    L = 1.0 m, h = 1.0 m
Variable    y'
Specified   theta'
%-----
B.RotateY( N, theta )
N1.SetPosition( No, -L/2*Nx> ) % Nx> points from No to N2.
Bo.SetPosition( No, -y*Ny> )  % Ny> = By> is vertical.
B1.SetPosition( Bo, -L/2*Bx> ) % Bx> points from Bo to B2.
%-----
zeroDistanceSquared = B1.GetDistanceSquared( N1 ) - h^2
SolveQuadraticPositiveRootDt( zeroDistanceSquared, y )
%-----
% Calculate y and y' for given values of theta, theta'.
yValue = EvaluateToNumber( y, theta = 135 deg )
yDtValue = EvaluateToNumber( y', y = yValue, theta = 135 deg, theta' = 0.5 rad/sec )
%-----
Save MGBifilarPendulumYInTermsOfTheta.html
Quit
```



6.4.5 Rotation matrices and four-bar linkage configuration.

The figure to the right is a planar four-bar linkage consisting of uniform rigid links A , B , C and ground N . Link A is connected with revolute joints to N and B at points N_A and A_B , respectively. Link C is connected with revolute joints to N and B at points C_N and B_C , respectively.

Right-handed orthogonal unit vectors $\hat{\mathbf{a}}_i$, $\hat{\mathbf{b}}_i$, $\hat{\mathbf{c}}_i$, $\hat{\mathbf{n}}_i$ ($i = x, y, z$) are fixed in A , B , C , N , with $\hat{\mathbf{a}}_x$ directed from N_A to A_B , $\hat{\mathbf{b}}_x$ from A_B to B_C , $\hat{\mathbf{c}}_x$ from C_N to B_C , $\hat{\mathbf{n}}_x$ vertically-downward, $\hat{\mathbf{n}}_y$ from N_A to C_N , and $\hat{\mathbf{a}}_z = \hat{\mathbf{b}}_z = \hat{\mathbf{c}}_z = \hat{\mathbf{n}}_z$ parallel to the axes of the revolute joints.

Create a vector “**loop equation**” using a sum of position vectors that start and end at point N_A .

Result:

$$L_A \hat{\mathbf{a}}_x + L_B \hat{\mathbf{b}}_x + (-L_C \hat{\mathbf{c}}_x) + (-L_N \hat{\mathbf{n}}_y) = \vec{0}$$

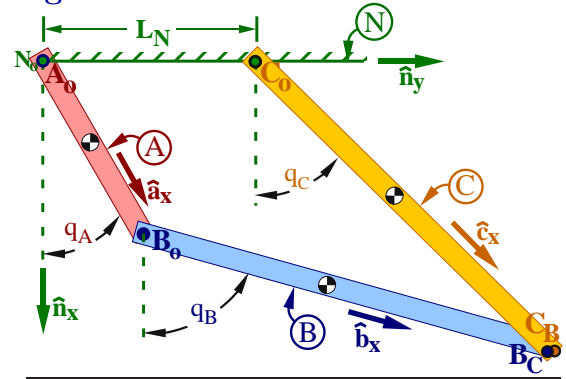
Dot the loop equation with $\hat{\mathbf{n}}_x$ and $\hat{\mathbf{n}}_y$ to create two equations $f_i = 0$ ($i = 1, 2$) that relate q_A , q_B , and q_C .¹ Next, determine values of q_B and q_C that satisfy these two equations when $q_A = 30^\circ$.

Result: Equations relating q_A , q_B , q_C .

Values when $q_A = 30^\circ$

$f_1 = L_A * \cos(q_A) + L_B * \cos(q_B) - L_C * \cos(q_C)$	$q_B = 74.4775^\circ$
$f_2 = L_A * \sin(q_A) + L_B * \sin(q_B) - L_C * \sin(q_C) - L_N$	$q_C = 45.5225^\circ$

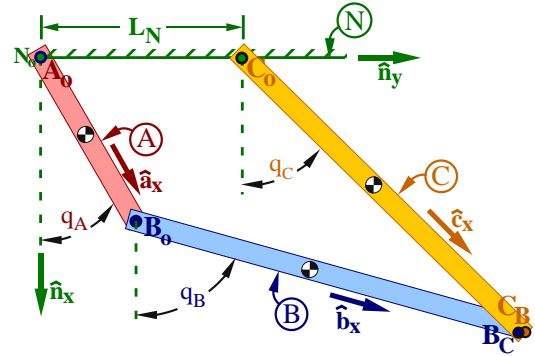
If $L_A < 1$ m, link A can be driven completely around, whereas if $L_A > 1$ m, it can only be driven 90° .



Quantity	Symbol	Value
Distance from N_A to A_B	L_A	1 m
Distance from A_B to B_C	L_B	2 m
Distance from B_C to C_N	L_C	2 m
Distance from C_N to N_A	L_N	1 m
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{a}}_x$	q_A	Variable
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{b}}_x$	q_B	Variable
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{c}}_x$	q_C	Variable

Engineering convention: Angles are drawn **positive**.

```
% File: FourBarConfiguration.al
% Problem: Solve for qB and qC from given value of qA
%-----
RigidBody A, B, C, N
Variable qA, qB, qC
Constant LA = 1 m, LB = 2 m, LC = 2 m, LN = 1 m
%-----
% Rotational kinematics
A.RotateZ( N, qA )
B.RotateZ( N, qB )
C.RotateZ( N, qC )
%-----
% Configuration constraints
Loop> = LA*Ax> + LB*Bx> - LC*Cx> - LN*Ny>
Loop[1] = Dot( Loop>, Nx> )
Loop[2] = Dot( Loop>, Ny> )
%-----
% Solve constraints with given constants, variables, etc.
Input qA = 30 deg
Solve( Loop, qB = 60 deg, qC = 20 deg )
%-----
% Save input together with program responses
Save FourBarConfiguration.all
Quit
```



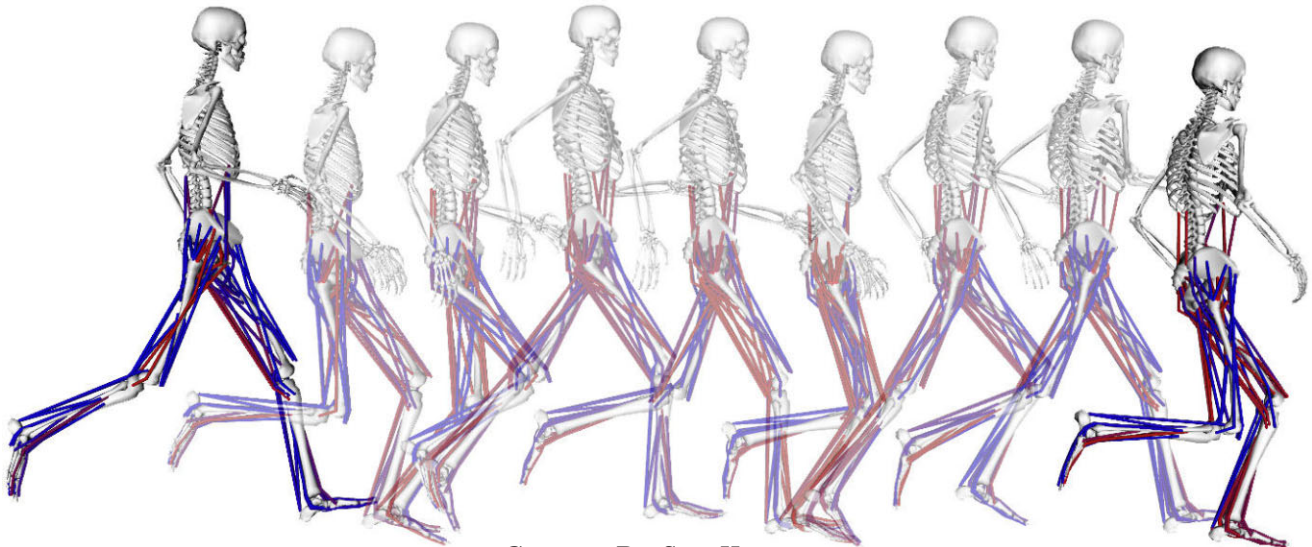
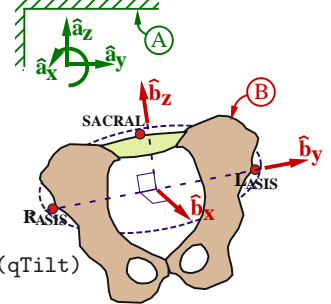
Related: Statics/dynamics in Section 9.20 and www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Four-bar linkage.

¹Dot-products can be calculated by definition (inspection of the figure) or with rotation matrices.

6.4.6 Calculating pelvis rotation matrices with MotionGenesis

The figure below shows two right-handed sets of orthogonal unit vectors, with $\hat{\mathbf{b}}_x$, $\hat{\mathbf{b}}_y$, $\hat{\mathbf{b}}_z$ fixed in a rigid pelvis B and $\hat{\mathbf{a}}_x$, $\hat{\mathbf{a}}_y$, $\hat{\mathbf{a}}_z$ fixed in an examination room A . The following MotionGenesis commands calculate the ${}^B R^A$ rotation matrix when $\hat{\mathbf{b}}_x$, $\hat{\mathbf{b}}_y$, $\hat{\mathbf{b}}_z$ is first aligned with $\hat{\mathbf{a}}_x$, $\hat{\mathbf{a}}_y$, $\hat{\mathbf{a}}_z$, respectively, and then B is subjected to right-handed successive-rotations characterized by $\theta_r \hat{\mathbf{b}}_z$, $\theta_o \hat{\mathbf{b}}_x$, and $\theta_t \hat{\mathbf{b}}_y$.

```
(1) % File: PelvisTiltObliquityRotation.al
(2) %-----
(3) RigidBody A          % Examination room
(4) RigidBody B          % Pelvis
(5) Variable qTilt      % Leaning forward is positive
(6) Variable qObliquity % Raising left-hip is positive
(7) Variable qRotation  % Rotating counter-clockwise is positive
(8) %-----
(9) B.Rotate( A, BodyXYZ, qTilt, qObliquity, qRotation )
-> (10) B_A[1,1] = cos(qRotation)*cos(qTilt) + sin(qObliquity)*sin(qRotation)*sin(qTilt)
-> (11) B_A[1,2] = sin(qRotation)*cos(qObliquity)
-> (12) B_A[1,3] = sin(qObliquity)*sin(qRotation)*cos(qTilt) - sin(qTilt)*cos(qRotation)
-> (13) B_A[2,1] = sin(qObliquity)*sin(qTilt)*cos(qRotation) - sin(qRotation)*cos(qTilt)
-> (14) B_A[2,2] = cos(qObliquity)*cos(qRotation)
-> (15) B_A[2,3] = sin(qRotation)*sin(qTilt) + sin(qObliquity)*cos(qRotation)*cos(qTilt)
-> (16) B_A[3,1] = sin(qTilt)*cos(qObliquity)
-> (17) B_A[3,2] = -sin(qObliquity)
-> (18) B_A[3,3] = cos(qObliquity)*cos(qTilt)
```



Courtesy Dr. Sam Hamner

6.5 MotionGenesis angular velocity and angular acceleration commands

Command	Description
A.SetAngularVelocity(N, theta'*Ax>)	Assigns A 's angular velocity in N . ${}^N\vec{\omega}^A = \dot{\theta} \hat{\mathbf{a}}_x$
B.GetAngularVelocity(N)	Calculates ${}^N\vec{\omega}^B$, B 's angular velocity in N
A.SetAngularAcceleration(N, theta''*Ax>)	Assigns A 's angular acceleration in N . ${}^N\vec{\alpha}^A = \ddot{\theta} \hat{\mathbf{a}}_x$
B.GetAngularAcceleration(N)	Calculates ${}^N\vec{\alpha}^B$, B 's angular acceleration in N
A.SetAngularVelocityAcceleration(N, theta'*Ax>)	Assigns ${}^N\vec{\omega}^A = \dot{\theta} \hat{\mathbf{a}}_x$ and ${}^N\vec{\alpha}^A = \ddot{\theta} \hat{\mathbf{a}}_x$
Note: N , A , and B have been named in a RigidBody, RigidBody, or NewtonianFrame declaration.	

Calculating angular velocity and angular acceleration with MotionGenesis

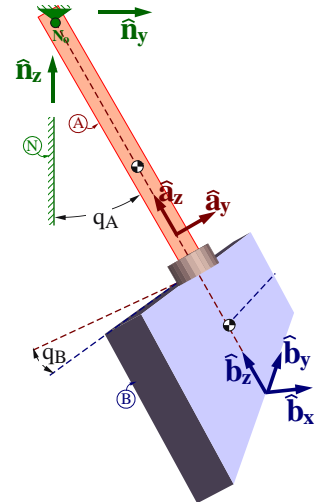
The following MotionGenesis commands calculate the angular velocity and angular acceleration of rigid bodies A and B in reference frame N .

```
(1) % File: BabybootAngularVelocityAngularAcceleration.txt
(2) %-----
(3) RigidBody N      % Reference frame
(4) RigidBody A      % Upper rod
(5) RigidBody B      % Lower plate
(6) Variable qA''    % Pendulum angle and its time-derivatives
(7) Variable qB''    % Plate angle and its time-derivative
(8) %-----
(9) %      Angular velocity and angular acceleration
(10) A.SetAngularVelocityAcceleration( N, qA'*Ax> )
-> (11) w_A_N> = qA'*Ax>
-> (12) alf_A_N> = qA''*Ax>

(13) B.SetAngularVelocityAcceleration( A, qB'*Az> )
-> (14) w_B_A> = qB'*Az>
-> (15) alf_B_A> = qB''*Az>

(16) w_B_N> = B.GetAngularVelocity( N )
-> (17) w_B_N> = qA'*Ax> + qB'*Az>

(18) alf_B_N> = B.GetAngularAcceleration( N )
-> (19) alf_B_N> = qA''*Ax> - qA'*qB'*Ay> + qB''*Az>
```



6.6 MotionGenesis: Calculating vector derivatives



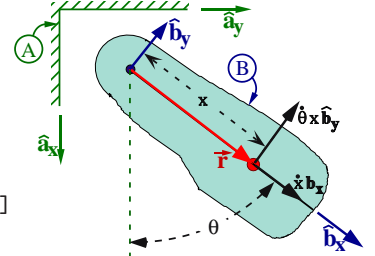
The **MotionGenesis** `Dt` and `D` commands calculate ordinary and partial derivatives. As shown below, `Dt` calculates the ordinary time-derivative of the vector $x \hat{\mathbf{b}}_x$ in rigid body B and in reference frame A .

```
(1) % File: VectorDerivativeExample.txt
(2) %-----
(3) RigidBody A      % Plane
(4) RigidBody B      % Rigid body
(5) Variable x'      % Declares x and its time-derivative x'
(6) Variable theta'  % Declares the angle theta and its time-derivative
(7) %-----
(8) B.RotateZ( A, theta )
-> (9) B_A = [cos(theta), sin(theta), 0; -sin(theta), cos(theta), 0; 0, 0, 1]
-> (10) w_B_A> = theta'*Bz>

(11) %-----
(12) r> = x*Bx>
-> (13) r> = x*Bx>

(14) DerivativeOfrInB> = Dt( r>, B )
-> (15) DerivativeOfrInB> = x'*Bx>

(16) DerivativeOfrInA> = Dt( r>, A )
-> (17) DerivativeOfrInA> = x'*Bx> + x*theta'*By>
```



6.6.1 Time-derivative of angular momentum with MotionGenesis

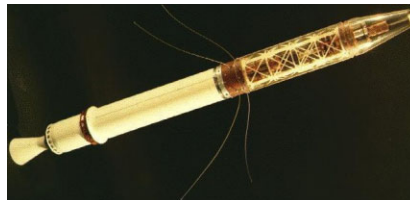
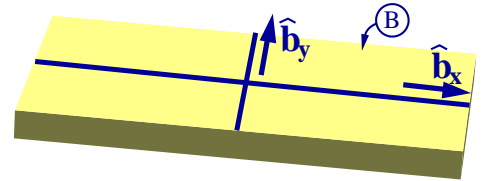
The following **MotionGenesis** commands calculate the time-derivative in reference frame N of the angular momentum of a spinning book (rigid body B) and creates an matrix with the $\hat{\mathbf{b}}_x$, $\hat{\mathbf{b}}_y$, $\hat{\mathbf{b}}_z$ measures.

```
(1) % File: DerivativeOfAngularMomentum.txt
(2) %-----
(3) NewtonianFrame N
(4) RigidBody B
(5) Variable wx', wy', wz'
(6) B.SetInertia( Bcm, Ixx, Iyy, Izz )
(7) %-----
(8) B.SetAngularVelocity( N, wx*Bx> + wy*By> + wz*Bz> )
-> (9) w_B_N> = wx*Bx> + wy*By> + wz*Bz>

(10) H> = Vector( B, Ixx*wx, Iyy*wy, Izz*wz )
-> (11) H> = Ixx*wx*Bx> + Iyy*wy*By> + Izz*wz*Bz>

(12) %-----
(13) Zero> = Dt( H>, N )
-> (14) Zero> = (Izz*wy*wz+Ixx*wx'-Iyy*wy*wz)*Bx> + (Ixx*wx*wz+Iyy*wy'-Izz*wx*
    wz)*By> + (Iyy*wx*wy+Izz*wz'-Ixx*wx*wy)*Bz>

(15) Zero = Matrix( B, Zero> )
-> (16) Zero[1] = Izz*wy*wz + Ixx*wx' - Iyy*wy*wz
-> (17) Zero[2] = Ixx*wx*wz + Iyy*wy' - Izz*wx*wz
-> (18) Zero[3] = Iyy*wx*wy + Izz*wz' - Ixx*wx*wy
```



6.6.2 Differential geometry: Ellipse circumference, area, normal, tangent with MotionGenesis

The following figure shows a point Q on the periphery of an ellipse B whose center is point B_o .

Right-handed orthogonal unit vectors $\hat{\mathbf{b}}_x, \hat{\mathbf{b}}_y, \hat{\mathbf{b}}_z$ are fixed in B with

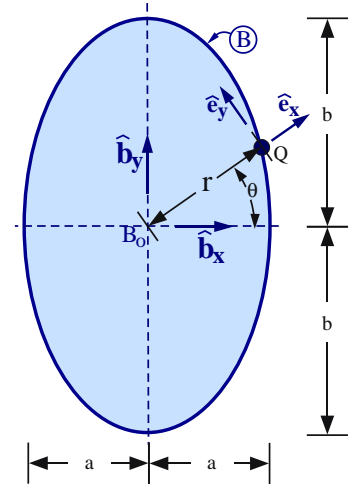
- $\hat{\mathbf{b}}_x$ horizontally-right and aligned with the ellipse's minor axis
- $\hat{\mathbf{b}}_y$ vertically-upward and aligned with the ellipse's major axis
- $\hat{\mathbf{b}}_z$ perpendicular to the ellipse.

Right-handed orthogonal unit vectors $\hat{\mathbf{e}}_x, \hat{\mathbf{e}}_y, \hat{\mathbf{e}}_z$ are initially directed with $\hat{\mathbf{e}}_i = \hat{\mathbf{b}}_i$ ($i = x, y, z$) and then are subjected to a right-handed rotation in B characterized by $\theta \hat{\mathbf{b}}_z$ so $\hat{\mathbf{e}}_x$ points from B_o to Q and $\hat{\mathbf{e}}_z = \hat{\mathbf{b}}_z$.

Description	Symbol	Type	Value
Half-diameter of ellipse minor axis	a	+ Constant	2
Half-diameter of ellipse major axis	b	+ Constant	4
Distance between N_o and Q	r	+ Variable	Varies
Angle from $\hat{\mathbf{b}}_x$ to $\hat{\mathbf{e}}_x$ with $+\hat{\mathbf{b}}_z$ sense	θ	Variable	Varies

When Q 's position from B_o is expressed in terms of scalars x and y as shown below-left, the equation of the ellipse can be written as shown below-right.

$$\vec{\mathbf{r}} \triangleq \vec{\mathbf{r}}^{Q/B_o} = x \hat{\mathbf{b}}_x + y \hat{\mathbf{b}}_y \quad \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$$



- Denoting $\vec{\mathbf{r}}$ as Q 's position from B_o , form $|^B d\vec{\mathbf{r}}|$ (the magnitude of the **vector differential** of $\vec{\mathbf{r}}$ in B). Next, write an integral (with appropriate limits) for the ellipse's circumference in terms of $d\theta$.² Then, express r and $\frac{dr}{d\theta}$ in terms of θ .

Result:

$$|^B d\vec{\mathbf{r}}| = \sqrt{(r d\theta)^2 + dr^2} \quad \text{Circumference} = \int_{\theta=0}^{2\pi} |^B d\vec{\mathbf{r}}| = \int_{\theta=0}^{2\pi} \sqrt{r^2 + \left(\frac{dr}{d\theta}\right)^2} d\theta$$

$$r = \frac{ab}{\sqrt{a^2 \sin^2(\theta) + b^2 \cos^2(\theta)}} \quad \frac{dr}{d\theta} = \frac{-ab(a^2 - b^2) \sin(\theta) \cos(\theta)}{\sqrt{a^2 \sin^2(\theta) + b^2 \cos^2(\theta)}^3}$$

- Find a formula for the circumference when $a = b$ is **constant** (the ellipse is a circle). Calculate the circumference of an ellipse (4⁺ significant digits) when $a = 2$ m and $b = 4$ m. Express the area $d\Delta$ of the triangle formed by points B_o , $Q(\theta)$, and $Q(\theta + d\theta)$: first as a function of $\vec{\mathbf{r}}$ and $d\vec{\mathbf{r}}$; then as a function of r and θ . Next, calculate the area of an ellipse (4⁺ significant digits) when $a = 2$ m and $b = 4$ m.³

Result:

$$\text{Circumference of circle} = 2\pi b \quad \text{Circumference of ellipse} = 19.377 \text{ m}$$

$$d\Delta = \frac{1}{2} |\vec{\mathbf{r}} \times d\vec{\mathbf{r}}| d\theta = \frac{1}{2} r^2 d\theta = \frac{1}{2} \frac{a^2 b^2}{a^2 \sin^2(\theta) + b^2 \cos^2(\theta)} d\theta \quad \text{Area of ellipse} = 25.133 \text{ m}^2$$

- There is an exact closed-form solution for the circumference of an ellipse. **True/False**.
There is an exact closed-form solution for the area of an ellipse. **True/False**.
- Draw** an outward normal vector $\vec{\mathbf{n}}$ and tangent vector $\vec{\mathbf{t}}$ at point Q . Express $\vec{\mathbf{n}}$ and $\vec{\mathbf{t}}$ in terms of $\hat{\mathbf{b}}_x, \hat{\mathbf{b}}_y, \hat{\mathbf{b}}_z$ when $x = \frac{a}{2}$.

Result:

Ellipse: $a = 2$ and $b = 4$	Circle: $a = 2$ and $b = 2$
$\vec{\mathbf{n}} = 1.0 \hat{\mathbf{b}}_x + 0.433 \hat{\mathbf{b}}_y$	$\vec{\mathbf{n}} = 0.5 \hat{\mathbf{b}}_x + 0.866 \hat{\mathbf{b}}_y$
$\vec{\mathbf{t}} = -0.433 \hat{\mathbf{b}}_x + 1.0 \hat{\mathbf{b}}_y$	$\vec{\mathbf{t}} = -0.866 \hat{\mathbf{b}}_x + 0.5 \hat{\mathbf{b}}_y$

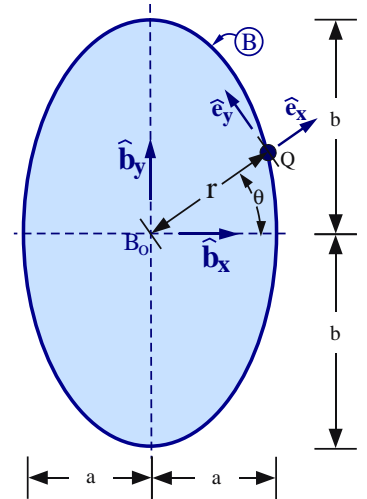
²The integral simplifies by noting that $d\theta$ is **positive** when the integral's upper-limit is larger than the integral's lower-limit.

³Verify your numerical integration result with the following formula for the area of ellipse: Area = $\pi a b$.

When $a = b$ (the ellipse is a circle), \vec{n} is always parallel to \vec{r}^{Q/B_o} **True/False**
 When $a \neq b$ (the ellipse is not a circle), \vec{n} is always parallel to \vec{r}^{Q/B_o} **True/False**

5. **Optional:** Show how the definition of an *ellipse* results in $F(x, y) = \frac{x^2}{a^2} + \frac{y^2}{b^2} - 1 = 0$.

```
% File: EllipseCircleNormalGradientCircumferenceArea.al
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
RigidBody B % Ellipse Bx> and By> are directed along ellipse axes
Point Q( B ) % Point on periphery of ellipse B
RigidFrame E % Ex> points from Bo to Q and Ez> = Bz>.
Variable r' % Distance between Bo and Q and dr/dtheta
Specified x, y % Locates Q's position from Bo
Constant a+, b+ % Dimensions of ellipse's minor/major axes
IndependentVariable theta
%-----
% Equation for an ellipse
EllipseCurve = (x/a)^2 + (y/b)^2 - 1
%-----
% The outward normal at Q is parallel to B's gradient at Q.
Gradient> = D( EllipseCurve, x )*Bx> + D( EllipseCurve, y )*By>
Tangent> = Cross( Bz>, Gradient> )
%-----
% Solve for positive value of y when x = a/2
CurveEvaluated = Evaluate( EllipseCurve, x = a/2 )
yAtX = SolveQuadraticPositiveRoot( CurveEvaluated = 0, y )
%-----
% Gradient and tangent when x = a/2, a = 2, and b = 4
EllipseGradient> = Evaluate( Gradient>, x = 4/2, y = yAtX, a = 2, b = 4 )
EllipseTangent> = Evaluate( Tangent>, x = 4/2, y = yAtX, a = 2, b = 4 )
%-----
% Circle gradient and tangent when x = a/2 and a = b = 2
CircleGradient> = Evaluate( Gradient>, x = 2/2, y = yAtX, a = 2, b = 2 )
CircleTangent> = Evaluate( Tangent>, x = 2/2, y = yAtX, a = 2, b = 2 )
%-----
% Rotation matrix relating Exyz> to Bxyz>
E.SetRotationMatrixZ( B, theta )
r> = r*Ex> % Q's position from point Bo
x = Dot( r>, Bx> )
y = Dot( r>, By> )
%-----
% Solve ellipse equation for positive r in terms of theta.
positiveRoot = GetQuadraticPositiveRoot( EllipseCurve = 0, r )
SetDt( r = a*b / sqrt(a^2*sin(theta)^2 + b^2*cos(theta)^2) )
%-----
% Calculate derivative of r with respect to theta in B - and its magnitude
DpDTheta> = Dt( r>, B )
magDpDTheta = GetMagnitude( DpDTheta> )
%-----
% Calculate circumference of circle and ellipse
magDpDThetaForCircle = Evaluate( magDpDTheta, a = b )
CircumferenceOfCircle = magDpDThetaForCircle * Integrate( 1, theta=0 : 2*pi )
magDpDThetaForEllipse = Evaluate( magDpDTheta, a=2, b=4 )
CircumferenceOfEllipse = Integrate( magDpDThetaForEllipse, theta = 0 : 2*pi )
%-----
% Differential area of ellipse (uses area of triangle)
dAreaDTheta> = 1/2 * Cross( r>, r> + DpDTheta> )
dAreaDTheta = Dot( dAreaDTheta>, Bz> )
%-----
% Calculate area of ellipse
dAreaDThetaForEllipse = Evaluate( dAreaDTheta, a=2, b=4 )
AreaOfEllipse = Integrate( dAreaDThetaForEllipse, theta = 0 : 2*pi )
ShouldApproximateZero = AreaOfEllipse - Evaluate( pi*a*b, a = 2, b = 4 )
%-----
Save EllipseCircleNormalGradientCircumferenceArea.all
If( abs(ShouldApproximateZero) < 1.0E-7 ) {Quit}
```



6.7 MotionGenesis translation commands (position, velocity, and acceleration)

Command	Description and associated formula
Q.Translate(P, posVector)	Sets Q's position from P. Sets Q's velocity and acceleration.
Q.Translate(P, posVector, B)	Sets Q's position from P. Sets Q's velocity and acceleration. (Both P and Q are fixed on reference frame B).
Q.Translate(P, posVector, B, BQ)	Sets Q's position from P. Sets Q's velocity and acceleration. (P is fixed on B. Q is moving on B and coincident with BQ).
Q.GetPosition(No)	Gets Q's position from No, i.e., $\vec{r}^{Q/No}$.
Q.SetPosition(P, posVector)	Sets Q's position from P, i.e., $\vec{r}^{Q/P} = \text{posVector}$
Q.GetDistance(P)	Gets Q's distance from P, i.e., $ \vec{r}^{Q/P} $.
Q.GetElongation(P)	Gets Q's elongation from P, i.e., the time-derivative of $ \vec{r}^{Q/P} $
Q.GetSpeed(N)	Gets Q's speed in N, i.e., $ \vec{v}^Q $
Q.GetVelocity(N)	Gets Q's velocity in N, i.e., \vec{v}^Q
Q.SetVelocity(N, velVector)	Sets Q's velocity in N, i.e., $\vec{v}^Q = \text{velVector}$
Q.SetVelocity(N, P)	Sets Q's velocity in N via $\vec{v}^Q = \vec{v}^P + \frac{d\vec{r}^{Q/P}}{dt}$
Q.SetVelocity(N, Bo, B)	Velocity of two points (Q and Bo) fixed on rigid object B. $\vec{v}^Q = \vec{v}^{Bo} + \vec{\omega}^B \times \vec{r}^{Q/Bo}$
Q.GetAcceleration(N)	Gets Q's acceleration in N, i.e., \vec{a}^Q
Q.SetAcceleration(N, accelVector)	Sets Q's acceleration in N, i.e., $\vec{a}^Q = \text{accelVector}$
Q.SetAcceleration(N, P)	Sets Q's acceleration in N via $\vec{a}^Q = \vec{a}^P + \frac{d^2\vec{r}^{Q/P}}{dt^2}$
Q.SetAcceleration(N, Bo, B)	Acceleration of two points (Bo and Q) fixed on rigid object B. $\vec{a}^Q = \vec{a}^{Bo} + \vec{\alpha}^B \times \vec{r}^{Q/Bo} + \vec{\omega}^B \times (\vec{\omega}^B \times \vec{r}^{Q/Bo})$
Q.SetVelocityAcceleration(N, velVector)	Sets $\vec{v}^Q = \text{velVector}$ and $\vec{a}^Q = \frac{d\text{velVector}}{dt}$
Q.SetVelocityAcceleration(N, P, ...)	Sets Q's velocity and acceleration (see previous related syntax).
Note: N and B are reference frames (or rigid bodies) whereas Q, No, Bo, and BQ are points (or particles).	

MotionGenesis translational kinematics (and dynamics) for an inverted pendulum on a cart

```

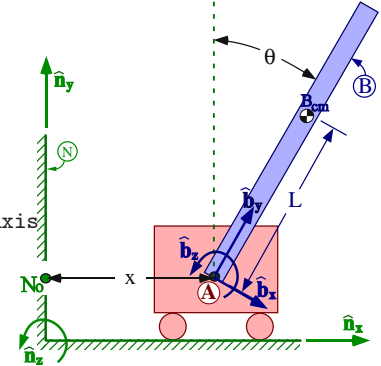
(1) % File: InvertedPendulumOnCartDynamics.txt
(2) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) NewtonianFrame N
(5) Particle A % Cart
(6) RigidBody B % Inverted pendulum
(7) %-----
(8) Variable x'' % Distance between No to A
(9) Variable theta'' % Angle from local vertical to B's long axis
(10) Specified Fc % Control force on cart
(11) Constant g+ = 9.81 m/s^2 % Gravitational constant
(12) Constant L+ = 0.5 m % Distance between A and Bcm
(13) A.SetMass( mA = 10 kg )
(14) B.SetMassInertia( mB = 1 kg, Izz = 1/12*mB*(2*L)^2, 0, Izz )
-> (15) Izz = 0.3333333*mB*L^2

(16) %-----
(17) % Rotational and translational kinematics
(18) B.RotateNegativeZ( N, theta )
-> (19) B_N = [cos(theta), -sin(theta), 0; sin(theta), cos(theta), 0; 0, 0, 1]
-> (20) w_B_N = -theta'*Bz>
-> (21) alf_B_N = -theta''*Bz>

(22) A.Translate( No, x*Nx> )
-> (23) p_No_A> = x*Nx>
-> (24) v_A_N> = x'*Nx>
-> (25) a_A_N> = x''*Nx>

(26) Bcm.Translate( A, L*By> )
-> (27) p_A_Bcm> = L*By>
-> (28) v_Bcm_N> = L*theta'*Bx> + x'*Nx>
-> (29) a_Bcm_N> = L*theta''*Bx> - L*theta'^2*By> + x''*Nx>

```



```

(30) %-----
(31) %           Relevant contact and distance forces
(32) System.AddForceGravity( -g*Ny> )
-> (33) Force_A> = -g*mA*Ny>
-> (34) Force_Bcm> = -g*mB*Ny>

(35) A.AddForce( Fc*Nx> )
-> (36) Force_A> = Fc*Nx> - g*mA*Ny>

(37) %-----
(38) %           Form and simplify equations of motion (via Newton/Euler)
(39) EquationsOfMotion[1] = Dot( Nx>, System(A,B).GetDynamics() )
-> (40) EquationsOfMotion[1] = mA*x'' + mB*x'' + L*mB*cos(theta)*theta'' - Fc
      - L*mB*sin(theta)*theta'^2

(41) EquationsOfMotion[2] = Dot( -Bz>, B.GetDynamics(A) )
-> (42) EquationsOfMotion[2] = Izz*theta'' + mB*L^2*theta'' - L*mB*(g*sin(theta)
      -cos(theta)*x'')

(43) FactorLinear( EquationsOfMotion, theta'', x'', Fc, g )
-> (44) EquationsOfMotion[1] = (mA+mB)*x'' + L*mB*cos(theta)*theta'' - Fc - L*
      mB*sin(theta)*theta'^2
-> (45) EquationsOfMotion[2] = L*mB*cos(theta)*x'' + (mB*L^2+Izz)*theta'' - g*L
      *mB*sin(theta)

(46) %-----

```

MotionGenesis input and output responses for particle moving on rotating Earth

```

(1) % File: ParticleInRiverOnFlatEarth.al
(2) %-----
(3) NewtonianFrame N           % Newtonian reference frame
(4) RigidBody      E           % Earth
(5) Particle      Q           % Particle in the river
(6) Point         EQ(E)       % Point of E that is coincident with Q
(7) %-----
(8) %           Angular velocity and angular acceleration
(9) EarthSpinInRadPerSec = ConvertUnits( 1 revolution/day, rad/sec )
-> (10) EarthSpinInRadPerSec = 7.272205E-05

(11) E.SetAngularVelocityAcceleration( N, EarthSpinInRadPerSec*Ez> )
-> (12) w_E_N> = 7.272205E-05*Ez>
-> (13) alf_E_N> = 0>

(14) %-----
(15) %           Positions
(16) Q.SetPosition( Eo, 3.0E+6*Ex> )
-> (17) p_Eo_Q> = 3000000*Ex>

(18) EQ.SetPosition( Eo, Q.GetPosition(Eo) )
-> (19) p_Eo_EQ> = 3000000*Ex>

(20) %-----
(21) %           Velocities
(22) Eo.SetVelocity( N, 0> )           % Eo is fixed in N
-> (23) v_Eo_N> = 0>

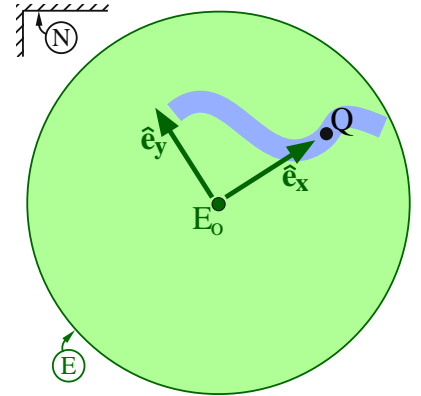
(24) Q.SetVelocity( E, 10*Ex> )         % Given: Q is moving downstream at 10 m/sec
-> (25) v_Q_E> = 10*Ex>

(26) EQ.SetVelocity( N, Eo, E )         % Two points (EQ and Eo) fixed on Earth (E)
-> (27) v_EQ_N> = 218.1662*Ey>

(28) Q.SetVelocity( N, Eo, E, EQ )     % Point Q is moving on E and is coincident with EQ
-> (29) v_Q_N> = 10*Ex> + 218.1662*Ey>

(30) %-----
(31) %           Acceleration

```



```

(32) Eo.SetAcceleration( N, 0> )      % Eo is fixed in N
-> (33) a_Eo_N> = 0>

(34) Q.SetAcceleration( E, 2*Ex> )    % Given: acceleration due to a steep grade
-> (35) a_Q_E> = 2*Ex>

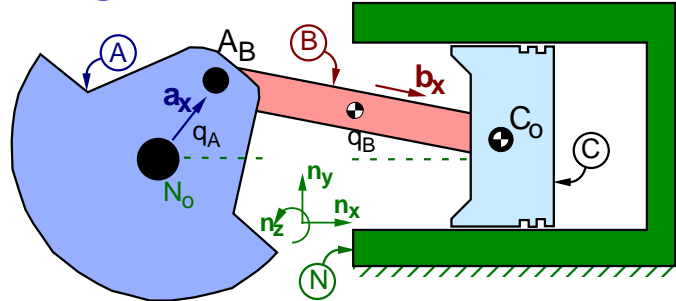
(36) EQ.SetAcceleration( N, Eo, E )   % Two points (EQ and Eo) fixed on Earth (E)
-> (37) a_EQ_N> = -0.01586549*Ex>

(38) Q.SetAcceleration( N, Eo, E, EQ ) % Point Q is moving on E and is coincident with EQ
-> (39) a_Q_N> = 1.984135*Ex> + 0.001454441*Ey>

```

6.8 Configuration constraints and straight slots with MotionGenesis

The figure to the right shows a piston C sliding in a cylinder that is fixed in a reference frame N . The piston is connected to a connecting rod B by a revolute joint at C_o . The connecting rod is connected to a crankshaft A by a second revolute joint at A_B . The center of the crankshaft is connected to N by a third revolute joint at N_o .



The point of this problem is to form a configuration constraint of the form $f(q_A, q_B) = 0$. After noticing that the configuration constraint is **nonlinear** in q_A and q_B , it is helpful to differentiate the configuration constraint to form a motion constraint $\dot{f}(\dot{q}_A, \dot{q}_B, q_A, q_B) = 0$ which is **linear** in \dot{q}_A and \dot{q}_B .

Solving nonlinear configuration constraints with MotionGenesis

```

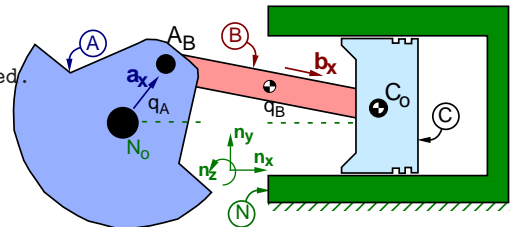
(1) % File: PistonEngineConstraintKinematics.txt
(2) % Problem: Kinematic analysis of a piston engine
(3) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(4) %-----
(5) NewtonianFrame N
(6) RigidBody A      % Crankshaft
(7) RigidBody B      % Connecting rod
(8) RigidBody C      % Piston
(9) Point AB        % Point connecting A and B
(10) %-----
(11) Variable qA', qB' % qA and qB are angles
(12) Constant LA = 10 cm % Length between No and AB
(13) Constant LB = 20 cm % Length between AB and Ccm
(14) Constant kp = 0 noUnits % For constraint stabilization
(15) %-----
(16) % Rotational kinematics
(17) A.RotatePositiveZ( N, qA )
-> (18) A_N = [cos(qA), sin(qA), 0; -sin(qA), cos(qA), 0; 0, 0, 1]
-> (19) w_A_N> = qA'*Az>

(20) B.RotateNegativeZ( N, qB )
-> (21) B_N = [cos(qB), -sin(qB), 0; sin(qB), cos(qB), 0; 0, 0, 1]
-> (22) w_B_N> = -qB'*Bz>

(23) %-----
(24) % Translational kinematics
(25) AB.SetPositionVelocity( No, LA*Ax> )
-> (26) p_No_AB> = LA*Ax>
-> (27) v_AB_N> = LA*qA'*Ay>

(28) Co.SetPositionVelocity( AB, LB*Bx> )
-> (29) p_AB_Co> = LB*Bx>
-> (30) v_Co_N> = LA*qA'*Ay> - LB*qB'*By>

```



```

(31) %-----
(32) %           Position constraint  $f(q_A, q_B) = 0$ 
(33)  $f = \text{Dot}(\text{Co.GetPosition}(\text{No}), \text{Ny} > )$ 
-> (34)  $f = L_A \sin(q_A) - L_B \sin(q_B)$ 

(35) %-----
(36) %           Input constants, variables, etc.
(37) Input     $q_A = 45 \text{ deg}$ 
(38) %-----
(39) %           Find initial value of  $q_B$  for given input values (guess  $q_B=0$ )
(40) SolveSetInput(  $f$ ,  $q_B = 0 \text{ deg}$  )

-> % INPUT has been assigned as follows:
-> %     $q_B$                                 20.70481105463543          deg

(41) %-----
(42) %           Velocity constraint with constraint stabilization:  $Dt(f) + k_p f$ 
(43)  $\text{velocityConstraint} = \text{Dot}(\text{Co.GetVelocity}(\text{N}), \text{Ny} > ) + k_p f$ 
-> (44)  $\text{velocityConstraint} = k_p f + L_A \cos(q_A) q_A' - L_B \cos(q_B) q_B'$ 

(45) Solve(  $\text{velocityConstraint}$ ,  $q_B'$  )
-> (46)  $q_B' = (k_p f + L_A \cos(q_A) q_A') / (L_B \cos(q_B))$ 

(47) %-----
(48) %           Simulate compressor with constant speed motor of 10 rad/sec
(49)  $q_A' = 10$ 
-> (50)  $q_A' = 10$ 

(51) %-----
(52) %           Input constants, variables, etc. for ODE command (for motion)
(53) Input     $t_{\text{Final}} = 6 \text{ sec}$ ,  $t_{\text{Step}} = 0.02 \text{ sec}$ ,  $\text{absError} = 1.0\text{E-}7$ 
(54) %-----
(55) %           List output quantities and solve ODEs (or write MATLAB, C, ... code).
(56) Output     $t \text{ sec}$ ,  $q_A \text{ deg}$ ,  $q_B \text{ deg}$ ,  $f \text{ cm}$ 
(57) ODE() PistonEngineConstraintKinematics

```

6.9 MotionGenesis commands for a particle



Particle Q	Declares Q as a particle
Q.SetMass(mQ)	Declares mQ as a non-negative constant (if mQ is not already defined) Assigns mQ to the mass of particle Q
Q.GetMass()	Returns Q 's mass
Q.GetLinearMomentum()	Returns Q 's linear momentum in the Newtonian reference frame
Q.GetAngularMomentum(P)	Returns Q 's angular momentum about point P in the Newtonian reference frame
Q.GetGeneralizedMomentum()	Returns Q 's generalized momentum in the Newtonian reference frame
Q.GetEffectiveForce()	Returns Q 's effective force in the Newtonian reference frame
Q.GetMomentOfEffectiveForce(P)	Returns moment of Q 's effective force about point P in the Newtonian frame
Q.GetGeneralizedEffectiveForce()	Returns Q 's generalized effective force in the Newtonian reference frame
Q.GetKineticEnergy()	Returns Q 's kinetic energy in the Newtonian reference frame
Q.GetInertiaDyadic(P)	Returns Q 's inertia dyadic about point P

Calculating a particle's momentum, kinetic energy, etc., with MotionGenesis

```

(1) % File: InfantOnSwing.txt
(2) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) NewtonianFrame N      % Newtonian reference frame
(5) RigidFrame B          % Rod connecting No to Q
(6) Particle Q            % Infant on swing
(7) %-----
(8) Variable theta''
(9) Constant L
(10) Q.SetMass( m )
(11) SetGeneralizedSpeed( theta' )
(12) %-----
(13) % Rotational and translational kinematics
(14) B.RotateZ( N, theta )
-> (15) B_N = [cos(theta), sin(theta), 0; -sin(theta), cos(theta), 0; 0, 0, 1]
-> (16) w_B_N> = theta'*Bz>
-> (17) alf_B_N> = theta''*Bz>

(18) Q.Translate( No, -L*By> )
-> (19) p_No_Q> = -L*By>
-> (20) v_Q_N> = L*theta'*Bx>
-> (21) a_Q_N> = L*theta''*Bx> + L*theta'^2*By>

(22) %-----
(23) % Q's linear/angular/generalized momentum in N
(24) LinearMomentum> = Q.GetLinearMomentum()
-> (25) LinearMomentum> = m*L*theta'*Bx>

(26) AngularMomentum> = Q.GetAngularMomentum( No )
-> (27) AngularMomentum> = m*L^2*theta'*Bz>

(28) GeneralizedMomentum = Q.GetGeneralizedMomentum()
-> (29) GeneralizedMomentum = [m*L^2*theta']

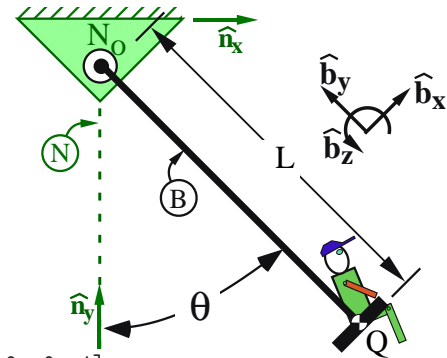
(30) %-----
(31) % Q's kinetic energy in N
(32) KineticEnergy = Q.GetKineticEnergy()
-> (33) KineticEnergy = 0.5*m*L^2*theta'^2

(34) %-----
(35) % Q's effective forces, etc., in N
(36) EffectiveForce> = Q.GetEffectiveForce()
-> (37) EffectiveForce> = m*L*theta''*Bx> + m*L*theta'^2*By>

(38) MomentOfEffectiveForce> = Q.GetMomentOfEffectiveForce( No )
-> (39) MomentOfEffectiveForce> = m*L^2*theta''*Bz>

(40) GeneralizedEffectiveForce = Q.GetGeneralizedEffectiveForce()
-> (41) GeneralizedEffectiveForce = [m*L^2*theta'']

```



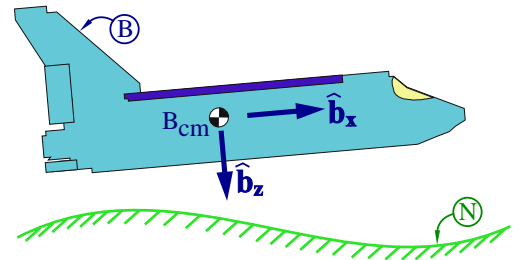
6.10 MotionGenesis commands for a rigid body



RigidBody B	Declares B as a rigid body
B.SetMass(mB)	Declares mB as a non-negative constant (if mB is not already defined) Assigns mB to the mass of B
B.SetInertia(Bcm, Ix, Iy, Iz, Ixy, Iyz, Izx)	Declares rigid body B 's moments/products of inertia about B_{cm}
B.GetMass()	Returns B's mass
B.GetInertiaDyadic(P)	Returns B's inertia dyadic about point P
B.GetLinearMomentum()	Returns B's linear momentum in the Newtonian reference frame
B.GetAngularMomentum(P)	Returns B's angular momentum about point P in the Newtonian reference frame
B.GetGeneralizedMomentum()	Returns B's generalized momentum in the Newtonian reference frame
B.GetEffectiveForce()	Returns B's effective force in the Newtonian reference frame
B.GetMomentOfEffectiveForce(P)	Returns moment of B's effective force about point P in the Newtonian frame
B.GetGeneralizedEffectiveForce()	Returns B's generalized effective force in the Newtonian reference frame
B.GetKineticEnergy()	Returns B's kinetic energy in the Newtonian reference frame

Calculating rigid-body's momentum, kinetic energy, etc., with MotionGenesis

```
% File: AirplaneExample.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
RigidBody B
B.SetMassInertia( m, Ixx, Iyy, Izz )
%-----
Variable vx', vy', vz', wx', wy', wz'
SetGeneralizedSpeed( vx, vy, vz, wx, wy, wz )
%-----
% B's angular velocity and angular acceleration in N
B.SetAngularVelocityAcceleration( N, wx*Bx> + wy*By> + wz*Bz> )
%-----
% Bcm's velocity and acceleration in N
Bcm.SetVelocityAcceleration( N, vx*Bx> + vy*By> + vz*Bz> )
%-----
% B's momentum calculations in N
LinearMomentum> = B.GetLinearMomentum()
AngularMomentum> = B.GetAngularMomentum( Bcm )
GeneralizedMomentum = B.GetGeneralizedMomentum( )
%-----
% B's effective force calculations in N
EffectiveForce> = B.GetEffectiveForce()
MomentEffectiveForce> = B.GetMomentOfEffectiveForce( Bcm )
GeneralizedEffectiveForce = B.GetGeneralizedEffectiveForce()
%-----
% B's kinetic energy in N
KineticEnergy = B.GetKineticEnergy()
Save AirplaneExampleLong.all
Quit
```





Chapter 7

Computing mass and inertia properties

7.1 MotionGenesis mass and center of mass commands



<code>Q.SetMass(mQ)</code>	Declares Q 's mass as the non-negative constant mQ (if mQ is not already defined)
<code>B.SetMass(mB = 3)</code>	Declares B 's mass as mB which is assigned to 3
<code>Q.GetMass() + B.GetMass()</code>	Returns the sum of Q 's mass and B 's mass
<code>System.GetMass()</code>	Returns the system's mass
<code>System.GetCMPosition(P)</code>	Returns the position vector of the system mass center from point P
<code>System.GetCMVelocity(N)</code>	Returns the velocity of the system's mass center in reference frame N
<code>System.GetCMAcceleration(N)</code>	Returns the acceleration of the system's mass center in reference frame N

Example: Calculating mass and center of mass of four particles with MotionGenesis

```

(1) % File: CenterOfMassOfFourParticles.txt
(2) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) RigidBody N
(5) Particle Q1, Q2, Q3, Q4
(6) %-----
(7) Q1.SetMass( 1 )
(8) Q2.SetMass( 2 )
(9) Q3.SetMass( 2 )
(10) Q4.SetMass( 2 )
(11) %-----
(12) Q1.SetPosition( No, -Nx> )
-> (13) p_No_Q1> = -Nx>

(14) Q2.SetPosition( No, Ny> )
-> (15) p_No_Q2> = Ny>

(16) Q3.SetPosition( No, Nx> )
-> (17) p_No_Q3> = Nx>

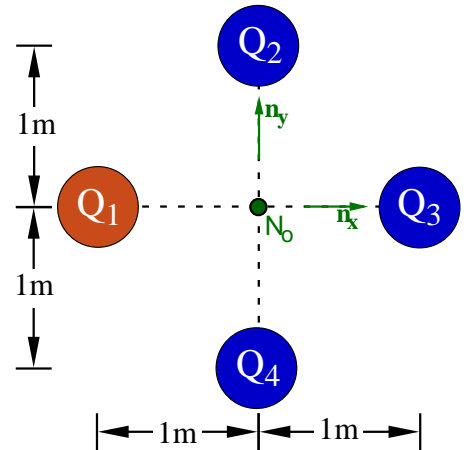
(18) Q4.SetPosition( No, -Ny> )
-> (19) p_No_Q4> = -Ny>

(20) %-----
(21) % Mass, center of mass, and centroid
(22) MassOfSystem = System.GetMass()
-> (23) MassOfSystem = 7

(24) CMPositionFromNo> = System.GetCMPosition( No )
-> (25) CMPositionFromNo> = 0.1428571*Nx>

(26) CentroidPositionFromNo> = 1/4*( Q1.GetPosition(No) + Q2.GetPosition(No) &
    + Q3.GetPosition(No) + Q4.GetPosition(No) )
-> (27) CentroidPositionFromNo> = 0>

```



7.2 MotionGenesis inertia commands



<code>B.SetInertia(Bp, Ix,Iy,Iz, Ixy,Iyz,Izx)</code>	Declares rigid body B 's moments/products of inertia about Bp
<code>B.SetMassInertia(m, Ix,Iy,Iz, Ixy,Iyz,Izx)</code>	Same as: <code>B.SetMass(m); B.SetInertia(Bcm, Ix,Iy,Iz, Ixy,Iyz,Izx)</code>
<code>B.GetInertiaDyadic(Bcm)</code>	Returns B 's inertia dyadics about point B_{cm}
<code>System.GetInertiaDyadic(P)</code>	Returns the system's inertia dyadic about point P
<code>System.GetInertiaDyadic(P, N)</code>	Returns the system's inertia dyadic about point P expressed in $\hat{n}_x, \hat{n}_y, \hat{n}_z$
<code>System.GetInertiaMatrix(P, N)</code>	Returns the system's inertia matrix about point P for $\hat{n}_x, \hat{n}_y, \hat{n}_z$
<code>System.GetMomentOfInertia(P, Nx>)</code>	Returns the system's moment of inertia about point P for \hat{n}_x
<code>System.GetRadiusOfGyration(P, Nx>)</code>	Returns the system's radius of gyration about point P for \hat{n}_x
<code>System.GetProductOfInertia(P, Nx>, Ny>)</code>	Returns the system's product of inertia about point P for \hat{n}_x and \hat{n}_y

7.2.1 Example: Inertia properties for an infant on a swing with MotionGenesis

```

(1) % File: InertiaPropertiesOfInfantOnSwing.txt
(2) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) RigidBody N
(5) RigidBody B
(6) Particle Q
(7) Constant L
(8) Variable theta
(9) Q.SetMass( m )
(10) %-----
(11) % Q's position vector from No
(12) Q.SetPosition( No, -L*By> )
-> (13) p_No_Q> = -L*By>

(14) %-----
(15) % Q's inertia dyadic about No expressed in B
(16) QInertiaDyadicAboutNo>> = Q.GetInertiaDyadic( No, B )
-> (17) QInertiaDyadicAboutNo>> = m*L^2*Bx>*Bx> + m*L^2*Bz>*Bz>

(18) %-----
(19) % B's rotation matrix in N
(20) B.RotateZ( N, theta )
-> (21) B_N = [cos(theta), sin(theta), 0; -sin(theta), cos(theta), 0; 0, 0, 1]

(22) %-----
(23) % Q's moments of inertia about No for various directions
(24) IBxBx = Q.GetMomentOfInertia( No, Bx> )
-> (25) IBxBx = m*L^2

(26) IByBy = Q.GetMomentOfInertia( No, By> )
-> (27) IByBy = 0

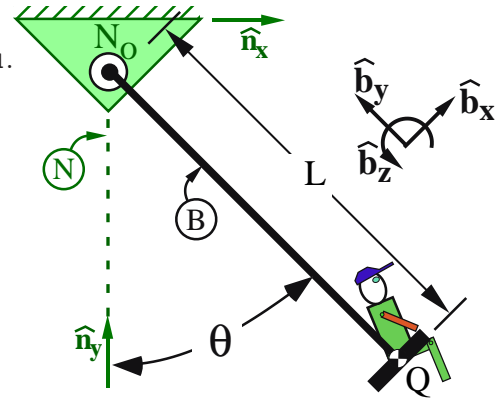
(28) IBzBz = Q.GetMomentOfInertia( No, Bz> )
-> (29) IBzBz = m*L^2

(30) INxNx = Q.GetMomentOfInertia( No, Nx> )
-> (31) INxNx = m*L^2*cos(theta)^2

(32) %-----
(33) % Q's products of inertia about No for various directions
(34) IBxBy = Q.GetProductOfInertia( No, Bx>, By> )
-> (35) IBxBy = 0

(36) INxNy = Q.GetProductOfInertia( No, Nx>, Ny> )
-> (37) INxNy = m*L^2*sin(theta)*cos(theta)

```



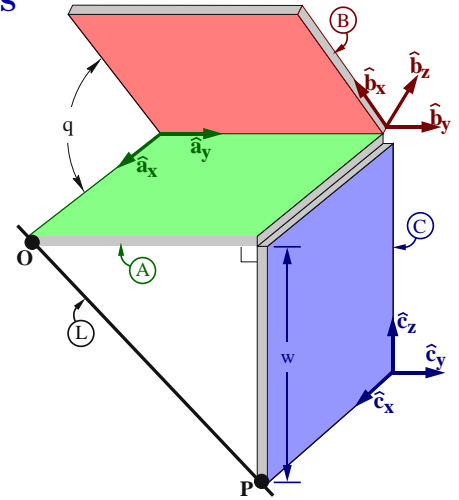
7.3 Mass, mass center, and inertia calculations

The figure to the right shows three identical uniform hinge-connected plates A , B , and C .

Right-handed sets of mutually perpendicular unit vectors $\hat{\mathbf{a}}_i$, $\hat{\mathbf{b}}_i$, and $\hat{\mathbf{c}}_i$ ($i = x, y, z$) are fixed in A , B , and C , respectively, with $\hat{\mathbf{a}}_y = \hat{\mathbf{b}}_y = \hat{\mathbf{c}}_y$ parallel to the hinge connecting A and B .

The plates are thin and square and have dimension $w = 1$ meter and mass $m = 12$ kg.

Points O and P mark corners of A and C and the angle q characterizes B 's orientation in A .



Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow [Plate mass/inertia.](#)

- Find the distance between line \overline{OP} and the center of mass S_{cm} of the system formed by A , B , C .

Result:

$$\text{Distance} = 0.1178511 \sqrt{42 + \cos^2(q) + 6 \sin(q) - 16 \cos(q)}$$

- For $q = 90^\circ$, find λ_i ($i = 1, 2, 3$), the system's principal moments of inertia about S_{cm} . Next, find the angle between $\hat{\mathbf{a}}_y$ and the principal axis associated with this system's **minimum** moment of inertia.

Result:

$$\lambda_1 = 5 \text{ kg m}^2 \quad \lambda_2 = 13 \text{ kg m}^2 \quad \lambda_3 = 14 \text{ kg m}^2 \quad \text{Angle} = 65.90516^\circ$$

- The system's radius of gyration about line \overline{OP} is a measure of how far the mass distribution is from line \overline{OP} and is defined as $\sqrt{I/m}$ where I is the system's moment of inertia about \overline{OP} and m is the system's mass. Using **physical intuition**, estimate the values of q that produce the smallest and largest radii of gyration about line \overline{OP} and provide a reason for choosing these values.

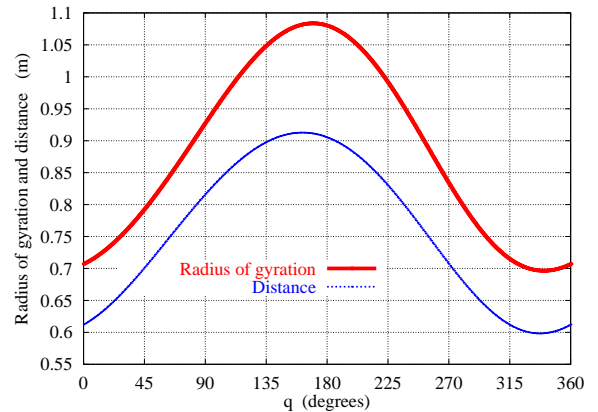
Result: $q_{\text{small}} \approx 340^\circ$ $q_{\text{large}} \approx 160^\circ$ ($0 \leq q \leq 360^\circ$)

Reason: These values minimize or maximize the distance from line OP to B 's mass center.

Plot the system's mass center distance from line \overline{OP} and the system's radius of gyration about line \overline{OP} for $0 \leq q \leq 360^\circ$. Determine the minimum/maximum distance and radius of gyration

- and associated values of q .

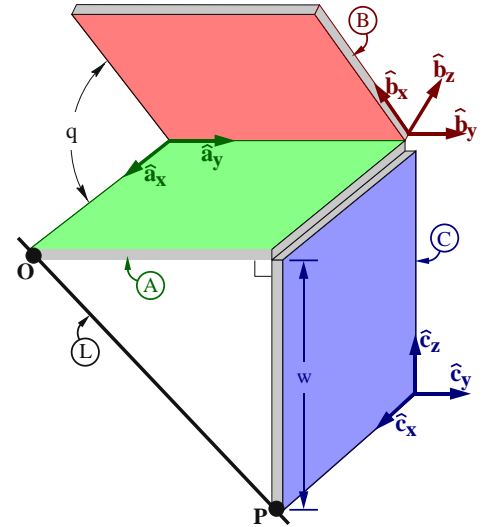
	Minimum Value q		Maximum Value q	
Distance	0.598 m	337°	0.913 m	161°
Gyration	0.696 m	340°	1.084 m	169°



```

% MotionGenesis file:  ThreePlatesMassInertia.txt
% Copyright (c) 2009 Motion Genesis LLC.  All rights reserved.
%-----
%      Physical declarations
RigidBody  A, B, C          % Right, middle, top plate
Point      O, P            % Points on L
Point      SystemCM        % System's center of mass
%-----
%      Mathematical declarations
Variable    q               % Angle between plates A and B
Constant    w = 1 m        % Width of plate
A.SetMassInertia( m = 12 kg, I = m*w^2/12, I, 2*I )
B.SetMassInertia( m,        I,          I, 2*I )
C.SetMassInertia( m,        I,          2*I, I )
%-----
%      Geometry relating unit vectors
B.RotateNegativeY( A, q )
C.SetRotationMatrix( A, IdentityMatrix(3) )
%-----
%      Position vectors
Acm.SetPosition( O, -0.5*w*Ax> + 0.5*w*Ay> )
Bcm.SetPosition( O, -w*Ax> + 0.5*w*Ay> + 0.5*w*Bx> )
Ccm.SetPosition( O, w*Ay> - 0.5*w*Cx> - 0.5*w*Cz> )
P.SetPosition( O, w*Ay> - w*Cz> )
%-----
%      System's center of mass position from point O.
P_O_SystemCM> = System.GetCMPosition( O )
uL> = P.GetUnitVector( O ) % Unit vector parallel to line L
DistanceFromLineLToSystemCM = GetMagnitude( Cross(uL>, SystemCM.GetPosition(O) ) )
%-----
%      System's inertia matrix about its center of mass.
SetAutoEpsilon( 1.0E-14 ) % Round to integers
SystemInertiaMatrix = EvaluateToNumber( System.GetInertiaMatrix( SystemCM, A ) )
%-----
%      Extract 1st eigenvector and calculate angle between Ay>
Lamba = GetEigen( Evaluate( SystemInertiaMatrix, q = 90 deg), EigenVecs )
EigenColumnMatrix1 = GetColumn( EigenVecs, 1 )
EigenVec1> = Vector( A, EigenColumnMatrix1 )
AngleBetweenEigenVec1AndAy = GetAngleBetweenUnitVectorsDegrees( EigenVec1>, Ay> )
%-----
%      System's moment of inertia and radius of gyration of system about line OP.
MomentOfInertiaAboutLineL = System.GetMomentOfInertia( O, uL> )
RadiusOfGyrationAboutL = System.GetRadiusOfGyration( O, uL> )
%-----
%      Create Output and write MATLAB (or C or Fortran) program.
Output q degs, DistanceFromLineLToSystemCM m, RadiusOfGyrationAboutL m
CODE Algebraic() [ q deg = 0, 360, 1 ] ThreePlatesMassInertia.m
%-----
Save ThreePlatesMassInertia.html
Quit

```



7.3.1 Example: Mass properties of three particles with MotionGenesis

```

(1) % File: ThreeParticlesOnParellelpipedMassPropertiesEigen.al
(2) %-----
(3) RigidBodyFrame N
(4) Particle A, B, C
(5) Point CM
(6) A.SetMass( 1 )
(7) B.SetMass( 1 )
(8) C.SetMass( 1 )
(9) %-----
(10) % Position vectors
(11) A.SetPosition( No, 2*Nx> )
-> (12) p_No_A> = 2*Nx>

(13) B.SetPosition( No, 2*Ny> )
-> (14) p_No_B> = 2*Ny>

(15) C.SetPosition( No, Nz> )
-> (16) p_No_C> = Nz>

(17) %-----
(18) % Calculate and set CM's position from point No
(19) CM.SetPosition( No, System.GetCMPosition(No) )
-> (20) p_No_CM> = 0.6666667*Nx> + 0.6666667*Ny> + 0.3333333*Nz>

(21) %-----
(22) % System's inertia dyadic about No (expressed in N basis)
(23) InertiaDyadicAboutNo>> = System.GetInertiaDyadic( No, N )
-> (24) InertiaDyadicAboutNo>> = 5*Nx>*Nx> + 5*Ny>*Ny> + 8*Nz>*Nz>

(25) %-----
(26) % System moment of inertia and radius of gyration about diagonal of parallelepiped
(27) Diagonal> = GetUnitVector( 2*Nx> + 2*Ny> + Nz> )
-> (28) Diagonal> = 0.6666667*Nx> + 0.6666667*Ny> + 0.3333333*Nz>

(29) MomentOfInertiaAboutDiagonal = System.GetMomentOfInertia( No, Diagonal> )
-> (30) MomentOfInertiaAboutDiagonal = 5.333333

(31) RadiusOfGyration = Sqrt( MomentOfInertiaAboutDiagonal / System.GetMass() )
-> (32) RadiusOfGyration = 1.333333

(33) %-----
(34) % System's inertia dyadic about CM expressed in N basis
(35) InertiaDyadicAboutCM>> = System.GetInertiaDyadic( CM, N )
-> (36) InertiaDyadicAboutCM>> = 3.333333*Nx>*Nx> + 1.333333*Nx>*Ny> + 0.6666667
    *Nx>*Nz> + 1.333333*Ny>*Nx> + 3.333333*Ny>*Ny> + 0.6666667*Ny>*Nz>
    + 0.6666667*Nz>*Nx> + 0.6666667*Nz>*Ny> + 5.333333*Nz>*Nz>

(37) %-----
(38) % System's principal moments of inertia about CM and corresponding directions
(39) InertiaMatrixAboutCM = System.GetInertiaMatrix( CM, N )
-> (40) InertiaMatrixAboutCM = [3.333333, 1.333333, 0.6666667; 1.333333, 3.333333,
    0.6666667; 0.6666667, 0.6666667, 5.333333]

(41) eigenValues = GetEigen( InertiaMatrixAboutCM, eigenVectorMatrix )
-> (42) eigenVectorMatrix = [0.7071068, -0.5773503, 0.4082483; -0.7071068, -0.5773503,
    0.4082483; 6.048673E-17, 0.5773503, 0.8164966]

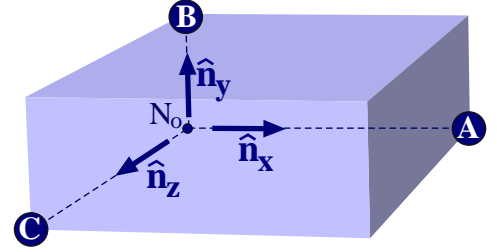
-> (43) eigenValues = [2; 4; 6]

(44) direction1> = Vector( N, GetColumn(eigenVectorMatrix,1) )
-> (45) direction1> = 0.7071068*Nx> - 0.7071068*Ny> + 6.048673E-17*Nz>

(46) direction2> = Vector( N, GetColumn(eigenVectorMatrix,2) )
-> (47) direction2> = -0.5773503*Nx> - 0.5773503*Ny> + 0.5773503*Nz>

(48) direction3> = Vector( N, GetColumn(eigenVectorMatrix,3) )
-> (49) direction3> = 0.4082483*Nx> + 0.4082483*Ny> + 0.8164966*Nz>

```





Chapter 8

Forces, torques, moments, and statics

8.1 MotionGenesis commands and syntax for force, torque, and moments

MotionGenesis command	Description and associated formula
Q.AddForce(forceVector)	Adds forceVector to the force on point Q.
Q.AddForce(P, forceVector)	Adds forceVector to the force on point Q from point P.
S.AddForceGravity(gravityVector)	Adds a uniform gravitational force to the particle, body, or system S.
Q.AddForceGravity(P, G)	Adds an inverse-square gravity force on particle Q from particle P.
Q.AddForceSpring(P, k, Ln, ...)	Adds a spring force to point Q from point P.
Q.AddForceDamper(P, b, ...)	Adds a damper force to point Q from point P.
S.GetResultantForce()	Gets the resultant of all forces on S.
Q.GetResultantForce(P)	Gets the resultant of all forces on point Q from point P.
B.AddTorque(torqueVector)	Adds torqueVector to the torque on RigidFrame B.
B.AddTorque(A, torqueVector)	Adds torqueVector to the torque on RigidFrame B from RigidFrame A.
B.AddTorqueDamper(A, b, ...)	Adds a damper torque to RigidFrame B from RigidFrame A.
B.GetResultantTorque()	Gets the resultant of all torques on RigidFrame B.
B.GetResultantForce(A)	Gets the resultant of all torques on RigidFrame B from RigidFrame A.
S.GetMomentOfForces(P)	Gets the moment of all forces on S about point P.

The MotionGenesis syntax **Force_Q>** denotes a force on point Q.
The MotionGenesis syntax **Force_Q_P>** denotes the force on point Q from point P.
The MotionGenesis syntax **Torque_B>** denotes a torque on rigid frame (or rigid body) B.
The MotionGenesis syntax **Torque_B_A>** denotes a torque on B from a rigid frame (or rigid body) A.

Definition and description	Picture	How used & MotionGenesis commands
$\vec{F}^{Q/P}$ is the force on Q from P. Law of action/reaction for forces: $\vec{F}^{P/Q} = -\vec{F}^{Q/P}$ (P and Q are points)		Gravity, spring, damper, and other forces have special MotionGenesis commands. Q.AddForce(P, someVector)
<u>Resultant</u> force on point Q: $\vec{F}^Q \triangleq \sum_{i=1}^n \vec{F}^{Q/P_i}$ <u>Resultant</u> force on body B: $\vec{F}^B \triangleq \sum_{j=1}^n \vec{F}^{B_j}$		$\vec{F}^Q = \vec{0}$ (Statics) $\vec{F}^Q = m^Q \vec{a}^Q$ (Dynamics) Q.AddForce(someVector) Q.GetResultantForce() $\vec{F}^B = \vec{0}$ (Statics) $\vec{F}^B = m^B \vec{a}^{B_{cm}}$ (Dynamics)
<u>Resultant</u> force on system S: $\vec{F}^S \triangleq \sum_{k=1}^n \vec{F}^{Q_k}$		$\vec{F}^S = \vec{0}$ (Statics) $\vec{F}^S = m^S \vec{a}^{S_{cm}}$ (Dynamics) System.GetResultantForce()

8.2 Static truss analysis with MotionGenesis

```

(1) % File: TrussABCTopLoad.al (Truss analysis)
(2) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(3) %-----
(4) NewtonianFrame N % Ground
(5) RigidFrame S % Entire truss
(6) Point A(S), B(S), C(S) % Nodes on truss
(7) %-----
(8) Constant L % Twice the distance between A and C
(9) Constant theta % Angle between AC and AB
(10) Constant W % Weight applied to node B
(11) Variable FAx, FAy % Nx>, Ny> measures of external force on A
(12) Variable FCy % Ny> measure of external force on C
(13) Variable FAB % Force on A from AB member directed from A to B
(14) Variable FAC % Force on A from AC member directed from A to C
(15) %-----
(16) % Relevant external contact and distance forces on S
(17) A.AddForce( FAx*Nx> + FAy*Ny> )
-> (18) Force_A> = FAx*Nx> + FAy*Ny>

(19) B.AddForce( -W*Ny> )
-> (20) Force_B> = -W*Ny>

(21) C.AddForce( FCy*Ny> )
-> (22) Force_C> = FCy*Ny>

(23) %-----
(24) % Static analysis of entire system
(25) ResultantForceOnS> = S.GetResultantForce()
-> (26) ResultantForceOnS> = FAx*Nx> + (FAy+FCy-W)*Ny>

(27) MomentOfSAboutA> = Cross( L*Nx>, -W*Ny> ) + Cross( 2*L*Nx>, FCy*Ny> )
-> (28) MomentOfSAboutA> = -L*(W-2*FCy)*Nz>

(29) ZeroSystem[1] = Dot( ResultantForceOnS>, Nx> )
-> (30) ZeroSystem[1] = FAx

(31) ZeroSystem[2] = Dot( ResultantForceOnS>, Ny> )
-> (32) ZeroSystem[2] = FAy + FCy - W

(33) ZeroSystem[3] = Dot( MomentOfSAboutA>, Nz> )
-> (34) ZeroSystem[3] = -L*(W-2*FCy)

(35) Solve( ZeroSystem, FAx, FAy, FCy )
-> (36) FAx = 0
-> (37) FAy = 0.5*W
-> (38) FCy = 0.5*W

(39) %-----
(40) % Relevant external contact and distance forces on pin A
(41) UnitVectorFromAToB> = cos(theta)*Nx> + sin(theta)*Ny>
-> (42) UnitVectorFromAToB> = cos(theta)*Nx> + sin(theta)*Ny>

(43) A.AddForce( B, FAB * UnitVectorFromAToB> )
-> (44) Force_A_B> = cos(theta)*FAB*Nx> + sin(theta)*FAB*Ny>

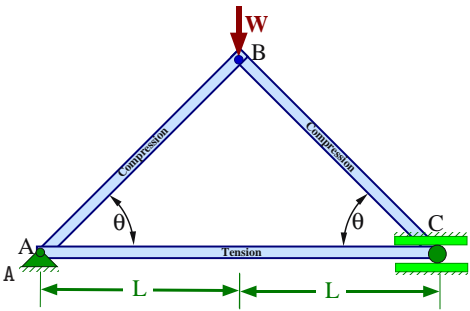
(45) A.AddForce( C, FAC * Nx> )
-> (46) Force_A_C> = FAC*Nx>

(47) %-----
(48) % Static analysis of node A
(49) ZeroA = Dot( A.GetResultantForce(), [Nx>; Ny>] ) % Creates 2x1 matrix
-> (50) ZeroA = [FAC + FAx + cos(theta)*FAB; FAy + sin(theta)*FAB]

(51) Explicit( ZeroA ) % Ensure results are explicit in W, L, theta, etc.
-> (52) ZeroA = [FAC + cos(theta)*FAB; 0.5*W + sin(theta)*FAB]

(53) Solve( ZeroA, FAB, FAC )
-> (54) FAB = -0.5*W/sin(theta)
-> (55) FAC = 0.5*W*cos(theta)/sin(theta)

```



8.3 Four-bar linkage – static equilibrium (see dynamics in Section 9.20)

The figure to the right shows a planar four-bar linkage consisting of frictionless-pin-connected uniform rigid links A , B , C and ground N .

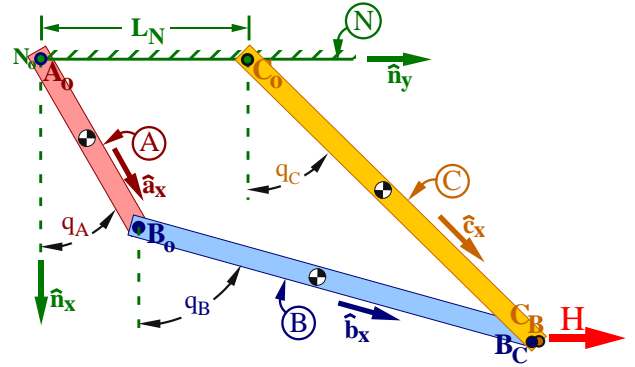
- Link A connects to N and B at points A_o and A_B
- Link B connects to A and C at points B_o and B_C
- Link C connects to N and B at points C_o and C_B
- Point N_o of N is coincident with A_o
- Point N_C of N is coincident with C_o

Right-handed orthogonal unit vectors $\hat{\mathbf{a}}_i$, $\hat{\mathbf{b}}_i$, $\hat{\mathbf{c}}_i$, $\hat{\mathbf{n}}_i$ ($i = x, y, z$) are fixed in A , B , C , N , with:

- $\hat{\mathbf{a}}_x$ directed from A_o to A_B
- $\hat{\mathbf{b}}_x$ directed from B_o to B_C
- $\hat{\mathbf{c}}_x$ directed from C_o to C_B
- $\hat{\mathbf{n}}_x$ vertically-downward
- $\hat{\mathbf{n}}_y$ directed from N_o to N_C
- $\hat{\mathbf{a}}_z = \hat{\mathbf{b}}_z = \hat{\mathbf{c}}_z = \hat{\mathbf{n}}_z$ parallel to pin axes

Create the following “**loop equation**” and dot-product with $\hat{\mathbf{n}}_x$ and $\hat{\mathbf{n}}_y$.

$$L_A \hat{\mathbf{a}}_x + L_B \hat{\mathbf{b}}_x - L_C \hat{\mathbf{c}}_x - L_N \hat{\mathbf{n}}_y = \vec{\mathbf{0}}$$



Quantity	Symbol	Value
Length of link A	L_A	1 m
Length of link B	L_B	2 m
Length of link C	L_C	2 m
Distance between N_o and N_C	L_N	1 m
Mass of A	m^A	10 kg
Mass of B	m^B	20 kg
Mass of C	m^C	20 kg
Earth's gravitational acceleration	g	$9.81 \frac{\text{m}}{\text{s}^2}$
$\hat{\mathbf{n}}_y$ measure of force applied to C_B	H	200 N
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{a}}_x$ with $+\hat{\mathbf{n}}_z$ sense	q_A	Variable
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{b}}_x$ with $+\hat{\mathbf{n}}_z$ sense	q_B	Variable
Angle from $\hat{\mathbf{n}}_x$ to $\hat{\mathbf{c}}_x$ with $+\hat{\mathbf{n}}_z$ sense	q_C	Variable

Complete the following **MG road-map** to determine this systems’s **static configuration**.

Make a “cut” between points B_C and C_B and introduce a constraint force $\vec{\mathbf{F}}^{C_B}$ on C_B from B_C .

Variable	Translate/Rotate	Direction (unit vector)	System S	FBD of S	About point	MG road-map equation	Additional Unknowns
q_A	Rotate	$\hat{\mathbf{n}}_z$	A, B	Draw	A_o	$\hat{\mathbf{a}}_x \cdot \vec{\mathbf{M}}^{S/A_o} = 0$	$F_x^{C_B}, F_y^{C_B}$
q_B	Rotate	$\hat{\mathbf{n}}_z$	B	Draw	B_o	$\hat{\mathbf{a}}_y \cdot \vec{\mathbf{M}}^{B/B_o} = 0$	$F_x^{C_B}, F_y^{C_B}$
q_C	Rotate	$\hat{\mathbf{n}}_z$	C	Draw	C_o	$\hat{\mathbf{a}}_y \cdot \vec{\mathbf{M}}^{C/C_o} = 0$	$F_x^{C_B}, F_y^{C_B}$
* Additional scalar constraint equation:				$-L_A \sin(q_A) \dot{q}_A - L_B \sin(q_B) \dot{q}_B + L_C \sin(q_C) \dot{q}_C = 0$			
* Additional scalar constraint equation:				$L_A \cos(q_A) \dot{q}_A + L_B \cos(q_B) \dot{q}_B - L_C \cos(q_C) \dot{q}_C = 0$			

Hint: Consider clever replacement of gravity forces for more efficient MG road-maps.

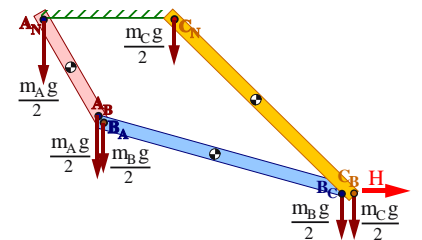
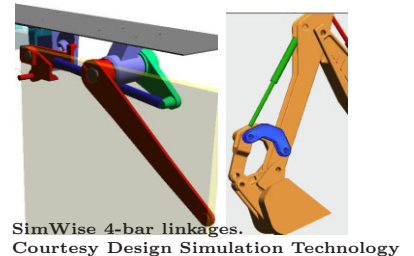
Using the **MG road-map**, determine the **static equilibrium** values of q_A , q_B , q_C . Using your intuition (guess), circle the **stable** solution.

Solution 1	$q_A \approx 20.0^\circ$	$q_B \approx 71.7^\circ$	$q_C \approx 38.3^\circ$
Solution 2	$q_A \approx 249.3^\circ$	$q_B \approx 140.2^\circ$	$q_C \approx 199.1^\circ$
Solution 3	$q_A \approx 30.7^\circ$	$q_B \approx 226.1^\circ$	$q_C \approx 254.7^\circ$

Determine at least one solution. Solutions are for $H = 200$ N.

Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow **Four-bar linkage**

Efficient replacement of gravity forces: **Real** gravity forces are complicated as there is a gravity force on each of the $\approx 6.02 \times 10^{23}$ particles of the rod. Although it is common to replace the **real** gravitational forces on each rod with “ $m g$ ” at each link’s center of mass, an efficient alternative replaces gravity forces on each rod with half the gravity force at each end as shown right.



Form the 3 generalized forces $\mathcal{F}_{\dot{q}_A}$, $\mathcal{F}_{\dot{q}_B}$, $\mathcal{F}_{\dot{q}_C}$ when the linkage is “cut” between points B_C and C_B so there is a constraint force $\vec{F}^{C_B} = F_x \hat{n}_x + F_y \hat{n}_y$ on C_B from B_C .

Form **Kane/Lagrange** statics equations with these 3 generalized forces, augmenting by the 2 scalar constraint (loop) equations.

Result: (replace gravity forces as before and use ${}^N\vec{v}^{B_C} = L_C \dot{q}_C \hat{c}_y$)

$$\mathcal{F}_{\dot{q}_A} = 0 = L_A [F_x \sin(q_A) - F_y \cos(q_A) - 0.5(m^A + m^B)g \sin(q_A)]$$

$$\mathcal{F}_{\dot{q}_B} = 0 = L_B [F_x \sin(q_B) - F_y \cos(q_B)]$$

$$\mathcal{F}_{\dot{q}_C} = 0 = L_C [H \cos(q_C) + F_y \cos(q_C) - F_x \sin(q_C) - 0.5(m^B + m^C)g \sin(q_C)]$$

$$\text{Scalar constraint (loop) equation: } -L_A \sin(q_A) \dot{q}_A - L_B \sin(q_B) \dot{q}_B + L_C \sin(q_C) \dot{q}_C = 0$$

$$\text{Scalar constraint (loop) equation: } L_A \cos(q_A) \dot{q}_A + L_B \cos(q_B) \dot{q}_B - L_C \cos(q_C) \dot{q}_C = 0$$

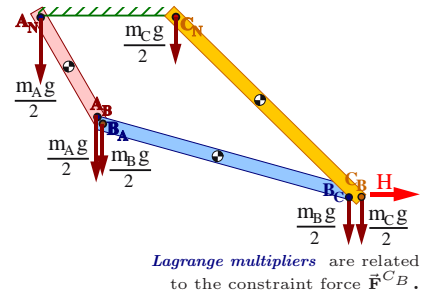
Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow Four-bar linkage

Form this system’s generalized force $\mathcal{F}_{\dot{q}_A}$ via **Kane/Lagrange**. To use the “embedded method” which accounts for constraints and eliminates all constraint/reaction forces, differentiate the loop equation and solve for \dot{q}_B and \dot{q}_C in terms of \dot{q}_A (shown right).

Result: (replace gravity forces as before and use ${}^N\vec{v}^{B_C} = L_C \dot{q}_C \hat{c}_y$)

$$\mathcal{F}_{\dot{q}_A} = \frac{-L_A}{2} \left\{ (m^A + m^B)g \sin(q_A) + \frac{\sin(q_A - q_B)}{\sin(q_B - q_C)} [2H \cos(q_C) - (m^B + m^C)g \sin(q_C)] \right\} = 0$$

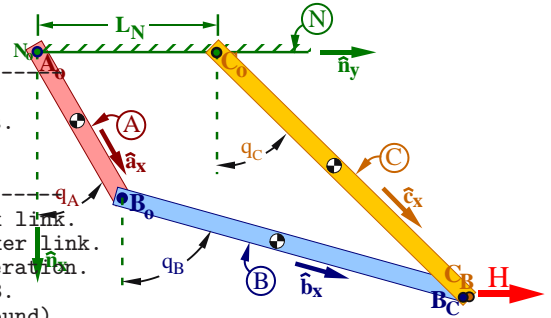
Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow Four-bar linkage



$$\dot{q}_B = \frac{-L_A \sin(q_A - q_C)}{L_B \sin(q_B - q_C)} \dot{q}_A$$

$$\dot{q}_C = \frac{-L_A \sin(q_A - q_B)}{L_C \sin(q_B - q_C)} \dot{q}_A$$

```
% MotionGenesis file: MGFourBarStaticsKaneEmbedded.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N % Ground link.
RigidBody A, B, C % Crank, coupler, rocker links.
Point BC( B ) % Point of B connected to C.
Point CB( C ) % Point of C connected to B.
%-----
Constant LN = 1 m, LA = 1 m % Length of ground link, crank link.
Constant LB = 2 m, LC = 2 m % Length of coupler link, rocker link.
Constant g = 9.81 m/s^2 % Earth's gravitational acceleration.
Constant H = 200 Newtons % Horizontal force at point CB.
Variable qA', qB', qC' % Link angles (relative to ground).
SetGeneralizedSpeed( qA' )
%-----
A.SetMass( mA = 10 kg )
B.SetMass( mB = 20 kg )
C.SetMass( mC = 20 kg )
%-----
% Rotational kinematics.
A.RotateZ( N, qA )
B.RotateZ( N, qB )
C.RotateZ( N, qC )
%-----
% Translational kinematics.
Bo.SetPositionVelocity( No, LA*Ax> )
CB.SetPositionVelocity( No, LN*Ny> + LC*Cx> )
%-----
% Add relevant forces (replace gravity forces with equivalent set).
Bo.AddForce( 0.5*(mA+mB)*g*Nx> )
CB.AddForce( 0.5*(mB+mC)*g*Nx> + H*Ny> )
%-----
% Loop (configuration) constraint.
Loop> = LA*Ax> + LB*Bx> - LC*Cx> - LN*Ny>
Loop[1] = Dot( Loop>, Nx> )
Loop[2] = Dot( Loop>, Ny> )
%-----
% For Kane's embedded statics method, solve qB', qC' in terms of qA'.
Solve( Dt(Loop) = 0, qB', qC' )
%-----
% Statics equation by setting generalized force to 0.
Statics = System.GetStaticsKane()
%-----
```



```
% Solve statics with constraints (nonlinear algebraic equations need a guess for its solution).
Solve( [Statics; Loop] = 0,    qA = 30 deg,  qB = 60 deg,  qC = 30 deg )
%-----
Save MGFourBarStaticsKaneEmbedded.html
Quit
```


Chapter 9

Equations of motion

9.1 MotionGenesis statics and dynamics commands

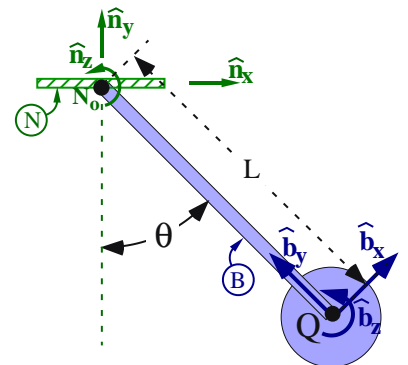
Command	Description
S.GetStatics()	Forms the resultant of all forces on point, particle, body, or system S.
S.GetStatics(P)	Forms the moment of all forces on S about point P.
S.GetDynamics()	Forms $\vec{F} = m\vec{a}$ for particle, body, or systems S.
S.GetDynamics(P)	Forms Euler's equation for S about point P.
System.GetDynamicsKane()	Forms Kane's equation for the System.
System.GetDynamicsLagrange()	Forms Lagrange's equation for the System.

9.2 Motion simulation of classic particle pendulum

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Pendulum.



```
% File: ClassicParticlePendulumEuler.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N      % Newtonian reference frame (ground)
RigidFrame      B      % Massless inextensible (rigid) string
Particle        Q      % Particle at end of string
%-----
Variable theta''      % Pendulum angle
Constant L = 50 cm     % Length of string
Constant g = 9.8 m/s^2 % Earth's gravitational acceleration
Q.SetMass( m = 2 kg )
%-----
% Rotational/translational kinematics and relevant forces.
B.RotateZ( N, theta )
Q.Translate( No, -L*By> )
Q.AddForceGravity( -g*Ny> )
%-----
% Equations of motion (angular momentum principle)
Zero> = System.GetDynamics( No )
Zero = Dot( Zero>, Nz> )
Solve( Zero, theta'' )
%-----
KE = System.GetKineticEnergy()
PE = Q.GetForceGravityPotentialEnergy( -g*Ny>, No )
MechanicalEnergy = KE + PE
%-----
% Integration parameters and initial values of variables.
Input tFinal = 10 sec, tStep = 0.02 sec, absError = 1.0E-08
Input theta = 30 deg, theta' = 0 deg/sec
%-----
% List output quantities and solve ODEs.
Output t sec, theta deg, theta' deg/sec, KE Joules, PE Joules, MechanicalEnergy Joules
ODE() ClassicParticlePendulumEuler
Save ClassicParticlePendulumEuler.all
Quit
```



Alternatively, Kane's equations for the classic particle pendulum

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Pendulum.

```
(1) % File: ClassicParticlePendulumKane.txt
(2) % Problem: Equation of motion for pendulum
(3) % Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
(4) %-----
(5) NewtonianFrame N % Newtonian reference frame (ground)
(6) RigidFrame B % Massless inextensible (rigid) string
(7) Particle Q % Particle at end of string
(8) %-----
(9) Variable theta'' % Pendulum angle
(10) Constant L = 50 cm % Length of string
(11) Constant g = 9.8 m/s^2 % Earth's gravitational acceleration
(12) Q.SetMass( m = 2 kg )
(13) SetGeneralizedSpeed( theta' )
(14) %-----
(15) % Rotational and translational kinematics
(16) B.RotateZ( N, theta )
-> (17) B_N = [cos(theta), sin(theta), 0; -sin(theta), cos(theta), 0; 0, 0, 1]
-> (18) w_B_N> = theta'*Bz>
-> (19) alf_B_N> = theta''*Bz>

(20) Q.Translate( No, -L*By> )
-> (21) p_No_Q> = -L*By>
-> (22) v_Q_N> = L*theta'*Bx>
-> (23) a_Q_N> = L*theta''*Bx> + L*theta'^2*By>

(24) %-----
(25) % Relevant forces
(26) Q.AddForceGravity( -g*Ny> )
-> (27) Force_Q> = -m*g*Ny>

(28) %-----
(29) % Equations of motion
(30) Dynamics = System.GetDynamicsKane()
-> (31) Dynamics = [m*L*(g*sin(theta)+L*theta'')]

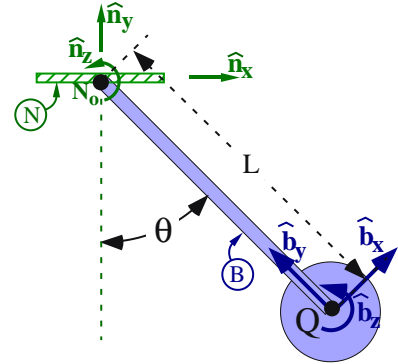
(32) Solve( Dynamics, theta'' )
-> (33) theta'' = -g*sin(theta)/L

(34) %-----
(35) % Kinetic and potential energy
(36) KE = System.GetKineticEnergy()
-> (37) KE = 0.5*m*L^2*theta'^2

(38) PE = Q.GetForceGravityPotentialEnergy( -g*Ny>, No )
-> (39) PE = -m*g*L*cos(theta)

(40) MechanicalEnergy = KE + PE
-> (41) MechanicalEnergy = PE + KE

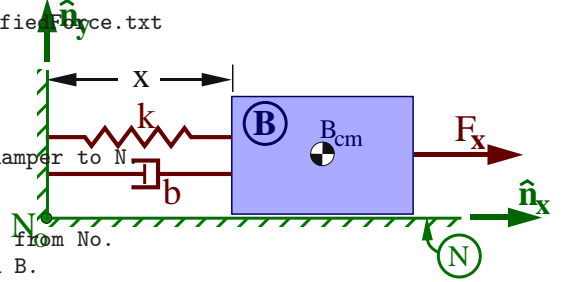
(42) %-----
(43) % Integration parameters and initial values of variables.
(44) Input tFinal = 10 sec, tStep = 0.02 sec, absError = 1.0E-08
(45) Input theta = 30 deg, theta' = 0 deg/sec
(46) %-----
(47) % List output quantities and solve ODEs.
(48) Output t sec, theta deg, theta' deg/sec, KE Joules, PE Joules, MechanicalEnergy Joules
(49) ODE() ClassicParticlePendulumKane
```



9.3 Motion simulation of a block on a rough surface



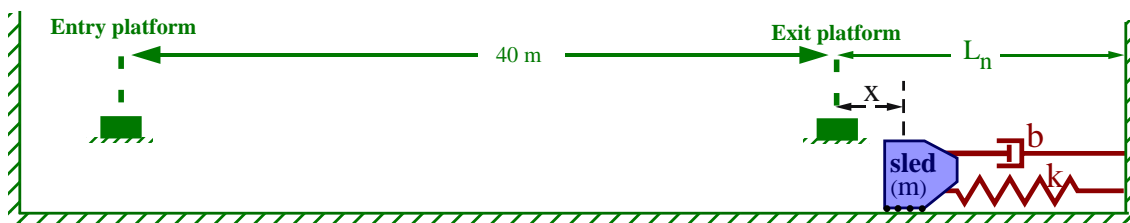
```
% MotionGenesis file: MGStickSlipMassSpringDamperPulledBySpecifiedForce.txt
% Copyright (c) 2017 Motion Genesis LLC. All rights reserved.
% Purpose: Simulate mass-spring-damper on rough table.
%-----
NewtonianFrame N % Ground (Earth).
Particle B % Block connected by spring/damper to N
B.SetMass( m = 1 kg )
%-----
Variable x'' % B's horizontal displacement from No.
Variable Fn % Resultant normal force on B.
Variable Ff % Resultant friction force on B.
Constant g = 9.8 m/s^2 % Earth's gravitational acceleration.
Constant k = 400 N/m % Linear spring constant.
Constant Ln = 4 m % Natural length of spring.
Constant b = 4 N*s/m % Linear damper constant (zeta = 0.1).
Constant epsilonV = 1.0E-5 m/s % For Continuous Friction law, small sliding speed.
Constant muK = 0.4 noUnits % Coefficient of kinetic friction.
Specified Fx = 20*cos(t) % Specified horizontal force on B.
%-----
% Position, velocity, acceleration
B.Translate( No, x*Nx> )
%-----
% Forces on B
B.AddForce( -m*g*Ny> ) % Gravitational force
stretch = x - Ln % Spring stretch.
B.AddForce( -(k*stretch + b*Dt(stretch))*Nx> ) % Spring/damper force
B.AddForce( Fn*Ny> + Ff*Nx> ) % Normal and friction forces
B.AddForce( Fx*Nx> ) % Specified force
%-----
% Equations of motion for B via F = m*a
Dynamics[1] = Dot( Nx>, B.GetDynamics() )
Dynamics[2] = Dot( Ny>, B.GetDynamics() )
%-----
% Equation governing Ff when B is sliding on N.
% Note: Use Continuous Friction Law to simulate sticking and sliding.
% Set epsilonV to a small positive number to avoid divide-by-zero problems.
magVelocity = B.GetSpeed( N )
magVelocityPlusEpsilon = magVelocity + epsilonV
Ff = Dot( -muK*Fn*B.GetVelocity(N) / magVelocityPlusEpsilon, Nx> )
%-----
% Solve sliding equation of motion for x''
Solve( Dynamics = 0, x'', Fn )
%-----
% Integration parameters and initial values.
Input tFinal = 12 sec, tStep = 0.01 sec, absError = 1.0E-07
Input x = Input(Ln) m, x' = 0 m/s
%-----
% List output quantities and solve ODEs.
Output t sec, x m, x' m/s, x'' m/s^2, Ff Newton
ODE() MGStickSlipMassSpringDamperPulledBySpecifiedForce
%-----
Save MGStickSlipMassSpringDamperPulledBySpecifiedForce.html
Quit
```



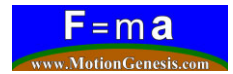
9.4 Equations of motion for a rocket sled ride



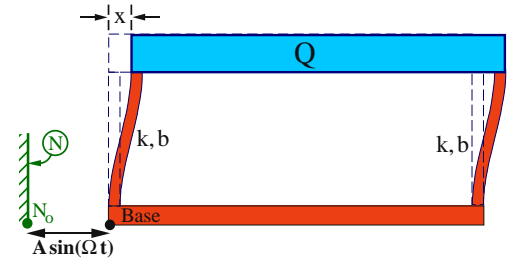
```
% File: DisneyRideEquationOfMotion.al
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
Particle      Q          % Rocket-sled
Variable      x''        % Nx> measure of Q's position from No
Constant  k = 139.2 N/m, b = 300 N*s/m % Spring/damper constants
Constant  Ln          % Spring's natural length
Q.SetMass( m = 200 kg)
%-----
%      Q's position vector, velocity, and acceleration
Q.Translate( No, x*Nx> )
%-----
%      Add relevant forces (spring and damper).
SpringLength = Ln - x
SpringStretch = SpringLength - Ln
Q.AddForce( k*Explicit(SpringStretch)*Nx> )
Q.AddForce( -b*Q.GetVelocity(N) )
%-----
%      Form Newton's equation of motion (translation)
Zero = Dot( Q.GetDynamics(), Nx> )
Solve( Zero, x'' )
%-----
%      Initial values of x, x'
Input  x = -40 meters,  x' = 0 m/s
%-----
%      Initial acceleration for given adult values
InitialAdultAcceleration> = EvaluateAtInput( Q.GetAcceleration(N) )
InitialAdultAccelerationMag = GetMagnitude( InitialAdultAcceleration> )
NumberOfGsForAdult = InitialAdultAccelerationMag / 9.8
%-----
%      Initial acceleration for given child values
InitialChildAcceleration> = EvaluateAtInput( Q.GetAcceleration(N), m = 100 kg )
NumberOfGsForChild = GetMagnitude( InitialChildAcceleration> ) / 9.8
%-----
%      Plot x(t) for 10 seconds.
Input  tFinal = 10 seconds
OutputPlot  t sec, x meters
Ode() DisneyRideEquationOfMotion
%-----
%      Calculate zeta and wn for both adult and child
zeta = b / (2*sqrt(m*k) );    wn = sqrt(k/m)
zetaAdult = EvaluateAtInput(zeta);    wnAdult = EvaluateAtInput(wn)
zetaChild = EvaluateAtInput(zeta,m=100);    wnChild = EvaluateAtInput(wn,m=100)
%-----
%      Save results
Save DisneyRideEquationOfMotion.all
Quit
```



9.5 Motion simulation of a building in an earthquake

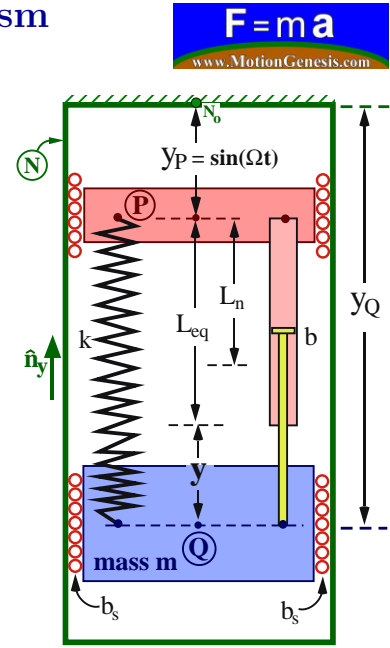


```
% File: SingleStoryBuildingInEarthquake.al
% Problem: Forced base motion of building
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N      % Newtonian reference frame
Point      Base      % Base of building (modeled as a point)
Particle    Q        % Roof of building (modeled as a particle)
%-----
Variable  x''          % Spring stretch
Constant  k = 10000 N/m % Linear spring constant
Constant  b = 500 N*sec/m % Linear damping constant
Constant  Amp = 0.1 m   % Earthquake amplitude
Q.SetMass( m = 5000 kg )
wn = EvaluateAtInput( sqrt(2*k/m) ) % Building's natural frequency
Constant  Omega = 0.8*wn rad/sec    % Earthquake forcing frequency
%-----
%      Position vectors, velocities, and accelerations
Base.Translate( No, Amp*sin(Omega*t)*Nx> )
Q.Translate( Base, x*Nx> )
%-----
%      Spring and damper forces
SpringForce> = -2*k*x*Nx>
DamperForce> = -2*b*x'*Nx>
Q.AddForce( SpringForce> + DamperForce> )
%-----
%      Form equations of motion via F = m*a and solve for x''
Zero = Dot( Q.GetDynamics(), Nx> )
Solve( Zero, x'' )
%-----
%      Integration parameters and initial values of variables.
Input  tFinal = 40 sec, tStep = 0.02 sec, absError = 1.0E-08
Input  x = 0 m, x' = 0 m/sec
%-----
%      List output quantities and solve ODEs.
Output t sec, x m, x' m/s, x'' m/s^2
ODE() SingleStoryBuildingInEarthquake
%-----
%      Record input and program responses
Save SingleStoryBuildingInEarthquake.all
Quit
```



9.6 Equation of motion of Scotch-yolked mechanism

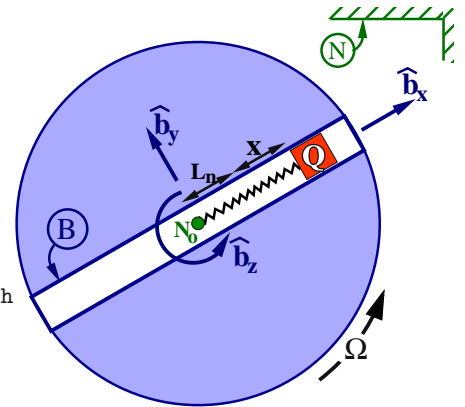
```
% File: MassSpringDamperVerticalForcedScotchYolk.al
% Problem: Forced harmonic motion of mass/spring/damper system.
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N % Newtonian reference frame
Point P % Platform (modeled as a moving point)
Particle Q % Roof of building (modeled as a moving particle)
%-----
Variable y'' % Spring stretch (from natural length)
Specified yP'' % Known vertical motion of Scotch-yolk
Constant k % Linear spring/damper spring constant
Constant Ln % Natural length of spring
Constant Leq % Equilibrium length of spring
Constant b % Linear spring/damper damping constant
Constant bs % Linear viscous damping constant between Q and N
Constant g % Local gravitational acceleration
Q.SetMass( m )
%-----
% Position vectors, velocities, and accelerations
P.Translate( No, -yP*Ny> )
Q.Translate( P, -(y+Leq)*Ny> )
%-----
% Relevant contact/distance forces
SpringStretch = y + Leq - Ln
SpringForce> = k * Explicit(SpringStretch) * Ny>
DamperForce> = b * Dt(SpringStretch) * Ny>
ViscousForce> = -2 * bs * Q.GetVelocity(N)
GravityForce> = -Q.GetMass() * g *Ny>
Q.AddForce( SpringForce> + DamperForce> + ViscousForce> + GravityForce> )
%-----
% Downward measure of Newton's equation of motion
Zero = Dot( Q.GetDynamics(), -Ny> )
%-----
% Use static equilibrium to simplify equation of motion
ZeroStatics = Evaluate( Zero, y=0, y'=0, y''=0, yP'=0, yP''=0 )
ZeroDynamics = Explicit( Zero - ZeroStatics )
%-----
% Examine bs = 0 and harmonic forcing for yP
Constant A, Omega % Amplitude/frequency of Scotch-yolk vertical motion
SetDt( yP = A*sin(Omega*t) )
ZeroHarmonicForcing = Evaluate( ZeroDynamics, bs=0, yP'' = -A*Omega^2*sin(Omega*t) )
FactorLinear( ZeroHarmonicForcing, y'', y', y )
%-----
% Record input and program responses
Save MassSpringDamperVerticalForcedScotchYolk.all
Quit
```



9.7 Equation of motion for a particle on spinning slot

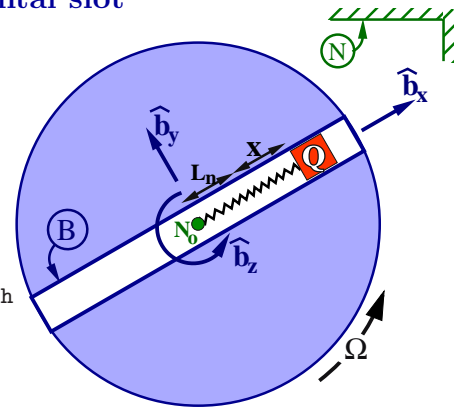


```
% File: ParticleOnSpinningSlotFMA.al
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
RigidBody B
Particle Q
%-----
Variable x'' % Bx> measure of Q's position from No
Specified Omega' % Bz> measure of B's angular velocity in N
Constant b, k, Ln % Damping constant. Spring constant/natural length
Q.SetMass( m )
%-----
% Rotational and translational kinematics
B.SetAngularVelocityAcceleration( N, Omega*Bz> )
Q.Translate( No, (Ln+x)*Bx> )
%-----
% Add relevant forces (spring and damper forces).
Q.AddForce( -(k*x + b*x')*Bx> )
%-----
% Relevant translational/rotational equations of motion
Zero = Dot( Q.GetDynamics(), Bx> )
%-----
% Record input together with responses
Save ParticleOnSpinningSlotFMA.all
Quit
```



Alternatively, Kane's equations for a particle in a horizontal slot

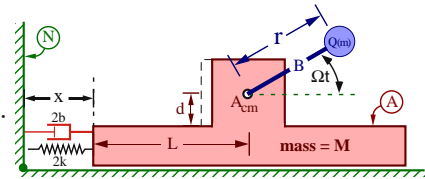
```
% File: ParticleOnSpinningSlotKane.al
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
RigidBody B
Particle Q
%-----
Variable x'' % Bx> measure of Q's position from No
Specified Omega' % Bz> measure of B's angular velocity in N
Constant b, k, Ln % Damping constant. Spring constant/natural length
Q.SetMass( m )
%-----
% Rotational and translational kinematics
B.SetAngularVelocityAcceleration( N, Omega*Bz> )
Q.Translate( No, (Ln+x)*Bx> )
Bcm.SetVelocity( N, 0> )
%-----
% Add relevant forces (spring and damper forces).
Q.AddForce( -(k*x + b*x')*Bx> )
%-----
% Form equation of motion
SetGeneralizedSpeed( x' )
Zero = System.GetDynamicsKane()
Solve( Zero, x'' )
%-----
% Record input together with responses
Save ParticleOnSpinningSlotKane.all
Quit
```



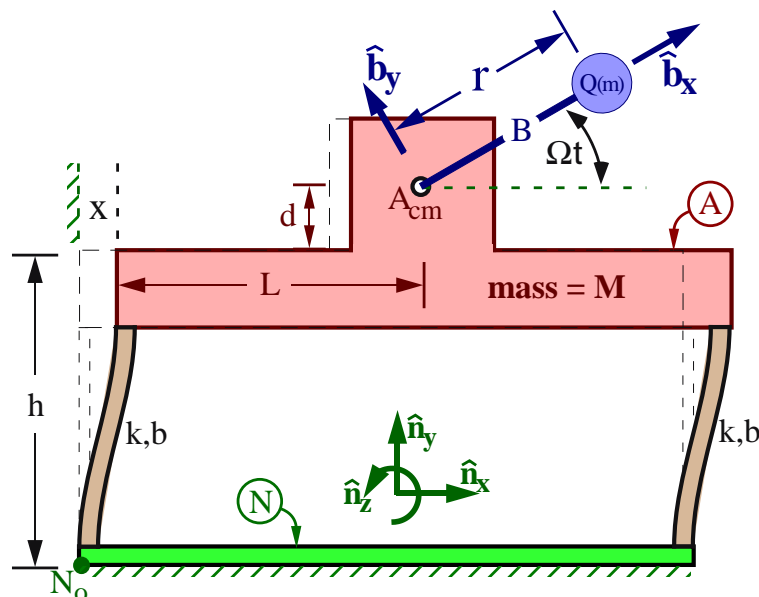
9.8 Equation of motion for unbalanced motor on roof



```
% File: EccentricParticleAirConditionerForcedVibrationFBD1.al
% Purpose: Determine the effect on the motion of a one-story building
% of an air-conditioner that is mounted on the roof - and
% which has an unbalance motor, modeled as an eccentric particle.
```



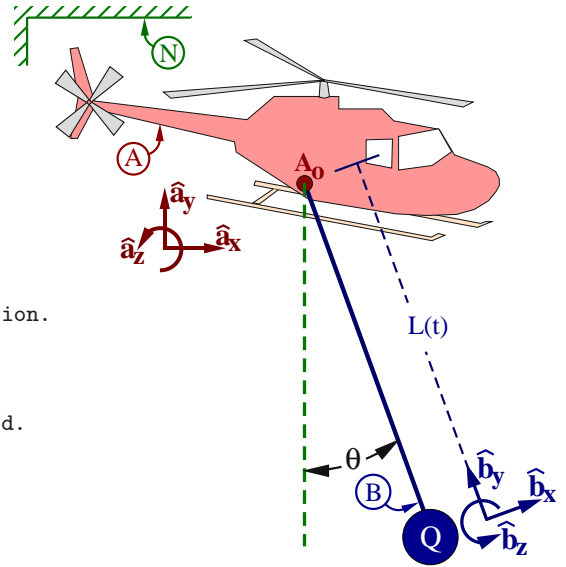
```
NewtonianFrame N
RigidFrame B % Air-conditioner motor Frame
Particle A % The roof modeled as a particle (no rotation)
Particle Q % Eccentric particle
%-----
Variable x'' % Horizontal displacement of A
Constant g % Local gravitational acceleration
Constant k % Effective stiffness in each column
Constant b % Effective damping in each column
Constant h % Height of roof
Constant d % Height of motor about roof
Constant L % Distance from left edge of roof to air-conditioner
Constant r % Eccentric distance (from A to Q)
Constant W % Motor spin rate
A.SetMass( mA )
Q.SetMass( mQ )
%-----
% Rotational and translational kinematics.
B.RotateZ( N, W*t )
A.Translate( No, (x+L)*Nx> + (h+d)*Ny> )
Q.Translate( A, r*bx> )
%-----
% Relevant forces on system A, B, Q
Q.AddForce( -mQ*g*Ny> ) % Gravity force on Q
SpringForce> = -2*k*x*Nx> % Spring force on A
DamperForce> = -2*b*x'*Nx> % Damper force on A
A.AddForce( SpringForce> + DamperForce> )
%-----
% Equation of motion for system A, B, Q
Zero = Dot( Nx>, System(A,Q).GetDynamics() )
Solve( Zero, x'' )
%-----
Save EccentricParticleAirConditionerForcedVibrationFBD1.all
Quit
```



9.9 Motion simulation of helicopter rescue



```
% MotionGenesis file: StationaryHelicopterRetrievalFma.txt
% Problem: Retrieval of capsized fishermen.
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N          % Earth.
RigidFrame B              % Cable.
Particle Q                % Rescue basket and fishermen.
%-----
Q.SetMass( m = 100 kg )
Constant g = 9.8 m/s^2    % Earth's gravitational acceleration.
Variable theta''          % Pendulum swing angle.
Variable Tension          % Tension in cable.
Constant L0 = 50 m        % Initial cable length.
Constant s = 2 m/s        % Rate at which cable is retrieved.
Specified L''             % Cable length (varies).
SetDt( L = L0 - s*t )
%-----
% Rotation and translation kinematics.
B.RotateZ( N, theta )
Q.Translate( No, -L*By> )
%-----
% Add relevant contact and distance forces.
Q.AddForce( -m*g*Ny> + Tension*By> )
%-----
% Form equations of motion using F = m * a.
Dynamics = Dot( Bx>, Q.GetDynamics() )
Solve( Dynamics = 0, theta'' )
%-----
% Input initial values and numerical integration parameters.
Input theta = 1 deg, theta' = 0 deg/sec
Input tFinal = 24.92 sec, tStep = 0.02 sec
%-----
% List output quantities and solve ODEs.
OutputPlot t sec, theta deg
ODE() StationaryHelicopterRetrievalFma
%-----
Save StationaryHelicopterRetrievalFma.html
Quit
```



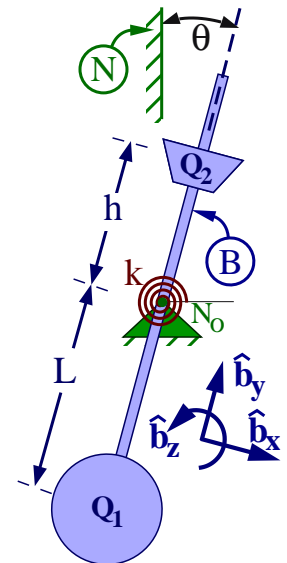
Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Helicopter rescue.

9.10 Equations of motion of a metronome

Related problem at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ pendulums.



```
% MotionGenesis file: MGMetronomePendulumEulerLagrangeKane.txt
% Problem: Equation of motion for two-particle metronome.
% Copyright (c) 2017 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N          % Earth with point No at metronome pivot.
RigidFrame B              % Metronome rod.
Particle Q1, Q2           % Particles fixed at distal ends of B.
%-----
Variable theta''          % Metronome angle and 1st/2nd derivative.
Constant L = 0.2 m        % Distance between point No and Q1.
Constant h = 0.1456 m     % Distance between point No and Q2.
Constant g = 9.8 m/s^2    % Earth's gravitational acceleration.
Constant k = 2 N*m/rad    % Torsional spring constant.
Q1.SetMass( m1 = 0.1 kg )
Q2.SetMass( m2 = 0.02 kg )
%-----
% Rotational and translational kinematics.
B.RotateNegativeZ( N, theta )
Q1.Translate( No, -L*By> )
Q2.Translate( No, h*By> )
%-----
% Relevant forces and torques.
System.AddForceGravity( -g*Ny> )
B.AddTorque( k*theta*Bz> )
%-----
% Euler's equation of motion (angular momentum principle about No).
Euler = Dot( System.GetDynamics(No), -Bz> )
Factor( Euler, theta'', g )
%-----
% Conservation of mechanical energy.
KineticEnergy = System.GetKineticEnergy()
PotentialEnergyGravity = System.GetForceGravityPotentialEnergy( -g*Ny>, No )
PotentialEnergySpring = 1/2 * k * theta^2
PotentialEnergy = PotentialEnergyGravity + PotentialEnergySpring
MechanicalEnergy = KineticEnergy + PotentialEnergy
%-----
% Lagrange's equation of motion.
SetGeneralizedCoordinate( theta )
Lagrange = System.GetDynamicsLagrange( SystemPotential = PotentialEnergy )
Factor( Lagrange, theta'', g )
%-----
% Kane's equation of motion.
SetGeneralizedSpeed( theta' )
Kane = System.GetDynamicsKane()
Factor( Kane, theta'', g )
%-----
% Save input and program responses.
Save MGMetronomePendulumEulerLagrangeKane.html
Quit
```

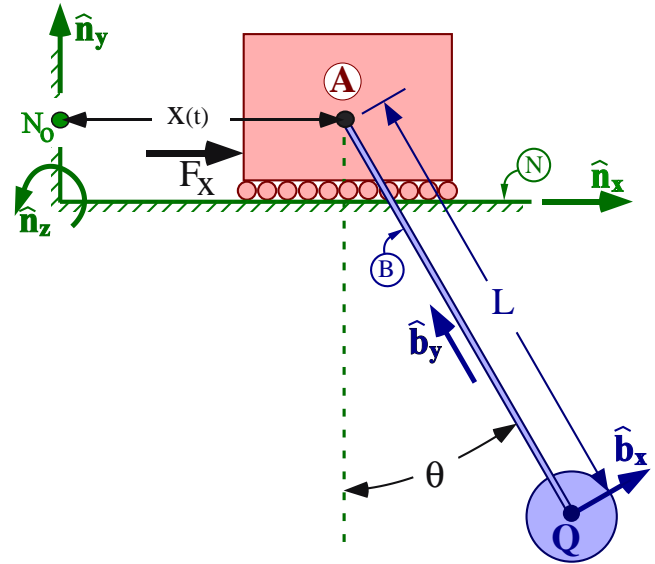


Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Pendulum–Metronome.

9.11 Equations of motion for a bridge crane



```
% File: BridgeCraneXTheta.al
%-----
NewtonianFrame N
Particle      A
RigidFrame    B
Particle      Q
%-----
A.SetMass( mA )
Q.SetMass( mQ )
%-----
Constant      g
Constant      L
Specified     x''
Variable      Fx
Variable      q''
%-----
B.RotateZ( N, q )
A.Translate( No, x*Nx> )
Q.Translate( A, -L*By> )
%-----
A.AddForce( Fx*Nx> )
System.AddForceGravity( -g*Ny> )
%-----
%      Kinetic energy, potential energy, and work done on system.
KE = System.GetKineticEnergy()
PE = System.GetForceGravityPotentialEnergy( -g*Ny>, No )
Variable WorkDoneBySystem' = Dot( Fx*Nx>, A.GetVelocity(N) )
EnergyConstant = KE + PE - WorkDoneBySystem
%-----
%      Lagrange's equations of motion
LhsLagrange = Dt( D( KE, q' ) ) - D( KE, q )
vQNPartial> = D( Q.GetVelocity(N), q', N )
RhsLagrange = Dot( vQNPartial>, Q.GetResultantForce() )
ZeroLagrange = Explicit( LhsLagrange - RhsLagrange )
%-----
%      Newton/Euler equations of motion
Zero[1] = Dot( Bz>, System(Q,B).GetDynamics(A) )
Zero[2] = Dot( Nx>, System(Q,B,A).GetDynamics() )
%-----
%      Solve equations of motion for q'' and Fx.
Solve( Zero, q'', Fx )
%-----
%      Record input together with responses
Save BridgeCraneXTheta.all
Quit
```

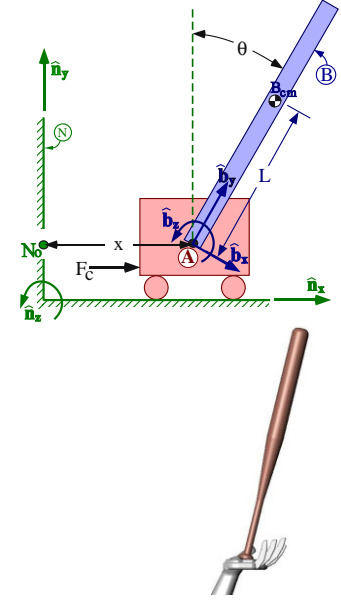


9.12 Dynamics and control of an inverted pendulum on cart

The figure to the right shows a rigid body B attached by an frictionless pin (revolute) joint to a cart A (modeled as a particle). The cart A slides on a horizontal frictionless track. The track is fixed in a Newtonian frame N .

Right-handed orthogonal unit vectors $\hat{n}_x, \hat{n}_y, \hat{n}_z$ and $\hat{b}_x, \hat{b}_y, \hat{b}_z$ are fixed in N and B respectively, with:

- \hat{n}_x horizontally-right and \hat{n}_y vertically-upward
- $\hat{n}_z = \hat{b}_z$ parallel to B 's axis of rotation in N
- \hat{b}_y directed from A to the distal end of B



Quantity	Symbol	Value
Mass of A	m^A	10.0 kg
Mass of B	m^B	1.0 kg
Distance between A and B_{cm} (B 's center of mass)	L	0.5 m
B 's moment of inertia about B_{cm} for \hat{b}_z	I_{zz}	0.08333 kg*m ²
Earth's gravitational constant	g	9.8 m/s ²
\hat{n}_x measure of feedback-control force applied to A	F_c	Specified
\hat{n}_x measure of A 's position from N_o (a point fixed in N)	x	Variable
Angle from \hat{n}_y to \hat{b}_y with $-\hat{n}_z$ sense	θ	Variable

- Form equations of motion for the system.

Result:

$$(m^A + m^B) \ddot{x} + m^B L \cos(\theta) \ddot{\theta} - m^B L \sin(\theta) \dot{\theta}^2 = F_c$$

$$m^B L \cos(\theta) \ddot{x} + (I_{zz} + m^B L^2) \ddot{\theta} - m^B g L \sin(\theta) = 0$$

- Consider the nominal solution $x = \dot{x} = \ddot{x} = \theta = \dot{\theta} = \ddot{\theta} = 0$.

Find F_{cnom} (the nominal value of F_c) required for this nominal solution to satisfy the equations of motion.

Result: The MotionGenesis output shows `Check = [-FcNominal; 0]`

which means for this solution, the nominal value of F_c is $F_{cnom} = 0$.

- After introducing the variables $dx, dx', dx'',$ and $dtheta, dtheta', dtheta''$ as perturbations of x and θ and their time-derivatives, linearize the equations of motion in perturbations about the aforementioned nominal solution.

Result:

$$\begin{bmatrix} m^A + m^B & m^B L \\ m^B L & I_{zz} + m^B L^2 \end{bmatrix} \begin{bmatrix} \ddot{x} \\ \ddot{\theta} \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & -m^B g L \end{bmatrix} \begin{bmatrix} x \\ \theta \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \end{bmatrix} [F_c]$$

- Put the linearized equations in the form $\dot{X} = A X + B F_c$ where A and B are 4×4 matrices and X is the 4×1 state matrix $[dx; dtheta; dx'; dtheta']$. A state-space controller specifies F_c as

$$F_c = k_1 x + k_2 \theta + k_3 \dot{x} + k_4 \dot{\theta} \quad \text{where } k_i (i=1, 2, 3, 4) \text{ are constants (feedback-control gains).}$$

Show that with $k_1 = k_2 = k_3 = k_4 = 0$,

the solution is **unstable**.

Show that with $k_1 = 1, k_2 = 244, k_3 = 5.4, k_4 = 59$,

the solution is **stable**.

Result: The MotionGenesis output shows the eigenvalues associated with the A matrix are

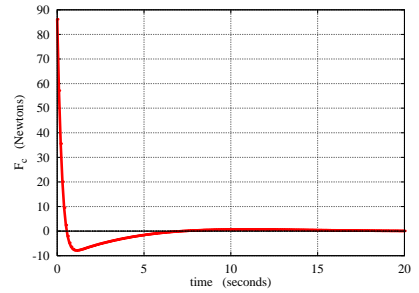
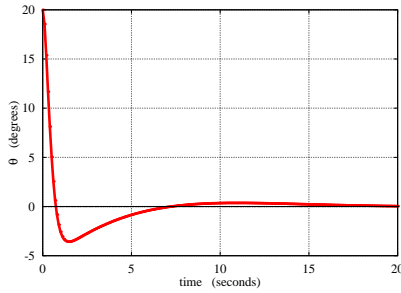
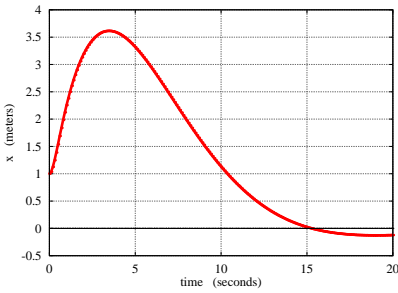
$$\text{RootsNoControl} \approx \begin{bmatrix} -4 & 0 & 0 & +4 \end{bmatrix} \quad \text{Since an eigenvalue is positive, the solution is } \mathbf{unstable}.$$

$$\text{RootsWithControl} \approx \begin{bmatrix} -3.8 - 1.3i & -3.8 + 1.3i & -0.22 - 0.2i & -0.22 + 0.2i \end{bmatrix}$$

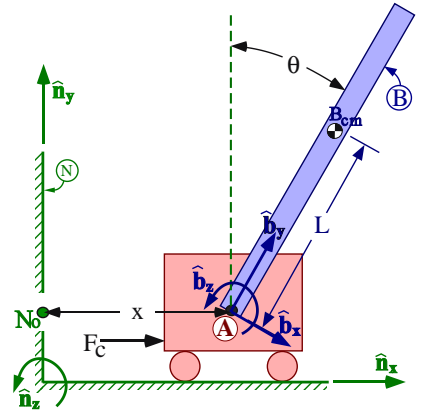
Since all eigenvalues for RootsWithControl have negative real parts, the solution is **stable** for "small" disturbances.

- Simulate the linearized and nonlinear controlled equations of motion for 20 sec. Use initial values $x = dx = 1$ m, $\theta = dtheta = 20^\circ$, $x' = dx' = 0$, $\theta' = dtheta' = 0$.

Plot x, θ , and F_c versus time.



```
% MotionGenesis file: InvertedPendulumOnCartWithControl.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
Particle A % Cart
RigidBody B % Inverted pendulum
%-----
Variable x'' % Distance between No to A
Variable theta'' % Angle from local vertical to B's long axis
Constant g+ = 9.81 m/s^2 % Gravitational constant
Constant L+ = 0.5 m % Distance between A and Bcm
Specified Fc
A.SetMass( mA = 10 kg )
B.SetMassInertia( mB = 1 kg, Izz = 1/12*mB*(2*L)^2, 0, Izz )
%-----
% Rotational and translational kinematics
B.RotateNegativeZ( N, theta )
A.Translate( No, x*Nx )
Bcm.Translate( A, L*By )
%-----
% Relevant contact and distance forces
System.AddForceGravity( -g*Ny )
A.AddForce( Fc*Nx )
%-----
% Form Kane's equations of motion
SetGeneralizedSpeed( x', theta' )
Zero = System.GetDynamicsKane()
```



```
%*****
% CONTROL SYSTEM / STABILITY ANALYSIS
%*****
% Linearization: Perturbation variables
Variable dx'' % Perturbations of x, x', x''
Variable dtheta'' % Perturbations of theta, theta', theta''
Specified dFc % Perturbation of Fc.
Variable FcNominal % Nominal solution for Fc.
SetImaginaryNumber( i )
%-----
% Check conditions for solution: x = x' = x'' = 0 and theta = theta' = theta'' = 0.
Check = Evaluate( Zero, x=0, x'=0, x''=0, theta=0, theta'=0, theta''=0, Fc=FcNominal )
%-----
% Linearize equations of motion about nominal solution
Perturb = Linearize1( Zero, x = 0 : dx, x' = 0 : dx', x'' = 0 : dx'', &
theta = 0 : dtheta, theta' = 0 : dtheta', theta'' = 0 : dtheta'', Fc = 0 : dFc )
Solve( Perturb, dx'', dtheta'' )
%-----
% Form matrix of perturbations and its time-derivative
Xm = [ dx ; dtheta ; dx' ; dtheta' ] % Matrix of perturbations
Xp = dt( Xm )
%-----
% Form/simplify matrices A and B so Xm' = A * Xm + B * Fc.
A = Expand( GetCoefficientMatrix( Xp, Xm ) )
B = Expand( GetCoefficientMatrix( Xp, dFc ) )
%-----
% Stability when uncontrolled (Fc = 0) and with control.
RootsNoControl = GetEigen( EvaluateToNumber( A ) )
```

```

%-----
%      Stability with feedback control (k1, k2, k3, k4 are gains).
Constant k1 = 1.0 N/m, k2 = 244 N/rad, k3 = 5.4 N*s/m, k4 = 59 N*s/rad
K = [k1, k2, k3, k4]
RootsControlled = GetEigen( EvaluateToNumber( A + B*K ) )
%-----
Fc = k1*x + k2*theta + k3*x' + k4*theta'
dFc = k1*dx + k2*dtheta + k3*dx' + k4*dtheta'
%-----
%      Integration parameters and initial values.
Input tFinal = 20, tStep = 0.1, absError = 1.0E-08
Input x = 1 m, theta = 20 deg, x' = 0 m/s, theta' = 0 rad/s
Input dx = 1 m, dtheta = 20 deg, dx' = 0 m/s, dtheta' = 0 rad/s
%-----
%      Quantities to be output by ODE command.
Output t sec, x m, dx m, dtheta deg, Fc Newtons
ODE( Zero, x'', theta'' ) InvertedPendulumOnCartWithControl
%-----
Save InvertedPendulumOnCartWithControl.html
Quit

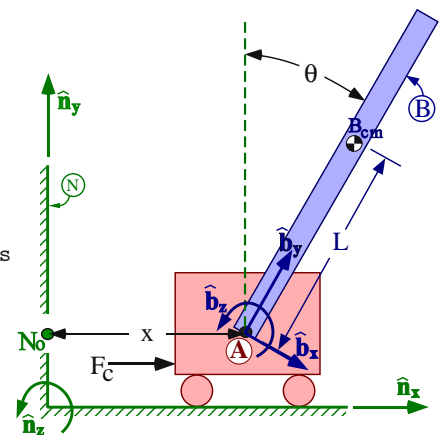
```

Alternatively, form equations of motion via Newton/Euler

```

% File: InvertedPendulumOnCartDynamics.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
Particle A % Cart
RigidBody B % Inverted pendulum
%-----
Variable x'' % Distance between No to A
Variable theta'' % Angle from local vertical to B's long axis
Specified Fc % Control force on cart
Constant g+ = 9.81 m/s^2 % Gravitational constant
Constant L+ = 0.5 m % Distance between A and Bcm
A.SetMass( mA = 10 kg )
B.SetMassInertia( mB = 1 kg, Izz = 1/12*mB*(2*L)^2, 0, Izz )
%-----
%      Rotational and translational kinematics
B.RotateNegativeZ( N, theta )
A.Translate( No, x*Nx> )
Bcm.Translate( A, L*By> )
%-----
%      Relevant contact and distance forces
System.AddForceGravity( -g*Ny> )
A.AddForce( Fc*Nx> )
%-----
%      Form and simplify equations of motion (via Newton/Euler)
EquationsOfMotion[1] = Dot( Nx>, System(A,B).GetDynamics() )
EquationsOfMotion[2] = Dot( -Bz>, B.GetDynamics(A) )
FactorLinear( EquationsOfMotion, theta'', x'', Fc, g )
%-----
Save InvertedPendulumOnCartDynamics.all

```



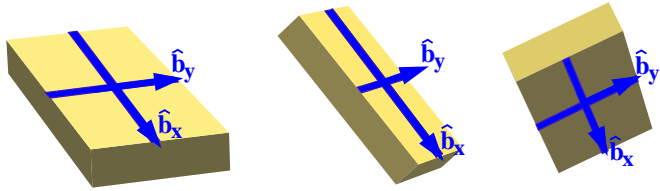
9.13 3D spin stability (application to Top-gun and Explorer I)



Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ 3D spin stability.

```
% MotionGenesis file: MGSpinStability3DRigidBody.txt
% Purpose: Spin stability of rigid body (books, footballs, aircraft).
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
RigidBody B
%-----
Variable wx', wy', wz' % Angular velocity measures
B.SetInertia( Bcm, Ixx = 1 kg*m^2, Iyy = 2 kg*m^2, Izz = 3 kg*m^2 )
%-----
% Rotational and translational kinematics.
B.SetAngularVelocityAcceleration( N, wx*Bx> + wy*By> + wz*Bz> )
Bcm.SetVelocity( N, 0> )
%-----
% Form equations of motion via angular momentum principle.
Dynamics[1] = Dot( Bx>, B.GetDynamics(Bcm) )
Dynamics[2] = Dot( By>, B.GetDynamics(Bcm) )
Dynamics[3] = Dot( Bz>, B.GetDynamics(Bcm) )
%-----
% Verify equations of motion via Kane's method.
SetGeneralizedSpeed( wx, wy, wz )
KaneDynamics = System.GetDynamicsKane()
ShouldBeZero = Dynamics - KaneDynamics
%-----
% Solve equations of motion for wx', wy', wz'.
Solve( Dynamics = 0, wx', wy', wz' )
%-----
% Numerical integration parameters and initial values.
Input tFinal = 4 sec, tStep = 0.02 sec, absError = 1.0E-08
Input wx = 0.2 rad/sec, wy = 7.0 rad/sec, wz = 0.2 rad/sec
%-----
% Form and list output quantities (to be plotted).
H> = B.GetAngularMomentum( Bcm )
Hmag = GetMagnitude( H> )
Wmag = B.GetAngularSpeed( N )
theta = acos( Dot( H>, B.GetAngularVelocity(N) ) / (Hmag * Wmag) )
OutputPlot t sec, wx rad/sec, wy rad/sec, wz rad/sec
OutputPlot t sec, theta deg, Hmag kg*m^2/sec^2
%-----
% Solve ODEs and generate plots.
% ODE() MGSpinStability3DRigidBodyFastInstability
%-----
% The exponential rate at which the instability grows depends on the
% inertia ratio Iratio = (Ixx - Iyy) * (Izz - Iyy) / (Ixx * Izz)
% Hence decreasing the difference between Ixx and Iyy results in a
% longer time for the instability to be visible (Dzhanibekov Effect).
% This is demonstrated in space-station YouTube videos.
% To further slow this effect to 70 seconds, use Izz = 2.1 kg*m^2.
%-----
Input Ixx := 1.9 kg*m^2, tFinal := 16 sec
ODE() MGSpinStability3DRigidBodySlowInstability
%-----
Save MGSpinStability3DRigidBody.html
Quit
```



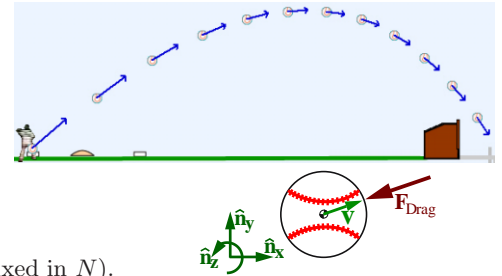


9.14 Kane's equations for projectile motion of a baseball



The following figure shows a baseball (particle Q) in projectile motion over Earth N (a Newtonian reference frame). Aerodynamic forces on the baseball are modeled as $-b\vec{v}$ where \vec{v} is Q 's velocity in N .

Description	Symbol	Type	Value
Mass of baseball (5.1 ozm)	m	Constant	145 gram
Earth's gravitational acceleration ($32.2 \frac{\text{ft}}{\text{s}^2}$)	g	Constant	9.8 m/s^2
Coefficient in drag force $-b\vec{v}$	b	Constant	$0.05 \frac{\text{N s}}{\text{m}}$
\hat{n}_x measure of Q 's position from N_o	x	Variable	
\hat{n}_y measure of Q 's position from N_o	y	Variable	



\hat{n}_x is horizontally-right, \hat{n}_y is vertically-upward, and N_o is home-plate (point fixed in N).

- Form **Kane's equations** for **generalized speeds** \dot{x} and \dot{y} . Then solve for \ddot{x} and \ddot{y} .

Kane's equation for generalized speed \dot{x} : $m\ddot{x} = -b\dot{x} \Rightarrow \ddot{x} = \frac{-b}{m}\dot{x}$

Kane's equation for generalized speed \dot{y} : $m\ddot{y} = -b\dot{y} - mg \Rightarrow \ddot{y} = -g - \frac{b}{m}\dot{y}$

- Form **Lagrange's equations** for the **generalized coordinates** x and y .
Lagrange's and Kane's equations produce the same results (for this problem). **True/False**.

- This system has a kinetic energy **True/False**.

This system has a potential energy (i.e., all forces are conservative) **True/False**.

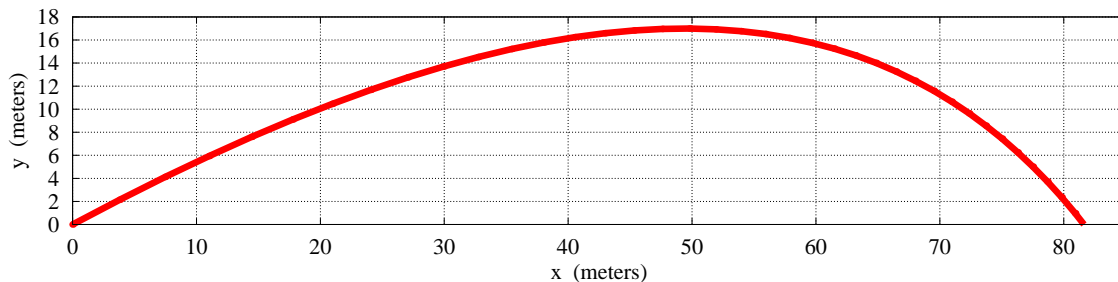
The definition of a **system** potential energy U requires all forces to be conservative so the system conserves mechanical energy (sum of kinetic plus potential energy is constant), i.e., $K + U = \text{Constant}$.

- Express the work done by aerodynamic drag in terms of \dot{x} and \dot{y} . Form an expression having units of energy that stays constant during the baseball's flight in N .

$$W_{\text{Drag}} = -b \int (\dot{x}^2 + \dot{y}^2) dt \quad \text{EnergyConstant} = K - W_{\text{Drag}} + mgy$$

Optional: Create an "x-y" plot of the baseball's trajectory when hit from home plate ($x = 0, y = 0$) with initial speed $44.7 \frac{\text{m}}{\text{s}}$ (100 mph) and initial angle of 30° . Ensure the expression for EnergyConstant is approximately constant (to numerical integrator accuracy).

Result:

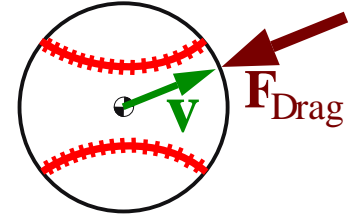
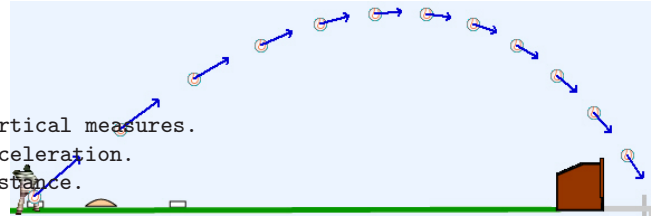


Solution at www.MotionGenesis.com \Rightarrow [Get Started](#) \Rightarrow Projectile motion.

```

% MotionGenesis file: MGProjectileMotionKane.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N          % Earth (baseball park).
Particle Q                 % Baseball.
%-----
Variable x'', y''         % Ball's horizontal and vertical measures.
Constant g = 9.8 m/s^2    % Earth's gravitational acceleration.
Constant b = 0.05 N*s/m   % Coefficient for air-resistance.
Q.SetMass( m = 145 grams )
SetGeneralizedSpeed( x', y' )
%-----
%      Translational kinematics (position, velocity, acceleration).
Q.Translate( No, x*Nx> + y*Ny> )
%-----
%      Add relevant forces (aerodynamic and gravity).
Q.AddForce( -m * g * Ny> )
Q.AddForce( -b * Q.GetVelocity(N) )
%-----
%      Form dynamics equations with Kane's method.
DynamicEqns = System.GetDynamicsKane()
%-----
%      Solve dynamics equations for x'', y''.
Solve( DynamicEqns = 0, x'', y'' )
%-----
%      Input integration parameters and initial values.
Input tFinal = 3.8 sec, tStep = 0.1 sec, absError = 1.0E-7
Input x = 0 m, x' = 44.7 * cosDegrees(30) m/s
Input y = 0 m, y' = 44.7 * sinDegrees(30) m/s
%-----
%      List output quantities and solve ODEs.
OutputPlot x m, y m
ODE() MGProjectileMotionKane
%-----
Save MGProjectileMotionKane.html
Quit

```



Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Projectile motion.

9.15 Implementing Kane's method with MotionGenesis



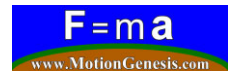
A representative procedure for forming Kane's equations of motion is summarized below.

Note: The MotionGenesis file **template.al** in MotionGenesis's MGToolbox folder outlines Kane's method.

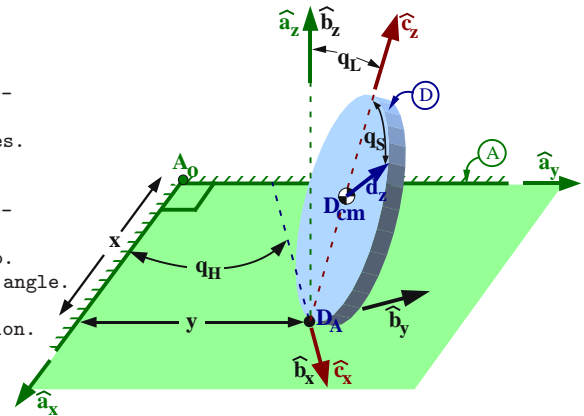
Concept	Representative MotionGenesis command
Choose a Newtonian reference frame	NewtonianFrame N
As needed, introduce additional reference frames	RigidFrame A
Name each rigid body	RigidBody B
Name each point (e.g., where a force is applied)	Point P
Name each particle	Particle Q
Assign a mass to each body and particle	B.SetMass(mB)
Assign an inertia dyadic to each rigid body	B.SetInertia(Bcm, IBxx, IByy, IBzz ...)
Choose variables and their time-derivatives	Variable q''
Choose generalized speeds	SetGeneralizedSpeed(q)
If needed, create kinematical differential equations	qx' = wx*cos(qx) + ...
Form rotation matrices , angular velocities , and angular accelerations	A.RotateZ(N, q)
Form position vectors and velocities to mass centers and points where forces are applied. Form accelerations of body mass centers.	Q.Translate(No, x*Nx> + y*Ny> + z*Nz>)
Impose constraints (e.g., position or velocity constraints)	Solve(MotionConstraints, variables)
Apply forces to points	Q.AddForce(m*g*Ny>)
Apply torques to rigid bodies	B.AddTorque(k*q*Bz>)
Form equations of motion (statics or dynamics)	System.GetDynamicsKane()
Specify input and output and solve the governing differential equations	ODE() someFilename

9.16 Dynamics of a rolling disk

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Rolling disk.



```
% MotionGenesis file: MGRollingDiskKane.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
% Problem: Simulation of a rolling disk with Kane's method.
%-----
NewtonianFrame A % Horizontal plane.
RigidFrame B, C % Heading and tilt reference frames.
RigidBody D % Rolling disk.
Point DA( D ) % Point of D in contact with A.
%-----
Variable wx', wy', wz' % Disk D's angular velocity in A.
Variable x'', y'' % Locates contact point DA from Ao.
Variable qH', qL', qS' % Heading angle, lean angle, spin angle.
Constant r = 0.343 meters % Radius of disk (13.5 inches).
Constant g = 9.8 m/s^2 % Earth's gravitational acceleration.
D.SetMass( m = 2 kg )
D.SetInertia( Dcm, C, J = 0.25*m*r^2, I = 2*J, J )
SetGeneralizedSpeed( wx, wy, wz )
%-----
% Rotational kinematics.
B.RotatePositiveZ( A, qH )
C.RotateNegativeX( B, qL )
D.RotatePositiveY( C, qS )
%-----
% For efficient dynamics, make a change of variable.
wDA> = D.GetAngularVelocity( A )
ChangeVariables[1] = wx - Dot( wDA>, Cx> )
ChangeVariables[2] = wy - Dot( wDA>, Cy> )
ChangeVariables[3] = wz - Dot( wDA>, Cz> )
Solve( ChangeVariables = 0, qH', qL', qS' )
%-----
% Use the simpler version of angular velocity from here on.
D.SetAngularVelocity( A, wx*Cx> + wy*Cy> + wz*Cz> )
%-----
% Form angular acceleration (differentiating in D is more efficient)
alf_D_A> = Dt( D.GetAngularVelocity(A), D )
%-----
% Translational kinematics.
Dcm.Translate( Ao, x*Ax> + y*Ay> + r*Cz> )
DA.SetPositionVelocity( Dcm, -r*Cz> )
%-----
% v_DA_A> is simpler if explicit in wx, wy, wz.
Explicit( v_DA_A>, wx, wy, wz )
%-----
% Motion constraints (D rolls on A at point DA).
RollingConstraint[1] = Dot( DA.GetVelocity(A), Ax> )
RollingConstraint[2] = Dot( DA.GetVelocity(A), Ay> )
SolveDt( RollingConstraint = 0, x', y' )
%-----
% Add relevant forces.
Dcm.AddForce( -m*g*Az> )
%-----
% Form and simplify Kane's equations of motion.
Dynamics = System.GetDynamicsKane()
FactorQuadratic( Dynamics, wx, wy, wz )
Solve( Dynamics = 0, wx', wy', wz' )
%-----
% Integration parameters and initial values.
Input tFinal = 12 sec, tStep = 0.05 sec, absError = 1.0E-08
Input x = 0 m, y = 0 m, qH = 0 deg, qL = 10 deg, qS = 0 deg
Input wx = 0 rad/sec, wy = 5 rad/sec, wz = 0 rad/sec
%-----
% List output quantities and solve ODEs.
OutputPlot t sec, x meters, y meters, qL degrees, qH degrees
ODE() MGRollingDiskKane
%-----
Save MGRollingDiskKane.html
Quit
```

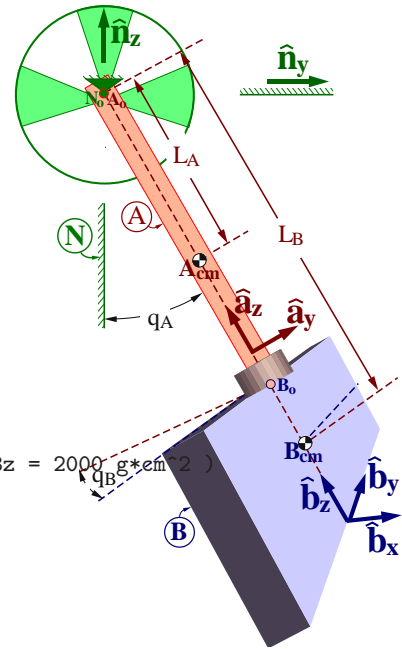


9.17 Kane's equations for a chaotic 3D pendulum



Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Chaotic pendulum.

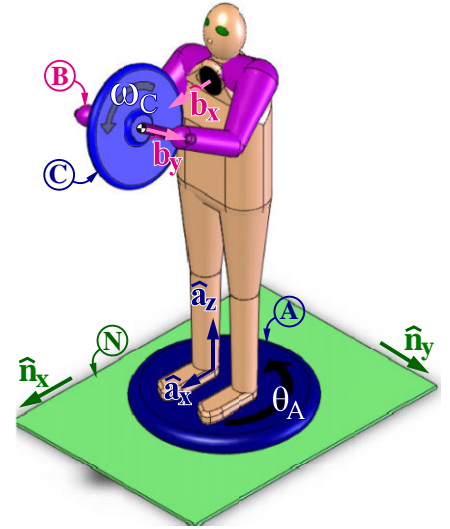
```
% MotionGenesis file: BabybootWithKaneLagrange.txt
% Problem: Analysis of 3D chaotic double pendulum.
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
SetDigits( 5 )      % Number of digits displayed for numbers
%-----
NewtonianFrame N
RigidBody      A      % Upper rod
RigidBody      B      % Lower plate
%-----
Variable      qA''    % Pendulum angle and its time-derivatives
Variable      qB''    % Plate angle and its time-derivative
Constant      LA = 7.5 cm      % Distance from pivot to A's mass center
Constant      LB = 20 cm      % Distance from pivot to B's mass center
Constant      g = 9.81 m/s^2   % Earth's gravitational acceleration.
A.SetMassInertia( mA = 10 grams, IAx = 50 g*cm^2, IAY, IAZ )
B.SetMassInertia( mB = 100 grams, IBx = 2500 g*cm^2, IBy = 500 g*cm^2, IBz = 2000 g*cm^2 )
%-----
%      Rotational and translational kinematics.
A.RotateX( N, qA )
B.RotateZ( A, qB )
Acm.Translate( No, -LA*Az> )
Bcm.Translate( No, -LB*Az> )
%-----
%      Add relevant forces.
System.AddForceGravity( -g*Nz> )
%-----
%      D'Alembert's equations of motion (MG road-maps).
Dynamics[1] = Dot( Ax>, System(A,B).GetDynamics(No) )
Dynamics[2] = Dot( Bz>, B.GetDynamics(Bcm) )
%-----
%      Kane's equations of motion (uses generalized speeds).
SetGeneralizedSpeed( qA', qB' )
Dynamics := System.GetDynamicsKane()
%-----
%      Kinetic and potential energy.
KE = System.GetKineticEnergy()
PE = System.GetForceGravityPotentialEnergy( -g*Nz>, No )
Energy = KE + PE
%-----
%      Lagranges's equations of motion (uses generalized coordinates).
SetGeneralizedCoordinates( qA, qB )
Dynamics := System.GetDynamicsLagrange( SystemPotential = PE )
Solve( Dynamics = 0, qA'', qB'' )
%-----
%      Integration parameters and initial values.
Input tFinal = 10 sec, tStep = 0.02 sec, absError = 1.0E-07, relError = 1.0E-07
Input qA = 90 deg, qA' = 0.0 rad/sec, qB = 1.0 deg, qB' = 0.0 rad/sec
%-----
%      List output quantities and solve ODEs.
OutputPlot t sec, qA deg, qB deg, Energy N*m
ODE() Babyboot
%-----
%      Record input together with responses
Save BabybootWithKaneLagrange.html
Quit
```



9.18 Kane's equations for a gyro on a turntable



```
% MotionGenesis file: MGHHumanOnTurntableWithGyroFBD.txt
%-----
NewtonianFrame N
RigidBody A % Turntable, human legs, torso, and head.
RigidFrame B % Human shoulder, arms, and wheel's axle.
RigidBody C % Bicycle wheel (rotor).
%-----
Variable qA'' % Az> measure of angle from Nx> to Ax>
Specified qB'' % Ax> measure of angle from Ay> to By>
Variable wC' % By> measure of C's angular velocity in B
Constant Lz = 1.2 m % Az> measure of Bo from No
Constant Lx = 0.5 m % Ax> measure of Ccm from Bo
C.SetMass( mC = 2 kg )
A.SetInertia( Acm, IAXX, IAYY, IAZZ = 0.64 kg*m^2 )
C.SetInertia( Ccm, B, IC = 0.12 kg*m^2, JC = 0.24 kg*m^2, IC )
%-----
% Rotational kinematics.
A.RotateZ( N, qA )
B.RotateX( A, qB )
C.SetAngularVelocityAcceleration( B, wC*By> )
%-----
% Translational kinematics.
Acm.SetVelocity( N, O> )
Bo.Translate( No, Lz*Az> )
Ccm.Translate( Bo, Lx*Ax> )
%-----
% Form equations of motion (angular momentum principle).
Dynamics[1] = Dot( Az>, System(A,C).GetDynamics(Bo) )
Dynamics[2] = Dot( By>, C.GetDynamics(Ccm) )
FactorQuadratic( Dynamics, qA', qB', wC )
%-----
% System angular momentum about vertical axis passing through Ao.
% Note: Its magnitude is not constant whereas Az> measure is constant.
SystemAngularMomentumAboutBo> = System.GetAngularMomentum( Bo )
SystemAngularMomentumAboutBoZ = Dot( SystemAngularMomentumAboutBo>, Az> )
SystemAngularMomentumMagnitudeAboutBo = GetMagnitude( SystemAngularMomentumAboutBo> )
SystemAngularMomentumMagnitudeAboutNo = GetMagnitude( System.GetAngularMomentum(No) )
%-----
% Bicycle C's angular momentum about vertical axis passing through Ccm.
% Note: Its magnitude is not constant whereas By measure is constant.
CAngularMomentumAboutCcm> = C.GetAngularMomentum( Ccm )
CAngularMomentumMagnitudeAboutCcm = GetMagnitude( CAngularMomentumAboutCcm> )
CAngularMomentumAboutCcmY = Dot( CAngularMomentumAboutCcm>, By> )
%-----
% Sum of kinetic and potential energy (not constant).
SystemKineticEnergy = System.GetKineticEnergy()
FactorQuadratic( SystemKineticEnergy, qA', qB', wC )
%-----
% Integration parameters and initial values.
Input tFinal = 8 sec, tStep = 0.02 sec, absError = 1.0E-08
Input qA = 0 deg, qA' = 0 rad/sec, wC = 40 rad/sec
%-----
% Specified expressions
SetDt( qB = pi/6 * sin( pi/2*t ) )
%-----
% Output quantities by ODE command.
Output t sec, qA deg, qA' deg/sec, qB deg, wC rad/sec, &
SystemAngularMomentumMagnitudeAboutNo kg*m^2/sec, &
SystemAngularMomentumMagnitudeAboutBo kg*m^2/sec, &
SystemAngularMomentumAboutBoZ kg*m^2/sec, &
CAngularMomentumMagnitudeAboutCcm kg*m^2/sec, &
CAngularMomentumAboutCcmY kg*m^2/sec, &
SystemKineticEnergy Joules
%-----
% Form and solve ODEs.
Solve( Dynamics = 0, qA'', wC' )
ODE() MGHHumanOnTurntableWithGyroFBD
%-----
Save MGHHumanOnTurntableWithGyroFBD.html
Quit
```

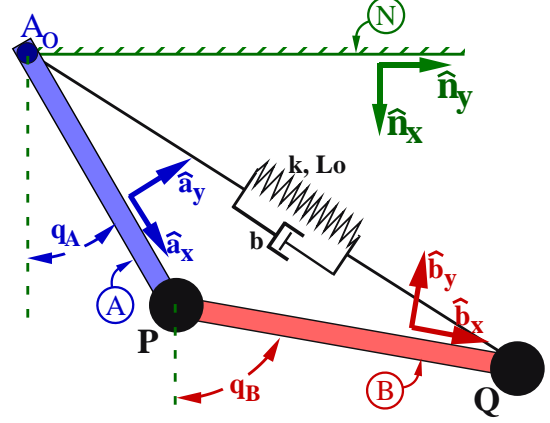


9.19 Spring-damper double pendulum: Forces and motion

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ [Pendulums](#).

The following figure shows two light rigid rods A and B and a spring-damper that support two particles P and Q in a Newtonian reference frame N . Rod A connects with frictionless revolute joints to N and B at points A_o and P , respectively. Right-handed sets of orthogonal unit vectors \hat{n}_i , \hat{a}_i , \hat{b}_i ($i = x, y, z$) are fixed in N , A , B , with \hat{n}_x vertically-downward, \hat{a}_x directed from A_o to P , \hat{b}_x directed from P to Q , and $\hat{n}_z = \hat{a}_z = \hat{b}_z$ parallel to the revolute joints' axes.

Quantity	Symbol	Value
Mass of P	m^P	10 kg
Mass of Q	m^Q	20 kg
Earth's gravitational constant	g	$9.8 \frac{m}{s^2}$
Distance from A_o to P	L_A	1 m
Distance from P to Q	L_B	2 m
Spring's natural length	L_o	1 m
Linear spring constant	k	$200 \frac{N}{m}$
Linear damping constant (force)	b	$100 \frac{N \cdot s}{m}$
Linear damping constant (torques)	c	$100 \frac{N \cdot m \cdot s}{rad}$
Angle from \hat{n}_x to \hat{a}_x with $+\hat{n}_z$ sense	q_A	Variable
Angle from \hat{n}_x to \hat{b}_x with $+\hat{n}_z$ sense	q_B	Variable



- Form statics equations governing q_A and q_B (when damping has stopped the system's motion).¹

Determine four static solutions for q_A and q_B between -180° and 180° .

Result: (Using intuition/guessing, circle the **stable** solutions).

$$\text{Static}_1 = L_A [L_B k s \frac{\sin(q_A - q_B)}{L_{\text{Spring}}} - g(m^P + m^Q) \sin(q_A)] = 0$$

$$\text{Static}_2 = L_B [-L_A k s \frac{\sin(q_A - q_B)}{L_{\text{Spring}}} - g(m^Q \sin(q_B))] = 0$$

$$\text{where } L_{\text{Spring}} = \sqrt{L_A^2 + L_B^2 + 2 L_A L_B \cos(q_A - q_B)} \quad s = L_{\text{Spring}} - L_o$$

Static solutions

#	q_A	q_B
1	0°	0°
2	-50.2°	35.2°
3	50.2°	-35.2°
4	180°	180°

- Form dynamics equations governing \ddot{q}_A and \ddot{q}_B (use air damping torque $\vec{T}^A = -c \dot{q}_A \hat{n}_z$ and $\vec{T}^B = -c \dot{q}_B$).

$$\text{Static}_1 + b L_A L_B \frac{\sin(q_A - q_B)}{L_{\text{Spring}}} \dot{s} - c \dot{q}_A = (m^P + m^Q) L_A^2 \ddot{q}_A + m^Q L_A L_B \cos(q_A - q_B) \ddot{q}_B + m^Q L_A L_B \sin(q_A - q_B) \dot{q}_B^2$$

$$\text{Static}_2 - b L_A L_B \frac{\sin(q_A - q_B)}{L_{\text{Spring}}} \dot{s} - c \dot{q}_B = m^Q L_A L_B \cos(q_A - q_B) \ddot{q}_A + m^Q L_B^2 \ddot{q}_B - m^Q L_A L_B \sin(q_A - q_B) \dot{q}_A^2$$

$$\text{where } \dot{s} = -L_A L_B \sin(q_A - q_B) (\dot{q}_A - \dot{q}_B) / L_{\text{Spring}}$$

- Plot q_A and q_B for $0 \leq t \leq 16$ sec when the system is released from **rest** with $q_A = 20^\circ$ and $q_B = 60^\circ$.

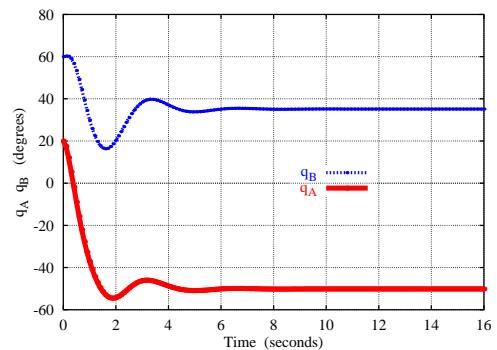
Verify the following static solution for q_A , q_B , and the \hat{n}_x and \hat{n}_y measures of the reaction force on A_o .

$$\text{Result: } q_A(t=16) \approx -50.2^\circ \quad q_B(t=16) \approx 35.2^\circ$$

$$F_x(t=16) \approx -294 \text{ N} \quad F_y(t=16) \approx 0 \text{ N}$$

- Optional:** Form a numerical integration energy checking function and verify it remains approximately constant.

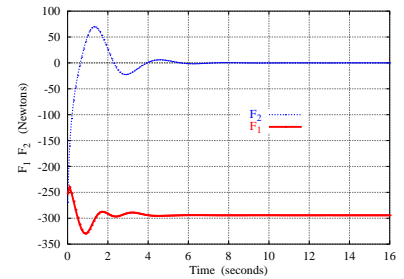
Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ [Pendulums](#).



¹Consider using **MG road-maps** or **Kane's equations** for **generalized speeds** \dot{q}_A , \dot{q}_B , or **Lagrange's equations** for **generalized coordinates** q_A , q_B . Generalized forces can be calculated from **potential energies**.

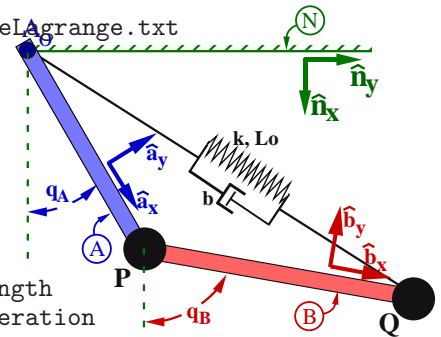
The set of contact forces exerted by N on A across the revolute joint at A_0 is equivalent to a couple of torque $T_x \hat{n}_x + T_y \hat{n}_y$ together with a force $F_x \hat{n}_x + F_y \hat{n}_y + F_z \hat{n}_z$ applied at A_0 . To plot F_x and F_y for $0 \leq t \leq 16$ sec, modify the file `MGSpringRestrainedDoublePendulumDynamics.txt` as follows:

- Augment generalized speeds with v_x and v_y as
`SetGeneralizedSpeed(qA', qB', vx, vy)`
- Set the actual values of the generalized speeds with:
`SetDt(vx = 0); SetDt(vy = 0)`
- Change the velocity of point A_0 in N to
`vx*Nx> + vy*Ny>`
- Augment the **ODE** and **Output** commands for F_x and F_y .



```
% MotionGenesis file: MGSpringRestrainedDoublePendulumStaticsKaneLagrange.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
```

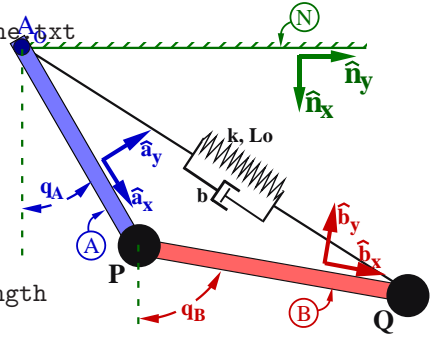
```
%-----
NewtonianFrame N
RigidFrame A, B % Rods
Particle P, Q % Particles at end of A, B
%-----
Variable qA', qB' % Angles for A and B
Constant LA = 1 m, LB = 2 m % Lengths of A, B
Constant k = 200 N/m, Ln = 1 m % Spring constant, natural length
Constant g = 9.8 m/s^2 % Earth's gravitational acceleration
P.SetMass( mP = 10 kg )
Q.SetMass( mQ = 20 kg )
%-----
% Rotational and translational kinematics.
A.RotateZ( N, qA )
B.RotateZ( N, qB )
P.SetPositionVelocity( No, LA*Ax> )
Q.SetPositionVelocity( P, LB*Bx> )
%-----
% Add relevant forces.
System.AddForceGravity( g*Nx> )
LSpring = Q.GetDistance( No )
SpringStretch = LSpring - Ln
UnitVectorFromNoToQ> = Q.GetPosition( No ) / LSpring
Q.AddForce( No, -k * SpringStretch * UnitVectorFromNoToQ> )
%-----
% Kane's equations for static equilibrium.
SetGeneralizedSpeed( qA', qB' )
Statics = System.GetStaticsKane()
%-----
% Lagrange equations for static equilibrium via potential energy U.
U = 1/2*k*SpringStretch^2 + System.GetForceGravityPotentialEnergy( g*Nx>, No )
SetGeneralizedCoordinate( qA, qB )
StaticsLagrange = System.GetStaticsLagrange( systemPotential = U )
%-----
% Solve nonlinear equations (requires a guess).
Solve( Statics, qA = -30 deg, qB = 30 deg )
%-----
Save MGSpringRestrainedDoublePendulumStaticsKaneLagrange.html
Quit
```



```

% MotionGenesis file: MGSpringRestrainedDoublePendulumDynamicsKane.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N
RigidFrame A, B % Rods
Particle P, Q % Particles at end of A, B
%-----
Variable qA'', qB'' % Angles for A and B
Constant LA = 1 m, LB = 2 m % Lengths of A, B
Constant k = 200 N/m, Ln = 1 m % Spring constant, natural length
Constant b = 100 N*s/m % Damping constant
Constant c = 100 N*m*s/rad % Damping constant
Constant g = 9.8 m/s^2 % Earth's gravitational acceleration
P.SetMass( mP = 10 kg )
Q.SetMass( mQ = 20 kg )
Variable Fx, Fy % Contact forces on Ao from No
Specified Stretch' % Elongation of spring between O and Q
%-----
Variable vx', vy' % Auxiliary generalized speeds to calculate Fx and Fy.
SetGeneralizedSpeed( qA', qB', vx, vy )
SetDt( vx = 0 ); SetDt( vy = 0 )
%-----
% Rotational and translational kinematics
A.RotateZ( N, qA )
B.RotateZ( N, qB )
Ao.SetVelocityAcceleration( N, vx*Nx> + vy*Ny> )
P.Translate( Ao, LA*Ax> )
Q.Translate( P, LB*Bx> )
%-----
% Relevant forces for statics (gravity, spring, contact forces).
System.AddForceGravity( g*Nx> )
LSpring = Q.GetDistance( Ao ) % Distance between Q and Ao
SetNoDt( Stretch = LSpring - Ln ) % Calculate spring stretch
UnitVectorFromAoToQ> = Q.GetPosition( Ao ) / LSpring
Q.AddForce( Ao, -k * Stretch * UnitVectorFromAoToQ> )
Ao.AddForce( No, Fx*Nx> + Fy*Ny> ) % Contact force on A from N
%-----
% Damping force and torques.
Stretch' = Dot( Q.GetVelocity(N), UnitVectorFromAoToQ> )
Q.AddForce( Ao, -b * Stretch' * UnitVectorFromAoToQ> )
A.AddTorque( -c * qA' * Az> )
B.AddTorque( -c * qB' * Bz> )
%-----
% Equations of motion via Kane's method.
Zero = System.GetDynamicsKane()
%-----
% Integration parameters and initial values.
Input tFinal = 16, tStep = 0.1, absError = 1.0E-07
Input qA = 20 deg, qB = 60 deg, qA' = 0 rad/sec, qB' = 0 rad/sec
%-----
% List output quantities and solve ODEs.
EnergyCheck = System.GetEnergyCheckKane()
OutputPlot t sec, qA deg, qB deg, Fx Newtons, Fy Newtons, EnergyCheck Joules
ODE( Zero, qA'', qB'', Fx, Fy ) MGSpringRestrainedDoublePendulumDynamicsKane
%-----
Save MGSpringRestrainedDoublePendulumDynamicsKane.html
Quit

```



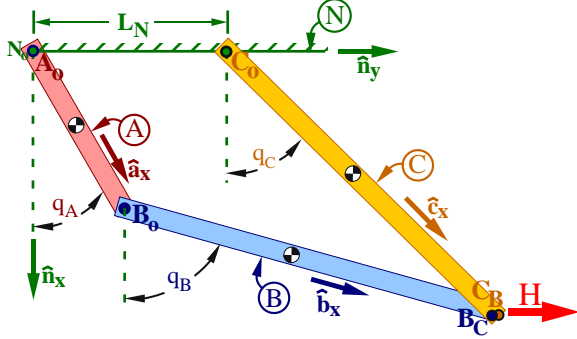
9.20 Four-bar linkage: Motion and contact forces (see statics in Section 8.3)

Form this system's dynamics equations with two or more of the following.

Type	<i>Kane's method</i>	<i>Lagrange's method</i>	<i>MG road-map</i>
Augmented	<i>generalized speeds</i> $\dot{q}_A, \dot{q}_B, \dot{q}_C$	<i>generalized coordinates</i> $\dot{q}_A, \dot{q}_B, \dot{q}_C$	$\dot{q}_A, \dot{q}_B, \dot{q}_C$
Embedded	<i>generalized speed</i> \dot{q}_A	No	No

Simulate the motion of the four-bar linkage. Use initial values $q_A = 30^\circ$, $\dot{q}_A = 0$. Plot q_A, q_B, q_C for 7 seconds. Use the plot to estimate the four-bar linkage's oscillation period as $\tau_{\text{period}} \approx 2$ sec. Note: $q_B = 74.47751219^\circ$, $q_C = 45.52248781^\circ$ satisfy the “**loop equation**” when $q_A = 30^\circ$.

†Optional: Determine τ_{period} by linearizing the dynamics about the static equilibrium solution.



Stable statics via dynamics with damping.

One way to find a **stable static** solution is to simulate the dynamic system with damping (e.g., $H = 200 - 80\dot{q}_C$) until the system settles (stops moving).

Determine q_A, q_B, q_C when the system stops moving.

Result: $q_A \approx 20.0^\circ$ $q_B \approx 71.7^\circ$ $q_C \approx 38.3^\circ$

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Four-bar linkage

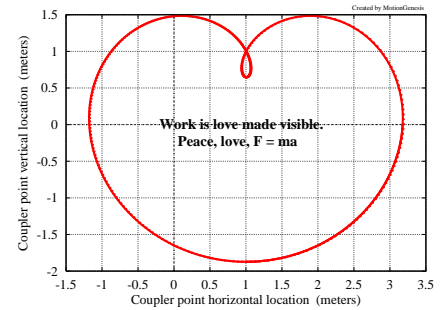
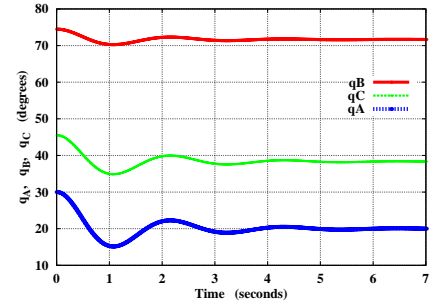
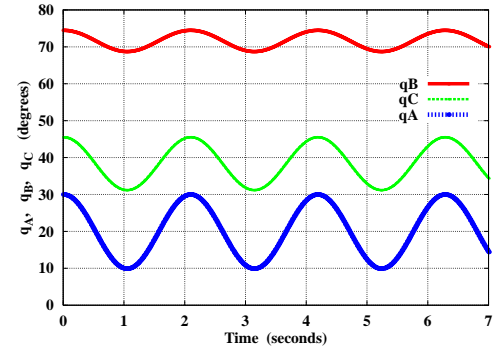
Work is love made visible.

Modify the four-bar so $L_N = 2$ m and $L_A = L_B = L_C = 4$ m. Use $m^A = m^B = m^C = 20$ kg. Remove the horizontal force H and add a motor of torque $T_A = 9600(\omega_{\text{des}} - \dot{q}_A)$ on crank-link A , where $\omega_{\text{des}} = 60 \frac{\text{deg}}{\text{sec}}$. Start from rest with coupler-link B horizontal ($q_B = 90^\circ$) and simulate for 7 seconds. Knowing the “coupler-point” position from B_0 is $2\hat{b}_x + 2\hat{b}_y$, plot the coupler point's vertical vs. horizontal location.

Solution at www.MotionGenesis.com ⇒ [Get Started](#) ⇒ Four-bar linkage

Form a numerical integration checking function and verify that it remains constant during numerical integration. Verify that the configuration constraint equation (which also serve as numerical integration checking functions) are satisfied during numerical integration.

Result: After running the file `MGFourBarForcesAndMotion.txt` in **MotionGenesis**, the file `MGFourBarForcesAndMotion.1` shows time-histories for `Loop[1]`, `Loop[2]`, and `ECheck`.



The set of contact forces exerted by N on A across the revolute joint at A_o is equivalent to a couple of torque $T_x^A \hat{n}_x + T_y^A \hat{n}_y$ together with a force $F_x^A \hat{n}_x + F_y^A \hat{n}_y + F_z^A \hat{n}_z$ applied at A_o .

The set of contact forces exerted by B on C across the revolute joint at C_B is equivalent to a couple of torque $T_x^C \hat{n}_x + T_y^C \hat{n}_y$ together with a force $F_x^C \hat{n}_x + F_y^C \hat{n}_y + F_z^C \hat{n}_z$ applied at C_B .

- Plot the time histories of F_x^C and F_y^C for $0 \leq t \leq 10$ sec.

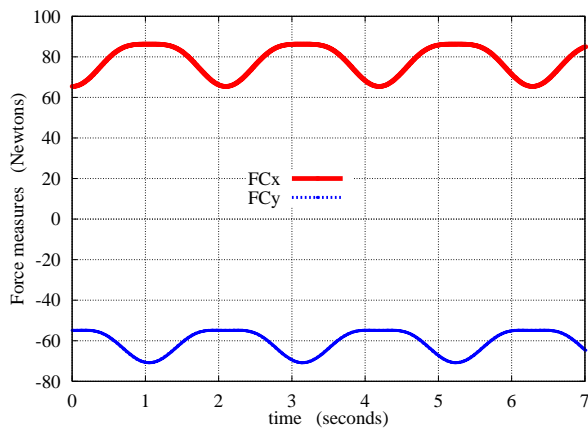
To find F_x^C and F_y^C , the file `MGFourBarForcesAndMotion.txt` was modified as follows:

Change these lines	To these lines
SetGeneralizedSpeed(qA') ODE(Zero, qA'')	SetGeneralizedSpeed(qA', qB', qC') ODE(Zero, qA'', FCx, FCy)
Uncomment/add these lines	CB.AddForce(BC, FCx*Nx> + FCy*Ny>) Output t sec, FCx N, FCy N

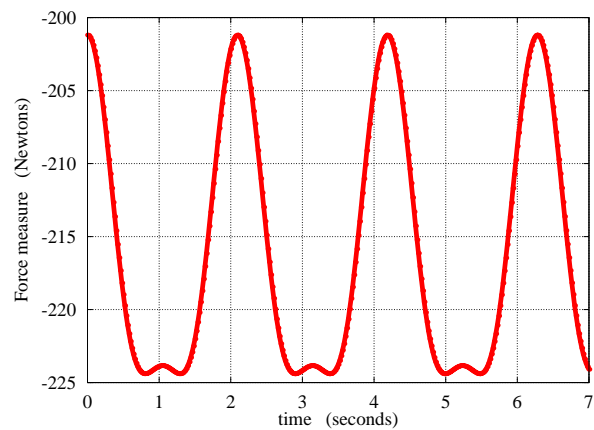
- Plot the time-history of F_x^A for $0 \leq t \leq 10$ sec.

To find F_x^A , again modify the file `MGFourBarForcesAndMotion.txt`.

Change these lines	To these lines
Variable qA'', qB'', qC''	Variable qA'', qB'', qC'', vx'
SetGeneralizedSpeed(qA', qB', qC') Ao.SetVelocityAcceleration(N, 0>) Solve(Zero, qA'', FCx, FCy) Output t sec, FCx N, FCy N	SetGeneralizedSpeed(qA', qB', qC', vx') Ao.SetVelocityAcceleration(N, vx*Nx>) Solve(Zero, qA'', FCx, FCy, FAx) Output t sec, FCx N, FCy N, FAx N
Uncomment/add these lines	SetDt(vx = 0)



Time-histories of F_x^C and F_y^C

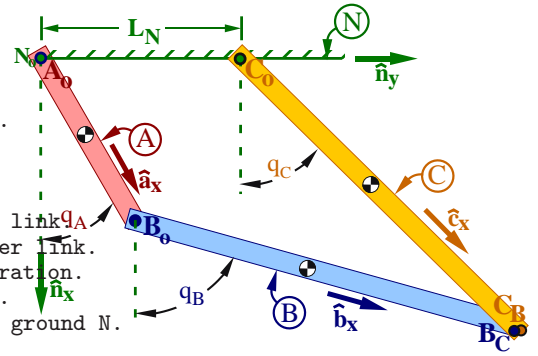


Time-history of F_x^A

```

% MotionGenesis file: MGFourBarForcesAndMotion.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N % Ground link.
RigidBody A, B, C % Crank, coupler, rocker links.
Point BC( B ) % Point of B connected to C.
Point CB( C ) % Point of C connected to B.
%-----
Constant LN = 1 m, LA = 1 m % Length of ground link, crank link.
Constant LB = 2 m, LC = 2 m % Length of coupler link, rocker link.
Constant g = 9.81 m/s^2 % Earth's gravitational acceleration.
Specified H = 200 % Horizontal force at point CB.
Specified TA = 0 % Motor torque on crank A from ground N.
Variable qA'', qB'', qC'' % , vx'
Variable FAX, FAY, FCx, FCy % Contact forces
SetGeneralizedSpeed( qA', qB', qC' ) % , vx
SetAutoZee( ON ) % Efficient calculations.
SetNoZeeSymbol( FAX, FAY, FCx, FCy )
%-----
A.SetMassInertia( mA = 10 kg, 0, IA = mA*LA^2/12, IA )
B.SetMassInertia( mB = 20 kg, 0, IB = mB*LB^2/12, IB )
C.SetMassInertia( mC = 20 kg, 0, IC = mC*LC^2/12, IC )
%-----
% Rotational kinematics.
A.RotateZ( N, qA )
B.RotateZ( N, qB )
C.RotateZ( N, qC )
%-----
% Translational kinematics.
Ao.Translate( No, 0> ) % Ao.SetVelocityAcceleration( N, vx*Nx> )
Acm.Translate( Ao, 0.5*LA*Ax> )
Bo.Translate( Ao, LA*Ax> )
Bcm.Translate( Bo, 0.5*LB*Bx> )
BC.Translate( Bo, LB*Bx> )
Co.Translate( No, LN*Ny> )
Ccm.Translate( Co, 0.5*LC*Cx> )
CB.Translate( Co, LC*Cx> )
%-----
% Add relevant forces and torques.
System.AddForceGravity( g * Nx> )
CB.AddForce( H * Ny> )
Ao.AddForce( FAX*Nx> + FAY*Ny> )
CB.AddForce( BC, FCx*Nx> + FCy*Ny> ) % "Cut" linkage at CB/BC
A.AddTorque( N, TA * Az> )
%-----
% Configuration (loop) constraints and their time-derivatives.
Loop> = LA*Ax> + LB*Bx> - LC*Cx> - LN*Ny>
Loop[1] = Dot( Loop>, Nx> )
Loop[2] = Dot( Loop>, Ny> )
%-----
% Use the loop constraints to solve for initial values of qB, qC and qB',qC'
% (results depend on constants and initial values of qA and qA').
Input qA = 30 deg, qA' = 0 rad/sec
SolveSetInputDt( Loop = 0, qB = 60 deg, qC = 20 deg )
%-----
% Equations of motion with Kane's method.
Dynamics = System.GetDynamicsKane()
%-----
% Numerical integration parameters.
Input tFinal = 7 sec, tStep = 0.02 sec, absError = 1.0E-07
%-----
% List quantities to be output from ODE.
ECheck = System.GetEnergyCheckKane()
Output t sec, qA deg, qB deg, qC deg, Loop[1] m, Loop[2] m, ECheck Joules
Output t sec, FCx Newtons, FCy Newtons % , FAX Newtons
%-----
% Numerically solve the ODEs for qA'' (and perhaps FCx, FCy, FAX).
ODE( Dynamics = 0, qA'', FCx, FCy ) MGFourBarForcesAndMotion
%
%
%

```



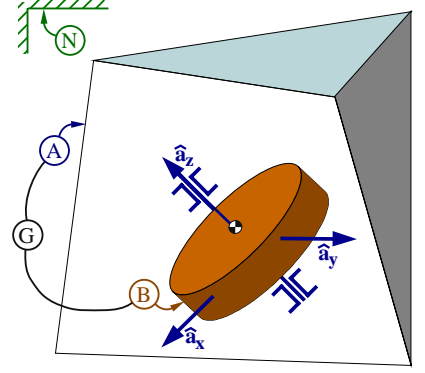
```

% Design 4-bar linkage to draw Valentines heart.
Input LN := 2, LA := 4 m, LB := 4 m, LC := 4 m, mA := 20 kg
%-----
% Control motor torque so qA' is nearly 60 degrees/sec.
Constant wMotor = 60 deg/sec % Desired motor angular speed.
Constant kw = 9600 N*m/rad % Control constant.
TA := kw * (wMotor - qA') % Motor torque on crank A from ground N.
H := 0 % Remove horizontal force at point CB.
%-----
% Output coupler point position for Valentines heart.
Point CouplerPoint( B )
Constant CouplerDistance = sqrt(8) m % For conventional heart, use 2.66
Constant CouplerAngle = 45 degrees % For conventional heart, use 41.20
couplerPointBx = CouplerDistance * cos(CouplerAngle)
couplerPointBy = CouplerDistance * sin(CouplerAngle)
CouplerPoint.SetPosition( Bo, couplerPointBx * Bx + couplerPointBy * By )
couplerPointHorizontal = Dot( CouplerPoint.GetPosition(No), Ny )
CouplerPointVertical = -Dot( CouplerPoint.GetPosition(No), Nx )
Output couplerPointHorizontal m, CouplerPointVertical m, qA' deg/sec, TA N*m
%-----
% Initial configuration is coupler link B horizontal (qB = 90 degrees).
% Use the loop constraints to solve for initial values of qA, qB.
% Use loopDt to solve for initial values of qB', qC'.
Input qB := 90 deg, qA' := Input(wMotor, noUnitSystem) deg/sec
SolveSetInput( Loop = 0, qA = -15 deg, qC = 15 deg )
%-----
% Augment dynamics with constraints and solve ODEs (plot results).
ODE( Dynamics = 0, qA'', FCx, FCy ) MGFourBarDynamicsValentinesHeart
Plot MGFourBarDynamicsValentinesHeart.3 [1, 2]
%-----
% Save input together with program responses.
Save MGFourBarForcesAndMotion.html
Quit

```

9.21 Gyrostat spin stabilization

The figure to the right shows a gyrostat G consisting of a carrier A and a thin uniform cylindrical rotor B moving in a Newtonian frame N . Dextral sets of orthogonal unit vectors $\hat{\mathbf{a}}_x, \hat{\mathbf{a}}_y, \hat{\mathbf{a}}_z$ are fixed in A and are parallel to G 's central principal inertia axes. The rotor B has a central moment of inertia of J about its symmetric axes, which is parallel to $\hat{\mathbf{a}}_z$. G 's central principal moments of inertia for $\hat{\mathbf{a}}_x, \hat{\mathbf{a}}_y, \hat{\mathbf{a}}_z$ are denoted I_{xx}, I_{yy}, I_{zz} , respectively. The generalized speeds $\omega_x, \omega_y, \omega_z$ are the $\hat{\mathbf{a}}_x, \hat{\mathbf{a}}_y, \hat{\mathbf{a}}_z$ measures of A 's angular velocity in N . The constant Ω is the $\hat{\mathbf{a}}_z$ measure of B 's angular velocity in A .



When numerical values are required, use $J = 0.07634 \text{ kg m}^2$, $I_{xx} = 1.25 \text{ kg m}^2$, $I_{yy} = 4.25 \text{ kg m}^2$, $I_{zz} = 5 \text{ kg m}^2$, $\omega_{z\text{nom}} = 1 \frac{\text{rad}}{\text{sec}}$.

- Form equations of motion which govern angular motions of the system.

Result:

$$I_{xx} \dot{\omega}_x + \omega_y [J\Omega - (I_{yy} - I_{zz})\omega_z] = 0$$

$$I_{yy} \dot{\omega}_y - \omega_x [J\Omega - (I_{xx} - I_{zz})\omega_z] = 0$$

$$I_{zz} \dot{\omega}_z - (I_{xx} - I_{yy})\omega_x \omega_y = 0$$

- Consider the nominal solution $\mathbf{w}_x = \mathbf{w}_y = \mathbf{w}_x' = \mathbf{w}_y' = \mathbf{w}_z' = 0$, $\mathbf{w}_z = \mathbf{nwz}$ where \mathbf{nwz} is a constant. After introducing the variables $\mathbf{dw}_x, \mathbf{dw}_y, \mathbf{dw}_z$ as perturbations of $\omega_x, \omega_y, \omega_z$, linearize the equations of motion in the perturbations about the nominal solution. Put the linearized equations in the form $\mathbf{X}' = \mathbf{A}*\mathbf{X}$ where \mathbf{A} is a 3×3 coefficient matrix and \mathbf{X} is the 3×1 state matrix $[\mathbf{dw}_x; \mathbf{dw}_y; \mathbf{dw}_z]$.

Result:

$$\mathbf{A} = \begin{bmatrix} 0 & -\frac{J\Omega - (I_{yy} - I_{zz})\omega_{z\text{nom}}}{I_{xx}} & 0 \\ \frac{J\Omega - (I_{yy} - I_{zz})\omega_{z\text{nom}}}{I_{yy}} & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

- Determine values of Ω which result in eigenvalues λ of \mathbf{A} that are positive.

Result: The MotionGenesis response $0.0011(9.824 + \Omega)(49.12 + \Omega) + \lambda^2 = 0$ shows

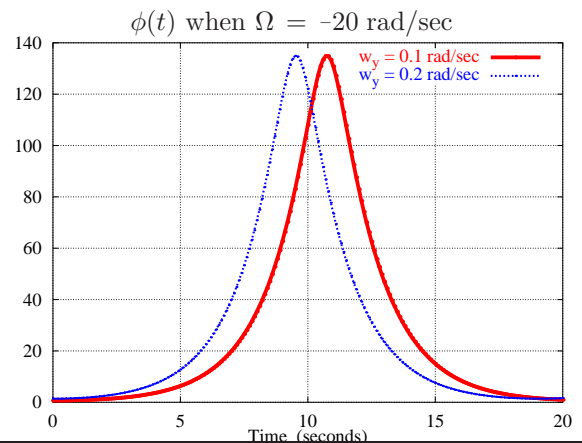
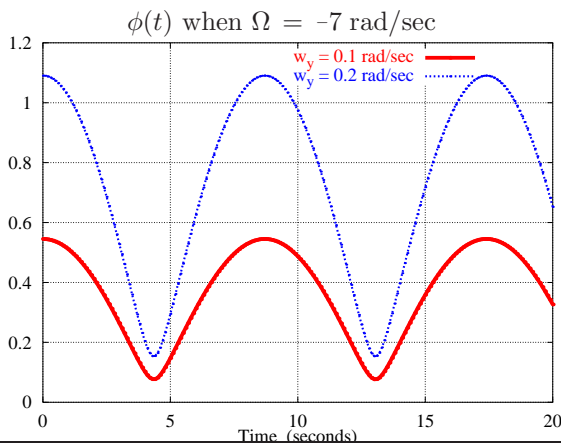
λ^2 is positive when	$-49.12 < \Omega < -9.82$	$\text{Real}(\lambda) > 0$	so solution is “unstable”
λ^2 is negative when	$\Omega > -9.82$ or $\Omega < -49.12$	$\text{Real}(\lambda) = 0$	so solution is “stable”

- Using the nonlinear equations of motion, run four 20 sec simulations, all with $\omega_x = 0$ and $\omega_z = 1$.

Plot the time-history of ϕ , the angle between $\hat{\mathbf{a}}_z$ and the inertial angular momentum of G . For each simulation, check that H (the magnitude of the gyrostat's angular momentum in N) is time-invariant.

Result: See the plots. By examining (or plotting) the simulation output file MGGyrostatSpinStability.1, it is clear that H is time-invariant.

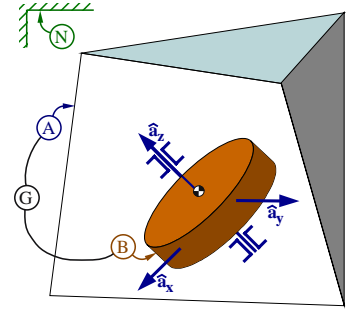
#	Ω	ω_y
A	-7	0.02
B	-7	0.01
C	-20	0.02
D	-20	0.01




```

% MotionGenesis file: MGGyrostatSpinStability.txt
% Copyright (c) 2009 Motion Genesis LLC. All rights reserved.
%-----
NewtonianFrame N          % Newtonian reference frame
RigidBody       A          % Carrier
RigidFrame     B          % Rotor
%-----
Constant Omega = -20 rad/sec % B's angular speed in A
Constant J = 0.07634 kg*m^2 % B's moment of inertia about spin axis
Variable wx', wy', wz'
%-----
%      Mass and inertia (attribute G's inertia to A)
A.SetMassInertia( m, Ix = 1.25 kg*m^2, Iy = 4.25 kg*m^2, Iz = 5 kg*m^2 )
%-----
%      Rotational and translational kinematics.
%      Note: Velocity/acceleration of G's mass center is 0>.
A.SetAngularVelocityAcceleration( N, wx*Ax> + wy*Ay> + wz*Az> )
B.SetAngularVelocityAcceleration( A, Omega*Az> )
Acm.SetVelocityAcceleration( N, 0> )
%-----
%      Form Kane's equations of motion.
SetGeneralizedSpeed( wx, wy, wz )
Dynamics = System.GetGeneralizedForce() + Frstar() + Gyrostat(FrStar,CYLINDER,A,B,J)
%-----
%      Calculate gyrostat's angular momentum .
H> = System.GetAngularMomentum(Acm) + Gyrostat(Angmom,CYLINDER,A,B,J)
H = GetMagnitude( H> )
phi = AngleBetweenUnitVectors( GetUnitVector( H> ), Az> )
%-----
%      Integration parameters and initial values.
Input tFinal = 20 sec, tStep = 0.1 sec, absError = 1.0E-07
Input wx = 0 rad/sec, wy = 0.02 rad/sec, wz = 1 rad/sec
%-----
%      List output quantities and solve ODEs.
Output t sec, phi degs, H kg*m^2/s
ODE( Dynamics, wx', wy', wz' ) MGGyrostatSpinStability
%*****
%      STABILITY ANALYSIS
%*****
%      Linearization: Perturbation vars + nominal solution parameters.
Variable dwx', dwy', dwz' % Perturbations of wx, wy, wz.
Constant nwz = 1 rad/sec % Nominal solution for wz.
%-----
%      Check nominal solution satisfies the equations of motion.
Check = Evaluate( Dynamics, wx=0, wx'=0, wy=0, wy'=0, wz=nwz, wz'=0 )
%-----
%      Linearize equations of motion about nominal solution.
Perturb = Linearize1( Dynamics, wx = 0 : dwx, wx' = 0 : dwx', wy = 0 : dwy, &
                    wy' = 0 : dwy', wz = nwz : dwz, wz' = 0 : dwz' )
Solve( Perturb, dwx', dwy', dwz' )
%-----
%      Form, X, X', and A matrices in the matrix equation X' = A * x
Xm = [ dwx; dwy; dwz ]
Xp = Dt( Xm )
Am = D( Xp, Transpose(Xm) )
%-----
%      To find eigenvalues of Am symbolically, find the roots of the
%      equation found by setting determinant( Lambda * I - A ) = 0.
Variable Lambda
det = Determinant( Lambda * GetIdentityMatrix(3) - Am )
det /= Lambda % Inspection of det shows Lambda = 0 is a root
%-----
%      Find values of Omega which result in Lambda > 0.
det := EvaluateToNumber( det, Omega = Omega )
%-----
Save MGGyrostatSpinStability.html
Quit

```



10 Other information

10.1 Functions and commands

Type **HELP** at a **MotionGenesis** line prompt to see an on-screen list of ≈ 100 commands.

Type **HELP commandName** for detailed help (e.g., type **HELP Solve** for help with the **Solve** command).

Most commands may be nested. For example, the **Dot** command may appear as an argument of the **acos** command, e.g., `theta = acos(Dot(Ax>, By>))`.

10.2 Stand-Alone commands

Many commands such as `Renee = cos(x)` use an equals sign to make an assignment. Alternately, “**stand-alone commands**” make assignments without an explicit equals sign. For example:

$$3*x + 4*y = 37$$

$$4*x - 2*y = -2$$

can be solved by executing the following **MotionGenesis** input file

```
(1) Variable  x, y
(2) Zero[1] = -37 + 3*x + 4*y
-> (3) Zero[1] = -37 + 3*x + 4*y
(4) Zero[2] = 2 + 4*x - 2*y
-> (5) Zero[2] = 2 + 4*x - 2*y
(6) Solve( Zero,  x, y )
-> (7) x = 3
-> (8) y = 7
```

In line 6, the command `Solve(Zero, x, y)` causes values to be assigned to `x` and `y`, but the command does not involve the typing of any equals sign. Other stand-alone commands include **Rotate**, **Translate**, **SetVelocity**,

10.3 Dual functions

The commands **Arrange**, **Expand**, **Explicit**, **Express**, **Factor**, and **Zee** are called *dual functions* because they can be used in two ways, namely, on the right-hand side of an equals sign, e.g., `NewY = Explicit(y)`, or as stand-alone commands. When a dual-function appears on an **MotionGenesis** input line in the form

`DualFunctionName(X, list_of_arguments)`

where `X` is the *name* of an expression (not the expression itself) and `list_of_arguments` stands for arguments of the dual function, then **MotionGenesis** interprets this line as

`X := DualFunctionName(X, list_of_arguments)`

Dual functions differ from other commands in that they avoid changing the functional character of an expression, but alter its appearance.

10.4 Creating your own commands: .A and .R files

You can create new commands by creating an ASCII file that resides in the **MotionGenesis MGTtoolbox** directory, the current working directory, or a directory that is specified in the **AUTOLEVPATH**. For example, suppose you wish to create a command called **SUM**, which adds two expressions and returns their sum. The syntax of the **SUM** command could be **SUM(x,y)**, where **x** and **y** are expressions. To create the command **SUM**, use a text editor to compose a file named **sum.r** with the following contents:

```
%SUM.R
%
%Function: Returns the sum of two expressions.
%
% Syntax: SUM(x,y)
%
#1# + #2#
```

The filename has a **.r** extension to inform **MotionGenesis** that **SUM** returns a result (**.r** for result). The **#1#** and **#2#** denote the two arguments of **SUM**. The comments at the top of the file contain text that appears on the screen when **Help SUM** is typed at the **MotionGenesis** line prompt. Typing **Wow = SUM(3,5)** results in **Wow = 8**.

Next, suppose you want to create a command which makes assignments. For example, to create a command called **POWER**, which squares and cubes an expression and assigns the results to new variables called *nameSQ* and *nameCUBE*, compose a file named **power.a** with the following contents:

```
%POWER.A
%
%Function: Put explanatory information about POWER here.
%
%Syntax: POWER(expression,name)
%
#2#SQ = (#1#)^2
#2#CUBE = (#1#)^3
```

The filename has a **.a** extension to inform **MotionGenesis** that **POWER** makes assignments (**.a** for assignment). Note the use of parentheses around the **#1#** argument. Liberal use of parentheses is helpful in making bug-free **.a** and **.r** files. To produce efficient results when **AUTOZ** is **ON**, use the **AUTOZ()** command which introduces intermediate symbols. Use of the **POWER** command is demonstrated below.

```
(1) POWER( 3, a )
->(2) aSQ = 9
->(3) aCUBE = 27
(4) Constants a, b
(5) POWER( a+b, Ed )
->(6) EdSQ = (a+b)^2
->(7) EdCUBE = (a+b)^3
```

There are special symbols to assist in creating **.a** and **.r** files. The symbol **#NUM_ARGS#** evaluates to the number of arguments passed to the **.a** or **.r** command; **#NewtonianFrame#** evaluates to the name declared in the **NewtonianFrame** declaration. In addition, one may use the logical **if** and **else** statements, the **ECHO** command, and the special symbols **\k**, **\a**, **\p**, **\n** that are used with **ECHO** (type **HELP ECHO** for more information). For example,

```
%SUM.R
%
%Function:  Returns the sum of two expressions.
%
% Syntax:  SUM(x,y)
%
if( #NUM_ARGS# != 2 )
    { Echo(\k\a"Error: wrong number of arguments to the SUM command"\p\n); }
else
    { #1# + #2# }
```

10.5 Default settings

The manner in which **MotionGenesis** responds to online commands or when executing a batch file depends on certain “settings”. For example, the factoring of expressions depends on whether **SetAutoFactor** is **ON** or **OFF**, as seen by comparing the two **MotionGenesis** sessions below:

(1) SetAutoFactor(OFF)	(1) SetAutoFactor(ON)
(2) Constants a, b, c	(2) Constants a, b, c
(3) $x = a*b + a*c + a^2$	(3) $x = a*b + a*c + a^2$
-> (4) $x = a*b + a*c + a^2$	-> (4) $x = a*(a+b+c)$
(5) $y = a/c + b/c$	(5) $y = a/c + b/c$
-> (6) $y = a/c + b/c$	-> (6) $y = (a+b)/c$

SetAutoRhs is another default setting.

- When **SetAutoRhs** is **ALL**, the right-hand side of all quantities are automatically substituted for the quantity.
- When **SetAutoRhs** is **ON**, substitutions are made when the right-hand side of quantities are “simple”.
- When **SetAutoRhs** is **OFF**, no substitutions are made.

One way to see the effect of **SetAutoRhs** is to compare the files below.

(1) SetAutoRhs(OFF)	(1) SetAutoRhs(ON)	(1) SetAutoRhs(ALL)
(2) Constant b	(2) Constant b	(2) Constant b
(3) a = 7	(3) a = 7	(3) a = 7
-> (4) a = 7	-> (4) a = 7	-> (4) a = 7
(5) $c = a + b$	(5) $c = a + b$	(5) $c = a + b$
-> (6) $c = a + b$	-> (6) $c = 7 + b$	-> (6) $c = 7 + b$
(7) $d = c*\sin(c)$	(7) $d = c*\sin(c)$	(7) $d = c*\sin(c)$
-> (8) $d = c*\sin(c)$	-> (8) $d = c*\sin(c)$	-> (8) $d = (7+b)*\sin(7+b)$

To view the values assigned to all settings, type **DEFAULTS** at a line prompt.

10.6 Special symbols (see also Reserved Names in Section 2.4)

%	Comment delimiter. Input following this character is ignored
%%	Inserts a comment in MATLAB, C, or FORTRAN code
&	Line continuation character (Autolev input files only)
' (prime)	Implies total differentiation with respect to T
>	The last symbol in the name of a vector
>>	The last two symbols in the name of a dyadic
>>>	The last three symbols in the name of a triadic or higher order polyadic
[]	Used to denote a matrix or designate an element of a matrix
{ }	Encloses indices in declarations
,	Separates arguments of a function and elements of a row in a matrix
;	Separates rows of a matrix
:	Designates a range, e.g., 1:3
()	Encloses mathematical expressions and function arguments
#	Delimits arguments in .A and .R files
"	Delimits string literals
.	Decimal point
+ - * / ^	Mathematical operators: addition, subtraction, multiplication, division, and exponentiation
= := += -=	Assignment operators: normal, overwrite, addition, and subtraction
*= /= ^ =	Assignment operators: multiplication, division, and exponentiation

10.7 Editing keys for online editing in PC/Windows

HOME	Moves cursor to the beginning of input line
END	Moves cursor to the end of input line
DEL	Deletes current character in line
BACKSPACE	Deletes previous character in line
RIGHT ARROW →	Moves cursor one space to the right
LEFT ARROW ←	Moves cursor one space to the left
UP ARROW ↑	Recovers previous line
INSERT	Toggles between insert and overstrike editing modes
Control-C	Abruptly terminates program execution